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ANALYSIS OF DYNAMIC IN SITU BACKFILL PROPERTY TESTS

Report 3

THE MODIFIED ONED PROGRAM FOR ONE-DIMENSIONAL PLANAR STRESS WAVE PROPAGATION

by

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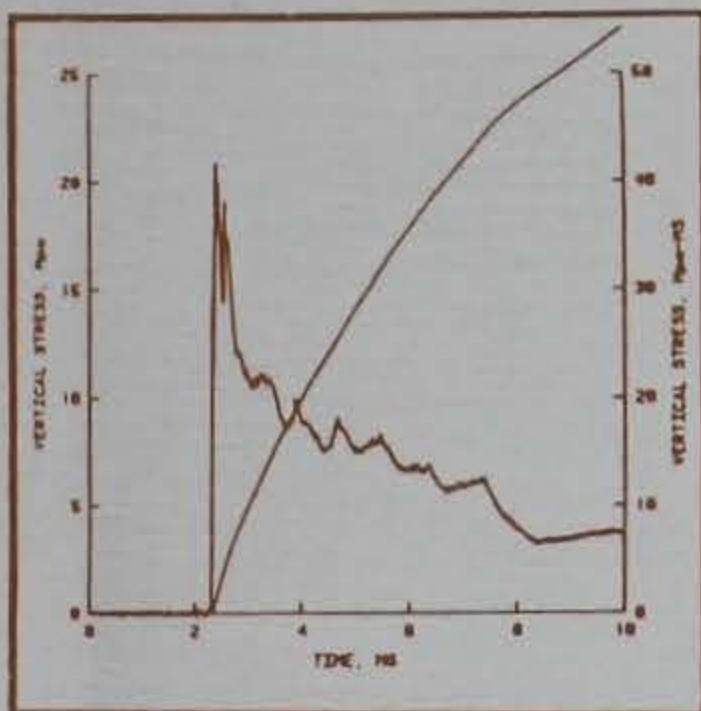
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<p>ONED is a small one-dimensional stress wave propagation computer program for simulating explosive loading on layers of soil. The method of artificial viscosity is used with an explicit computational technique. Materials treated can be solids, porous solids, gases, and explosives. The loading causing the waves can be provided either as a pressure history or by the detonation of an explosive layer. The materials are represented by three-dimensional models with separate treatments for the pressure and deviatoric stresses. The pressure for the porous material is given by loading and unloading curves defined by varying moduli. The deviator stress model describes elastic and plastic behavior with either a Mises or a Coulomb friction yield limit. The detonation of an explosive is provided by either a poly-tropic gas relation or a tabular isentrope.</p> <p>This version of ONED is an amplification of an earlier version used at the US Army Engineer Waterways Experiment Station for many years. The main changes are the conversion to</p> <p style="text-align: right;">(Continued)</p>					
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an explicit computational scheme, use of artificial viscosity, provision for large deformations, addition of explosive detonations, and use of separable material model routines.

PREFACE

The work described herein was performed by SRI International for the US Army Engineer Waterways Experiment Station (WES) under Contract No. DACA39-83-K-0002. It was sponsored by the Office, Chief of Engineers, US Army, as a part of Project 4A162719AT40, Task A0, Work Unit 024, "Ground Shock Prediction Techniques for Earth and Earth-Structure Systems." Technical Monitor for OCE was Mr. R. L. Wight.

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COL Dwayne G. Lee, CE, is Commander and Director of WES. Dr. Robert W. Whalin is Technical Director.

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CONVERSION FACTORS, NON-SI TO SI (METRIC)
UNITS OF MEASUREMENT

Non-SI units of measurement used in this report can be converted to SI (metric) units as follows:

<u>Multiply</u>	<u>By</u>	<u>To Obtain</u>
degrees (angle)	0.01745329	radians
dynes per square centimeter	0.1	pascals
ergs	0.1	microjoules
feet	0.3048	meters
grams per cubic centimeter	1,000	kilograms per cubic meter
inches	25.4	millimeters
kips (force)	4.448222	kilonewtons
kips (force) per square inch	6.894757	megapascals
pounds (force) per square inch	6.894757	kilopascals
pounds (mass)	0.4535924	kilograms
pounds (mass) per cubic foot	16.01846	kilograms per cubic meter

ANALYSIS OF DYNAMIC IN SITU
BACKFILL PROPERTY TESTS

THE MODIFIED ONED PROGRAM FOR ONE-DIMENSIONAL
PLANAR STRESS WAVE PROPAGATION

I INTRODUCTION

The ONED code is a computer program for calculating one-dimensional plane stress wave propagation through a column of soil and other materials in response to detonation of an explosive. The materials may be porous solids, solids, gases, or explosives. The stress waves being computed are initialized by detonation of an explosive or by prescription of a pressure history at the top boundary. Computations are made with the Lagrangian form of the equations of motion so that the coordinates move with the materials. An artificial viscosity is used to smear wave fronts over several computational cells.

A. Background

The present ONED code is a revision of a code that has been in regular use at the U. S. Army Engineer Waterways Experiment Station (WES) for many years. The first version of the ONED code was developed at WES around 1966 for predicting ground motion. In 1971, Radhakrishnan and Rohani documented the code¹ and it has remained in essentially the same form until the present revision.

The earlier version of ONED is a small-strain, small-displacement, finite-element type of program with an implicit solution scheme. Shock fronts were treated by a weak-shock approximation. A single constitutive relation was provided: a variable-modulus, one-dimensional form that closely represented measured uniaxial strain soil properties.

B. New Features

The following modifications were made to produce the current version of ONED:

- (1) The solution procedure was revised to provide an explicit technique, and the time sequencing of the displacement,

velocity, acceleration, and stress calculations were adjusted so that they provide an accurate centering of the finite difference equations for momentum conservation.

- (2) The treatment of strain and density was modified to represent a large displacement formulation.
- (3) The calculation of an artificial viscosity was added with both a quadratic and a linear term in $d\rho/dt$, the rate of change of density with time.
- (4) The subroutines that handle the stress-strain calculation were isolated so that these subroutines can be transported to other organizations and codes and so that other stress-strain routines can be readily added.
- (5) A three-dimensional model for porous material was constructed.

The following models were added:

- (6) Models for deviator stress, a Mie-Grüneisen model for solids, and polytropic and tabular isentrope models for explosives.

In addition some changes were made in the input and output, and the treatment of geostatic stress has been modified.

C. Scope

This report contains a discussion of the theoretical bases for the equations used in ONED, application of the theory to analysis of typical problems, details of implementation, and sample problems. The essential theory for one-dimensional wave propagation is given in Chapter II and the constitutive relations (stress-strain relations) in Chapter III. The

processes associated with initializing the material properties and the cell layout are described in Chapter IV. Sample input files are listed in Appendix A, followed by a glossary of all important parameters in Appendix B. The code is intended for modification and amplification by adding new stress-strain relations: guidelines for inserting new relations are described in Appendix C.

Appendix D is a listing of ONED and its subroutines. Internally the code uses the cgs system of units, but there are provisions for inserting some of the data in the SI or English systems. Some printout can also be obtained in the other systems.

II PROPAGATION CALCULATIONS (Subroutine CMPUTE)

The motion and stresses throughout the material are determined as a function of time in the code. The solution is obtained by integrating the mass and momentum conservation relations together with constitutive relations for the material. This section presents the conservation relations and their general solutions.

In the solution procedure, the material is first divided into discrete units or cells. Velocities, stresses, and other quantities are initialized in cells as required for the particular problem. Then a time step is taken and the motions and stresses are calculated for each cell using the conservation and constitutive relations. This process of stepping forward in time and performing calculations for each cell is repeated until the time has reached the duration of interest. The time step used is controlled by stability and smoothness criteria in the code. The stability considerations are described in this section.

A. Solution Procedure for Wave Propagation Equations

The ONED program is based on the solution of the Lagrangian equations governing one-dimensional planar motion of a continuous medium. The solution technique is called the method of artificial viscosity because of the introduction of viscous forces to permit a continuous-flow computation in regions of high stress gradients. Such regions are interpreted as locations of shock fronts although no discontinuities occur in the computed flow field. With this artificial viscosity method, the equations of continuous flow can be used everywhere, and no special equations are required for shock fronts. ONED uses the leapfrog method of von Neumann and Richtmyer² to integrate the flow equations. The following paragraphs introduce the governing differential equations for planar flow. These equations are changed to an integral form for solution in the program.

The one-dimensional planar Lagrangian differential equations to be solved are

$$\left(\frac{\partial V}{\partial t}\right)_H = -\frac{1}{D_o} \left(\frac{\partial R}{\partial H}\right)_t + g \quad \text{(momentum)} \quad (1)$$

$$\left(\frac{\partial X}{\partial t}\right)_H = V \quad \text{(velocity)} \quad (2)$$

$$\left(\frac{\partial D}{\partial t}\right)_H = -\frac{D^2}{D_o} \left(\frac{\partial V}{\partial X}\right)_t \quad \text{or equivalently} \quad \text{(mass)} \quad (3)$$

$$\left(\frac{\partial X}{\partial H}\right)_t = D_o / D$$

where

H = Lagrangian coordinate location (original position in laboratory coordinates)

X = Eulerian coordinate location (current position in laboratory coordinates)

t = time

V = particle velocity

D, D_o = current and original density

R = total mechanical stress

g = acceleration of gravity.

These equations relate velocity to the coordinate motion and provide for conservation of momentum and mass. In addition to these differential equations, there is an equation of state (or constitutive relation) that is a relationship between stress or pressure quantities and the density, history of loading, and so forth.

$$R = F(D, \dots) \quad \text{(equation of state)}$$

$$= P + \sigma' + Q \quad (4)$$

The total mechanical stress (in the direction of propagation), R , is composed of the pressure P , the deviator stress σ' in the direction of propagation, and an artificial viscous stress, Q .

In the code the four preceding equations are solved simultaneously by dividing the material into small elements. Then the quantities X , V , D , R , and so forth, are evaluated only at the discrete positions and times shown in Figure 1. The coordinate location X is obtained at integral values of j and n , the velocity is at j and $n + 1/2$, and all other quantities pertain to the midcell $(j+1/2, n)$ points. Here the cells are treated as constant strain finite elements (each cell has a constant value of all three principal strains throughout its volume).

The discrete values of the flow quantities are obtained from Eqs. (1) through (3), using the nomenclature of Figure 1. Here it is convenient to solve for quantities in the order D , R , V , and X . The density is obtained from conservation of mass by dividing the stored value of the cell mass, Z , by the thickness. The first form of Eq. (3) is not used here because it can give erroneous results for large density changes; instead, the second form of Eq. (3) is used:

$$D_{j+1/2}^n = \frac{Z_{j+1/2}}{X_{j+1}^n - X_j^n} \quad (5)$$

The velocity is obtained by a discretization of Eq. (1), or equivalently, by using "force equals mass times acceleration" and considering a mass pertaining to the j^{th} coordinate point.

$$V_j^{n+1/2} = V_j^{n-1/2} - \frac{R_{j+1/2}^n - R_{j-1/2}^n}{1/2(Z_{j+1/2} + Z_{j-1/2})} \Delta t^n + g \Delta t^n \quad (6)$$

Here $\Delta t^n = t^{n+1/2} - t^{n-1/2}$ is the time step centered at the n^{th} time.

Finally, the Eulerian position of the coordinate is computed from Eq. (2)

$$x_j^{n+1} = x_j^n + v_j^{n+1/2} \Delta t^{n+1/2} \quad (7)$$

where $\Delta t^{n+1/2} = t^{n+1} - t^n$. The computations proceed from top to bottom (or from left to right in Figure 1), one cell and coordinate at a time, updating the flow quantities to the new time $t^{n+1/2}$ or t^{n+1} , as appropriate. This process is continued until the bottom boundary is reached. Then computations resume at the top for the next time increment.

The foregoing integration method is essentially the leapfrog method of von Neumann and Richtmyer². With this approach, the derivatives in the equations of mass and momentum are correctly centered. That is, each of the conservation relations is replaced by a numerical approximation in which all terms pertain to the same point in time and space. For example, in the momentum Eq. (6), $\partial V / \partial t$ and $\partial R / \partial Z$ are both centered precisely at (n, j) , and therefore, the solution scheme is of second order although no numerical approximations to $\partial^2 V / \partial t^2$ or $\partial^2 R / \partial Z^2$ are needed.

In the code, the names of quantities are essentially those given above in the discretized equations. The coordinate quantities are $VEL(J) = v^{n+1/2}$ and $X(J) = x_j^{n+1}$, and the cell quantities are of the form $RMECH(J) = R_{j+1/2}^n$. The time step is $DT = \Delta t^{n+1/2}$. Hence the coordinate point and the cell to the right are both labeled J , and the midcell quantities at n and the coordinate quantities at $n+1/2$ or $n+1$ are stored in the arrays. Boundaries between materials are treated in the same fashion as coordinates within a material except that an extra coordinate is provided to permit separation of the layers.

B. Artificial Viscous Stress

The artificial viscous stress is required in finite difference wave propagation calculations to smooth shock waves so that the entire flow field can be treated by the conservation equations of continuous flow, Eqs. (1) through (3). The artificial viscous stress (Q) is the difference between the nonequilibrium mechanical stress (R) and the equilibrium thermodynamic stress (σ) given by the constitutive relations. Hence Q represents real stresses occurring in the nonequilibrium states of the shock front. However, the basis for computing Q is artificial, depending on the computational cell size and on viscosity coefficients, which are not related to real physical processes.

In the modified ONED, the usual linear and quadratic viscous stress forms are provided. The linear form is computed by the equation

$$Q = - C_L C_s \rho \Delta V \quad (8)$$

where C_L = dimensionless coefficient of linear artificial viscosity, C_s = sound speed, and $\Delta V = V_{j+1} - V_j$, the velocity difference across the cell. The linear artificial viscosity is similar in form and operation to the standard linear viscosity models used to represent material behavior. However, here the coefficient C_L is chosen to provide enough damping to minimize oscillations in the calculations and not to represent the real material viscosity. Useful values for C_L are in the range of 0.05 to 0.50.

The quadratic artificial viscous stress proposed originally by von Neumann and Richtmyer² has the form

$$Q = C_Q \rho (\Delta V)^2 \quad (9)$$

where C_Q = the dimensionless viscosity coefficient, and $\Delta V = V_{j+1} - V_j$, as before. The quadratic viscous stress is permitted to act only on compressive waves. For normal values of C_Q of 3 or 4, the shock front is rapidly spread over three to four cells and then maintains essentially a

constant thickness as the wave propagates. Because of the quadratic nature of the expression for Q , very little damping occurs outside the shock front. In contrast, the linear viscosity tends to continue to erode the wave fronts as long as they propagate.

Normally, both linear and quadratic artificial viscous stresses are used, so the artificial viscous stress Q is the sum of the linear and quadratic terms from Eqs. (8) and (9). The quadratic viscosity quickly establishes the shock front thickness. The linear viscosity damps the small oscillations that would otherwise occur near the shock front, but is given a small enough coefficient so that the wave front is not seriously eroded.

C. Time-Step Control

For the calculations to proceed in a stable manner, the time increment between cycles must be kept smaller than that given by the Courant-Friedrichs-Lewy condition (see Reference 3, p. 262). This criterion is simply

$$\Delta t \leq \frac{\Delta X}{C_e} \quad (10)$$

where ΔX is the cell size and C_e is the local effective sound speed (defined later).

The criterion means that the time step cannot be so large that the new points are outside the characteristic domain of dependence of the previous points. Referring to Figure 1, the new point $(n+1, j-1)$, for which the variables are computed from values at $(n, j-2)$, $(n, j-1)$, and (n, j) , must lie within the domain of dependence or range of waves from those points. This domain is contained between lines with speeds of C_e . A physical interpretation of the requirement is that a wavelet cannot be allowed to proceed from one coordinate point to beyond another in one time step because this would allow a material point to be affected by conditions at material points outside the true domain of dependence.

This simple criterion is modified to provide for added safety (the step used in ONED is 80% of the time step at the limit of stability) and to allow for the effect of artificial viscosity.

Artificial viscosity stiffens the material and therefore increases the apparent sound speed, reducing the allowable time step. For linear and quadratic viscosity coefficients (C_L and C_Q), Herrmann et al. (Reference 4, p. 37) derived a reduction factor to be applied to the time step. As shown in Reference 5, the effect of the artificial viscosity can be accounted for by computing an effective sound speed C_e and effective modulus M_e as follows:

$$C_e^2 = \frac{M_e}{\rho} = \frac{\Delta P + 2Q}{\Delta \rho} + \frac{4G}{3\rho} \quad (11)$$

where ΔP and $\Delta \rho$ are the changes in pressure and density during the current time step, and G is the shear modulus.

III CONSTITUTIVE RELATIONS

The constitutive relations provide the stress as a function of density, strains, and other quantities. This section describes the common constitutive relations and outlines the available constitutive models.

In the standard constitutive relations, the stress tensor is separated into a pressure and a stress deviator tensor. The pressure is the average stress

$$P = 1/3 \sum_i \sigma_{ii} \quad (12)$$

and the stress deviator elements are

$$\sigma'_{ij} = \sigma_{ij} - P \delta_{ij} \quad (13)$$

where σ_{ij} are stress tensor elements and δ_{ij} is the Kronecker delta. The pressure is usually presented as a function of density and internal energy. The deviator stress is calculated by elastic-plastic relations.

A. Pressure Model for a Solid (Subroutine EQST)

The pressure is computed from a pressure density relation for each material. The pressure-density relation used here is the Hugoniot for shock compaction given as the following series:

$$P_H = C\mu + D\mu^2 + S\mu^3 \quad (14)$$

where

$$\mu = \frac{\rho}{\rho_0} - 1$$

C = bulk modulus

D, S = coefficients with the units of moduli

ρ, ρ_0 = current and initial densities.

For the present calculations, we are neglecting the internal energy changes occurring in shock compaction because ONED is intended only for stresses up to a few kilobars. Thus, the Hugoniot is also being used as the unloading isentrope.

B. Deviator Stress Model (a Part of Subroutine CMPUTE)

The deviator stress is the part of the stress tensor that arises because of the resistance of the material to shearing deformation. In ONED, the standard model for deviator stresses accounts for elastic response and plastic flow. The yield strength that governs plastic flow can be either of the Mises or Coulomb types. First some general definitions are given and then the equations that are special to each yield type.

The elastic relations between stress and strain are cast in the following form:

$$\sigma'_{ij} = 2G(\epsilon_{ij}^E - \frac{\delta_{ij}}{3} \sum_{\ell} \epsilon_{\ell\ell}^E) \quad (15)$$

$$P = C \sum_{ii} \epsilon_{ii}^E \quad (16)$$

Here, σ'_{ij} and ϵ_{ij}^E are the deviatoric stress and elastic strain in the ij direction, G is the shear modulus, δ_{ij} is the Kronecker delta, P is pressure, and C is the bulk modulus. For the elastic case, $\epsilon_{ij} = \epsilon_{ij}^E$, all the strain is elastic. However, Eqs. (15) and (16) are also applicable to the plastic case where the strain increments are separated into elastic and plastic components:

$$d\epsilon_{ij} = d\epsilon_{ij}^E + d\epsilon_{ij}^P \quad (17)$$

where $d\epsilon_{ij}$ is the total strain increment and $d\epsilon_{ij}^p$ is the plastic strain increment. For convenience, the terms in the parentheses of Eq. (15) can be named a deviator strain, defined as follows:

$$\epsilon_{ij}^E = \epsilon_{ij}^E - \frac{\delta_{ij}}{3} \sum_l \epsilon_{ll}^E \quad (18)$$

Then Eq. (15) becomes

$$\sigma_{ij}' = 2G \epsilon_{ij}^E \quad (19)$$

The Mises or Reuss plasticity relations or "incremental plasticity with an associated flow rule" are considered here first⁶. Modifications to treat Coulomb friction are described later. Yield occurs when the effective stress reaches the yield strength. The effective stress is

$$\bar{\sigma} = \sqrt{\frac{3}{2} (\sigma_{ij}' \sigma_{ij}')} \quad (20)$$

where the repeated subscripts indicate summation. The yield criterion is

$$\bar{\sigma} = Y \quad (21)$$

where Y is the current yield strength. The Reuss flow rule indicates that the deviator stress in any direction is proportional to the plastic deviator strain in that direction (because the Reuss material sustains no plastic volume strain $\sum d\epsilon_{ii}^p = 0$). Hence,:

$$d\epsilon_{ij}^p = \lambda \sigma_{ij}' \quad (22)$$

where λ is a proportionality constant. Now we define a scalar effective plastic strain quantity as follows:

$$d\bar{\epsilon}^p = \sqrt{\frac{2}{3} d\epsilon_{ij}^p d\epsilon_{ij}^p} \quad (23)$$

As before, the repeated subscripts indicate summation. Now we square Eq. (22) and make use of the definitions of $\bar{\sigma}$ and $d\bar{\epsilon}^p$. Then

$$d\bar{\epsilon}^P = \frac{2}{3} \bar{\sigma} d\lambda \quad (24)$$

Combining this definition with Eq. (22), we find that

$$d\epsilon_{ij}^P = \sigma'_{ij} \frac{3d\bar{\epsilon}^P}{2\bar{\sigma}} \quad (25)$$

To obtain a solution for an increment of strain, we compute first the stress that would occur if the strain were entirely elastic, that is,

$$\sigma'_{ij}^N = 2G (\epsilon_{ij0}^E + \Delta\epsilon'_{ij}) = 2G (\epsilon_{ij}^E + \Delta\epsilon_{ij}^P) \quad (26)$$

where

ϵ_{ij0}^E = the elastic deviator up to the current strain step

$\Delta\epsilon'_{ij}$ = the total deviator strain increment

ϵ_{ij}^E = the elastic deviator strain after the current increment

$\Delta\epsilon_{ij}^P$ = the plastic strain increment.

The second equality in Eq. (26) is obtained by using Eq. (17) to decompose $\Delta\epsilon'_{ij}$ and by adding $\epsilon_{ij0}^E + \Delta\epsilon_{ij}^E$ to obtain ϵ_{ij}^E . Quantities ϵ_{ij}^E and $\Delta\epsilon_{ij}^P$ can both be replaced by stress quantities through the use of Eqs. (19) and (25). Then,

$$\sigma'_{ij}^N = \sigma'_{ij} (1 + 3G\Delta\bar{\epsilon}^P/\bar{\sigma}) \quad (27)$$

If both sides of Eq. (27) are squared and a quantity $\bar{\sigma}^N$ is introduced in analogy to the definition of $\bar{\sigma}$, then we obtain

$$\bar{\sigma}^N = \bar{\sigma}(1 + 3G\Delta\bar{\epsilon}^P/\bar{\sigma}) \quad (28)$$

Here, $\bar{\sigma} = Y$. Combining Eqs. (27) and (28) yields a solution for σ'_{ij} :

$$\sigma'_{ij} = \sigma'_{ij}^N \frac{\bar{\sigma}}{\bar{\sigma}^N} \quad (29)$$

Then, the elastic strain can be obtained from Eq. (19) and the effective plastic strain from Eq. (28):

$$\Delta \epsilon^P = \frac{\bar{\sigma}^N - \bar{\sigma}}{3G} \quad (30)$$

Finally, each component of plastic strain is found from Eq. (22) using $d\lambda$ obtained from Eq. (24).

The preceding process can be adapted to hardening plasticity where Y is not a constant. The equations are appropriate for steps from one plastic Mises state to another or from an elastic state to a plastic state.

The Coulomb plasticity model is quite different from the preceding Mises model. The yield strength is a function of normal stress on the plane of yield (a Tresca formulation); hence, the pressure as well as the deviator stresses are involved in yielding. No changes in pressure or volume occur during yielding so this is a Coulomb-without-dilatation model and hence, the flow law is non-associated. The fundamental Coulomb relation provides a shear yield stress τ_c as a function of cohesion c , normal stress σ_N , and the angle of internal friction ϕ :

$$\tau_c = c + \sigma_N \tan \phi \quad (31)$$

Following Terzaghi,⁷ this expression is transformed to

$$\sigma_1 = 2c \sqrt{N_\phi} + \sigma_3 N_\phi \quad (32)$$

where $N_\phi = \tan^2(45^\circ + \phi/2)$, σ_1 is the stress in the direction of propagation, and $\sigma_2 = \sigma_3$ is the lateral stress. A second yield condition, with σ_1 and σ_3 interchanged in Eq. (32), is obtained for the case where σ_3 is the most compressive stress. To solve for the yield limit on σ'_1 (σ'_{1y}), we replace σ_1 and σ_3 by deviator stresses and pressure and use Eqs. (12) and (13) to obtain

$$\sigma'_1 + \sigma'_2 + \sigma'_3 = 0 \quad (\sigma_2 = \sigma_3)$$

$$\sigma'_3 = -\frac{1}{2} \sigma'_1 \quad (33)$$

Using Eq. (33) in Eq. (32), we obtain the yield limit

$$\sigma'_{1y} = \frac{2c \sqrt{N_\phi} + P(N_\phi - 1)}{1 + N_\phi/2} \quad (34)$$

for σ_1 more compressive than σ_3 , and

$$\sigma'_{1y} = -\frac{2c \sqrt{N_\phi} + P(N_\phi - 1)}{1/2 + N_\phi} \quad (35)$$

for σ_3 more compressive than σ_1 .

The input quantities for the yield model are the yield strength (Y) for the Mises model or $2c$ and $\tan \phi$ for the Coulomb model. Because of the different natures of the two models, they cannot be combined. The one that is to be used is determined in the code by whether $\tan \phi$ is nonzero. The deviator stress model is used with either solid or porous models for pressure.

C. Model for Porous Material (Subroutine SSONED)

For simulating the soil response to wave propagation and representation of both vertical and horizontal stress histories, it is necessary to have a multidimensional soil model. As a preliminary step toward such a model, we developed a simple isotropic model with separate treatments for the pressure and deviatoric stresses. The deviatoric stresses follow the standard elastic-plastic behavior with a Coulomb yield criterion, as described above. The pressure follows the segmented loading and unloading paths given in the original version of ONED.¹ A possible pressure-strain diagram for the SSONED model is shown in Figure 2. In the figure, SL's are volume strains and EL's are bulk moduli.

During loading, the pressure is computed from the relation

$$P = P_i + E_i (\epsilon - \epsilon_i) \quad (36)$$

where

$P_i = P(i, M)$ = pressure at i^{th} node of the loading diagram

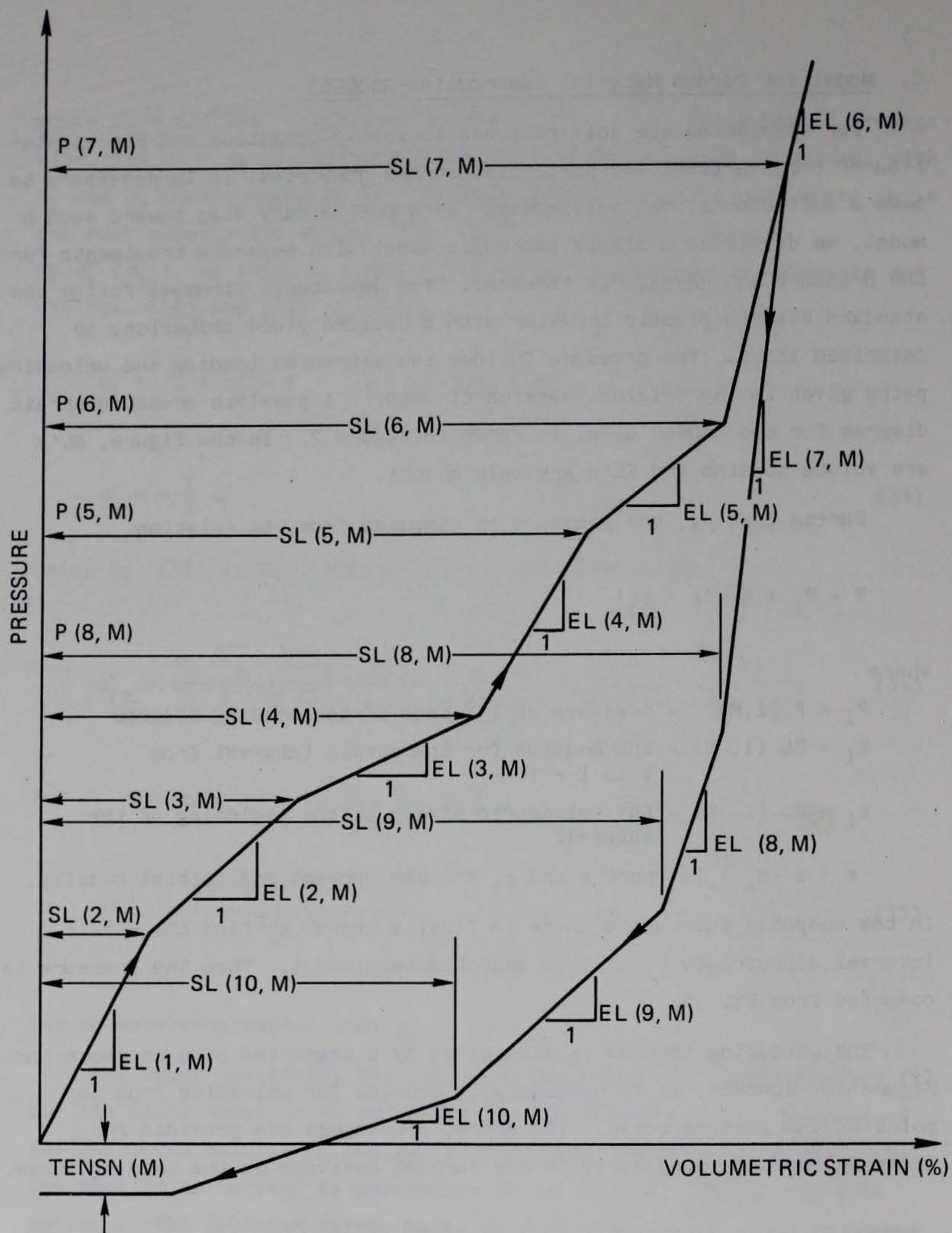
$E_i = EL(i, M)$ = the modulus for the strain interval from i to $i + 1$

$\epsilon_i = SL(i, M)$ = the volumetric strain at the beginning of the interval

$\epsilon = \rho / \rho_0 - 1$, where ρ and ρ_0 are the current and initial density.

In the computer subroutine there is first a search to find the strain interval appropriate to ϵ ; this search determines i . Then the pressure is computed from Eq. 36.

The unloading process is also given by a segmented path as shown in Figure 2. However, it is necessary to provide for unloading from any point on the loading curve. Therefore, procedures are provided for shifting the unloading curve to any current position on the loading curve.



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FIGURE 2 DEFINITION OF MODULUS AND STRAIN VALUES ON LOADING AND UNLOADING CURVES FOR THE SSORED MODEL

Three possibilities have been provided for prescribing the unloading curve:

- (1) The unloading curve is simply shifted horizontally to the strain at which the unloading begins.
- (2) The stresses on the unloading curve are reduced in proportion to the stress at which unloading begins, but the unloading moduli are unchanged. Then the curve is shifted to the strain at which the unloading begins.
- (3) The stresses on the unloading curve are reduced as in version 2, but the moduli are modified to preserve the strain increment amplitudes. Then the curve is shifted to the strain at which the unloading begins.

These three unloading processes, illustrated in Figure 3, are provided in the current model to facilitate testing of a variety of stress-strain possibilities.

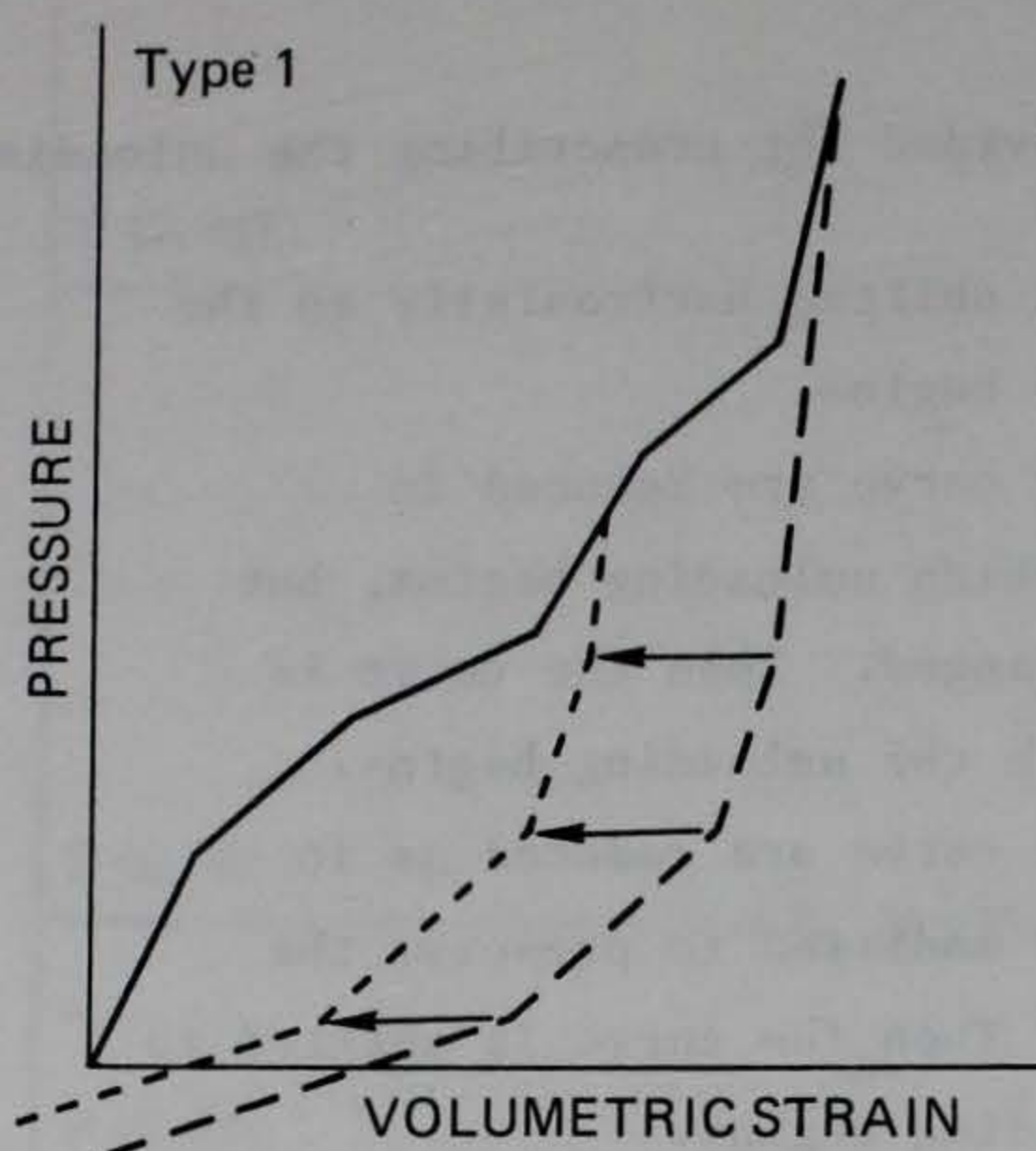
For all three unloading paths, we use the following expression for pressure:

$$P = P_i + E_i (\epsilon_Q - \epsilon_i) \quad (37)$$

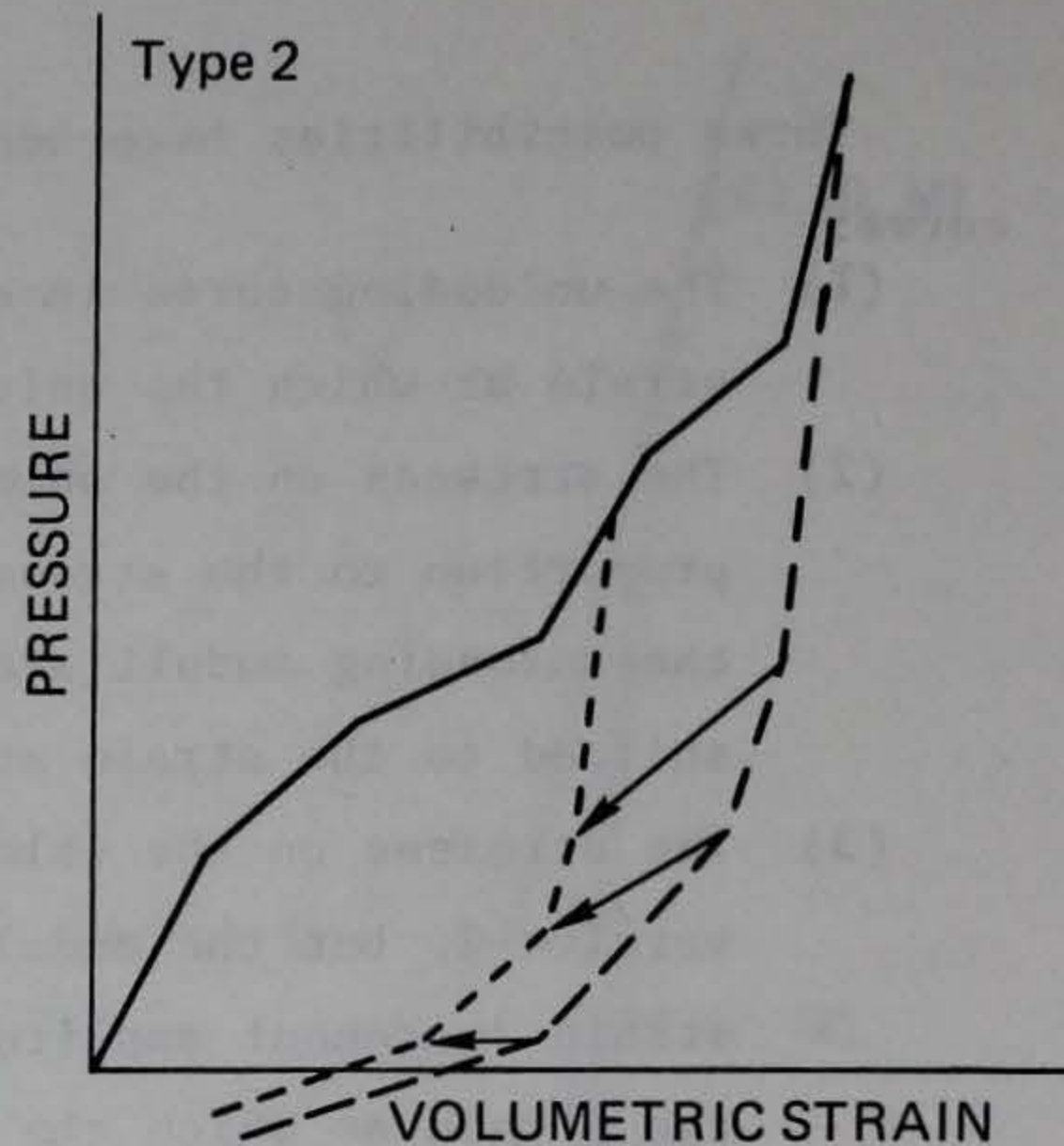
where ϵ_Q is a pseudo strain and the other quantities are the same as those in Eq. 36. The next task is to determine ϵ_Q for all three loading types.

For type 1 unloading, we see in Figure 3a that the strain at a pressure of P_{\max} on the unloading curve is

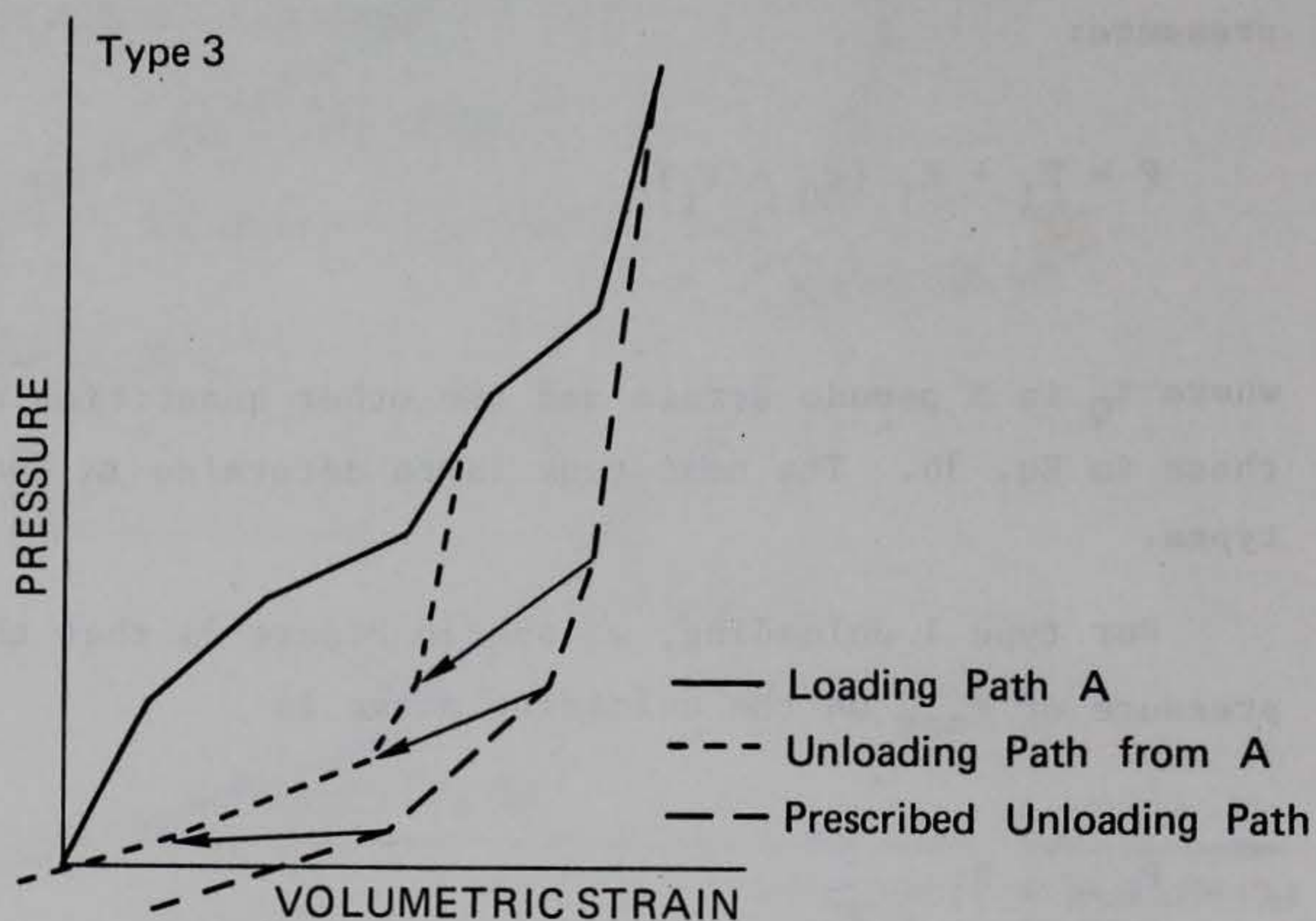
$$\frac{P_{\max} - P_i}{E_i} + \epsilon_i \quad (38)$$



(a) Shift Only



(b) Stresses Reduced Proportionately,
Moduli Unchanged



(c) Stresses Reduced Proportionately,
Strain Increments Unchanged

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FIGURE 3 THREE DESCRIPTIONS OF THE UNLOADING PROCESS PROVIDED BY THE
SSONED MODEL

where P_{\max} is the highest pressure reached before the current unloading and P_i is the pressure at an unloading node just above P_{\max} . The strain at P_{\max} is ϵ_{\max} . Then the appropriate pseudo strain is

$$\epsilon_Q = \epsilon + \frac{P_{\max} - P_i}{E_i} + \epsilon_i - \epsilon_{\max} \quad (39)$$

This expression is also satisfactory for unloading from pressures above the provided peak and for continuing into tension.

For type 2 unloading the peak point ($P_k \epsilon_k$) of the provided loading curve governs. The unloading curve is constructed by reducing the pressures at nodes and the strain increments by the ratio P_{\max}/P_k . The pseudo strain ϵ_Q is readily obtained by computing the incremental reduction from ϵ_k :

$$\epsilon_Q = \epsilon_k + \frac{P_k}{P_{\max}} (\epsilon - \epsilon_{\max}) \quad (40)$$

where ϵ_k is the volumetric strain at the peak point of the provided loading curve.

With type 3 unloading the strain increment sizes are preserved. Therefore, the expression is like that in Eq. 40, but without the ratio factor

$$\epsilon_Q = \epsilon_k + \epsilon - \epsilon_{\max} \quad (41)$$

With these special definitions of the pseudo strain, all three unloading types can be treated in a common fashion.

It is advisable to plot the loading curve and unloading curves from several nodes to decide whether the behavior is reasonable. Under some circumstances the unloading curves can cross the loading curves or otherwise provide for physically unreasonable behavior.

D. Model for an Explosive (Subroutine EXPLODE)

At the time of detonation, the chemical energy of an explosive is transformed into internal energy. This energy places the state for the material at a point on the equation-of-state surface termed the "constant-volume explosion" state. Subsequent motions of the material usually allow it to expand along the unloading isentrope that passes through the constant-volume explosion point.

For the ONED calculations, a very simple procedure is available for representing explosions. Only the isentrope must be provided. Then, at the beginning of the calculation, the pressure in the explosive jumps to the constant-volume explosion point on the isentrope. Subsequent rarefaction waves in the explosive allow the pressure to reduce along the isentrope in a natural fashion. The isentrope can be provided in two ways: with the analytical form for a polytropic gas or as a series of pressure-volume points.

The polytropic gas has the following expression for an isentrope:

$$PV^\gamma = \text{constant} = P_{CJ} V_{CJ}^\gamma \quad (42)$$

where γ is the gas constant $= \Gamma + 1$, and P_{CJ} and V_{CJ} are the pressure and specific volume at the Chapman-Jouguet point. The input quantities used to define the pressure-volume isentrope are the chemical energy Q and the Gruneisen ratio Γ . From these two quantities and the initial density ρ_0 , the detonation velocity D_x and quantities at the C-J point can all be determined.

$$D_x = \sqrt{2Q(\gamma + 1)(\gamma - 1)} \quad (43)$$

$$P_{CJ} = 2Q(\gamma - 1) \rho_0 \quad (44)$$

$$V_{CJ} = \frac{\gamma}{\rho_o(\gamma + 1)} \quad (45)$$

$$E_{CJ} = \frac{2Q\gamma}{\gamma + 1} \quad (46)$$

$$u_{CJ} = \sqrt{\frac{2Q(\gamma - 1)}{\gamma + 1}} \quad (47)$$

where E_{CJ} and u_{CJ} are internal energy and particle velocity, respectively. The quantities P_{CJ} , V_{CJ} , and D_x are computed in the subroutine EXPLODE during initialization. Then Eq. (42) is used to determine the pressure during calls by CMPUTE. The parameters of the C-J point are convenient for characterizing the explosives because these data are usually available; however, the C-J state is not normally reached in constant-volume explosion calculations.

The sound speed on the isentrope is given by

$$C = \sqrt{\gamma P / \rho} \quad (48)$$

The sound speed is computed at each call to aid in the time-step calculation.

IV INITIALIZATION

The READIT subroutine is called once at the beginning of each problem to read in all the data and initialize the COMMON storage. The sequence of major operations conducted by this subroutine is

- Call ZERO to zero the main arrays.
- Read general running instructions for the problem.
- Read properties for each material.
- Lay out a coordinate grid over all the materials.
- Initialize the coordinate and cell arrays, including the geostatic stress.
- Print initial coordinate and cell values.

A sample input file is shown in Figure 4. All the input quantities are provided by the user in free-field format, except for the first three lines of comments and the title. The input quantities are generally in English units; lengths in feet or inches, density in pounds per cubic foot, pressure in psi, and time in seconds. The exception is the Q or chemical energy of explosives in erg/g. The input quantities are explained in the next subsections, and some guidance for selecting values is provided. Please refer to the Glossary (Appendix B) for definitions of the terms.

A. Input of General Control Information

The first group of data identifies the computation and controls the type of simulation, the amount of printing and plotting, and the numbers of layers of materials.

The first three lines describe the computation. The first two lines should begin after column 5. The third line contains an identifying

FIRST TRIAL OF MODIFIED ONED WITH DATA, USING ZELASKO'S INPUT FOR DISKO 2, MODIFIED FORM TO FIT THE SRI ONED.			Comments and Title
1 DISKO 2 FOR WES, ONED TRIAL WITH AN EXPLOSIVE LAYER			
998, 0, 0.0035, 10, 1.	NOCOMP, BOTTOM BOUNDARY INDICATOR, TEND, NSTOP, GACC		General Running Controls
6, 0, 0, 0, 5, 0,	NLAYRS, MISP, NSP, MATTEN, NMIRLS, NFORCE		
11, 1, 3, 0, 0, 0, 0, 0, 1, 1, 1, 1, 0.005	NNODE, KNTROL (7), NHIST, MVT, NSV, KVEL, TND		
1.333, 2.9, 3.3, 3.8, 3.9, 4.0, 4.44, 5.44, 6.44, 7.44, 8.44	DEPTHS FOR HIST.		
1000., 0.0, 60., 0., 0., 0., 0.	SK (7)		
'CLAY', 128., 0.05, 4., 0., 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR CLAY			Clay Properties
0., 0., 0.	SHMOD, YIELD, TANPHI		
3, 1, 'MODU', 'STRA', 6.914627E4, 1.	KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT		
39750., 993750., 993750.	CLAY BERM MODEL, MODULI		
100., 0., STRAINS			
'PLYWUD', 42., 0.05, 4., 0., 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR			Plywood Properties
0., 0., 0.	SHMOD, YIELD, TANPHI		
8, 1, 'MODU', 'STRA', 6.914627E4, 1.	KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT		
65000., 16000., 8000., 18000., 21500., 30000., 2.E5, 2.E5, PLYWOOD, MODULI			
1.0, 2.5, 7., 9.5, 12.75, 100., 0.0,	STRAINS		
'FOAM', 1.25, 0.05, 4., 1.5, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR FOAM			Polyurethane Foam Properties
0., 0., 0.	SHMOD, YIELD, TANPHI		
10, 1, 'MODU', 'STRA', 6.914627E4, 1.	KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT		
72.73, 1891., 41., 1654., 7.5E4, 7.5E4, 75000., 40000., 14000., 14000., FOAM, MOD			
55., 92.0, 3260., 4892., 5000., 5100., 5064., 5058., 5052., STRAINS.			
'SAND', 106.3, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR SAND			Sand Properties
1.36E+9, 1.0E+4, 0.274	SHMOD, YIELD, TANPHI		
7, 3, 'P', 'D', 1., 1.0	SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT		
4.E7, 7.E7, 1.7E8, 4.5E8, 1.E9, 0., FLUME SAND, P, DYN/CM ² (K=2.92E9)			
1.7268, 1.7458, 1.793, 1.893, 2.065, 2.023	DENSITIES, G/CM ³		
'EXPL1', 2.7406, 0.05, 4., 0.0, 0, 1, 0, 0 GAM, DAMP, TENS, POR, PR, SOL, VAR (EXPL)			Explosive
0., 0., 0.	SHMOD, YIELD, TANPHI		
'POLY', 4.5E10, 0.2	Q AND GAMMA		
'OVERBDN', 95.035, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, NVAR			Overburden Soil
1.36E+9, 1.0E+4, 0.274	SHMOD, YIELD, TANPHI		
10, 1, 'P', 'D', 1.E7, 1.0	SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT		
8., 39., 69., 122., 230., 400., 100., 30., 0., SAND OVERBDN MODEL, PRESS., MPA			
1.603, 1.792, 1.904, 2.031, 2.176, 2.343, 2.290, 2.256, 2.207	DENSITIES		
2.6667, 3.0, 0, 6	SAND OVERBURDEN		Layers and Cells
3.0417, 1.0, 0, 5	EXPLOSIVE		
3.7917, 1.0, 0, 1	CLAY LAYER DB, SPL, NSP, M		
3.8547, 1.0, 0, 2	PLYWOOD LAYER		
3.9387, 0.5, 0, 3	FOAM		
9.9387, 1.0, 0, 4	SAND TEST BED		

Notes: The blank lines between materials were introduced for readability.
They must not appear in the input file.

FIGURE 4 SAMPLE INPUT FOR AN ONED SIMULATION OF WAVE PROPAGATION
THROUGH LAYERS OF CLAY, PLYWOOD, FOAM, AND SAND (DISKO-2)

number in column 5 and is alphanumeric thereafter. This third line is printed as a title with the remainder of the output.

The fourth line contains the parameters NOCOMP, LB, TEND, NSTOP, and GACC. NOCOMP controls the plotting process. For a calculation with NOCOMP greater than or equal to 999, the input stops and the results from a previous run are plotted. LB gives the boundary condition at the bottom of the last layer. LB is zero for a free boundary and 1 for a rigid boundary: neither condition represents the continuum of material below a test region. Therefore, sufficient distance should be provided below the region of interest so that no waves reflected from the bottom can return to the region of interest during the calculation time. TEND is the termination time of the calculation and NSTOP is the total number of time increments. The program halts when either of these stop parameters is reached. GACC is an indicator for the geostatic stress; a nonzero value causes initialization of a geostatic stress and use of a gravity loading throughout the computation.

The fifth line contains NLAYRS, MISP, NSP, MATTEN, NMTRLS, and NFORCE; these control layering, materials, and loading. NLAYRS is the number of layers, and MISP and NSP are indicators controlling the method of computing the cell sizes in each layer. MATTEN is an indicator that can allow ONED to approximate spherical attenuation and dispersion. NMTRLS is the number of materials; the number of materials need not correspond with the number of layers. NFORCE is the number of pressure-time points in the loading history; NFORCE = 0 is used for an explosive loading in which the explosion process is simulated.

The sixth line provides a series of plot and print controls: NNODE, KNTROL(7), NHIST, MVT, NSV, KVEL, and TNDPLT. NNODE is the number of nodes or cells at which histories are required, and KNTROL (7 parameters) controls the types of plots generated for each of those nodes. NHIST is the frequency for listing the array values at these nodes (the listing occurs every NHIST time steps). See Appendix B for KVEL and TNDPLT.

The seventh line contains the depths (in feet) at which the plots and printouts are required. There are NNODE of these depths.

The eighth line contains the scale factors SK(7) for the requested plots.

B. Material Property Input

In the present version of ONED, the information for the materials and for the layers is inserted separately. Therefore, although a material appears in several layers, it is only necessary to insert its properties once. Conversely, a material may appear in the data input but not be used for any layer.

For each material two lines of data are read by READIT. The first of these lines gives NAME, GAM, DAMPL, DAMPQ, TENSX, NPOR, NPR, NSOL, and NVAR. NAME is an eight-letter alphanumeric identifier for the material and GAM is its initial density in pounds per cubic foot. DAMPL is the coefficient of linear artificial viscosity. A value of 0.05 is appropriate for an elastic solid, but larger values, such as 0.15 to 0.5, should be used for porous materials. DAMPQ is the coefficient of quadratic artificial viscosity. A value of 4 is used for solids, but values up to 100 can be used for porous materials. The appropriate values of the viscous coefficients for each material and stress level are determined by trial calculations. The "best" values of DAMPL and DAMPQ allow the calculation to meet two criteria:

- Slight oscillations occur in the stress and velocity histories near the shock fronts and do not occur elsewhere.
- During a shock loading at a point in the material, the series of total stress, specific volume (RMECH,1/DENS) points from the initial state to the shocked state approximate a straight line in stress-volume space.

Excessive damping eliminates oscillations, increases the rise time of shock fronts, and causes a stress-volume curve that is convex upward. Too little damping will cause serious oscillations in the histories and may even cause cell distortions that can halt the calculation.

TENSX is the tensile strength. For TENSX set to zero, no strength criterion is used (i.e., infinite tension stress is allowed).

NPOR, NPR, and NSOL are a set of indicators for the type of material model to be used. Currently, only one material model of each type is provided:

NPOR = 1: porous model, subroutine SSONED

NPR = 1: explosive model, subroutine EXPLODE

NSOL = 1: solid model, subroutine EQST.

As more models are generated and inserted, they can be designated by NPOR = 2, NPR = 2, and so forth. Separate designators are used for each type of model so that combined models can be used. A combined model could describe a porous material that compacts to a solid or an explosive that begins as a solid.

NVAR is the number of extra variables provided for each cell of the material. The extra variables are stored in the COM array. The first extra variable for the J^{th} cell is at location $L = \text{LVAR}(J)$; that is, at $\text{COM}(L)$. Of the current models, SSONED requires two extra variables and the others require none. If histories of nonarray quantities are desired for any model, a simple coding change can allow storage of these in the COM array for later printing and plotting.

The second line read by READIT contains the shear modulus SHMOD, yield strength YIELD, and $\tan \phi$ TANPHI. These parameters control the deviator stress model used. SHMOD and YIELD are input in dyn/cm^2 .

The remainder of the material properties for each model are read by that model subroutine.

1. Input for the Porous Model SSONED

The input required for the porous model SSONED defines the pressure-volume relations to be used for loading and unloading. Three lines or more are read to provide the necessary properties.

The first line contains a series of control parameters needed for interpreting the remaining input: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT. KFIN is the number of pressure-strain segments used to define the loading and unloading curves. The pressure-strain curves can be entered

using pressures or moduli, and strain or density. In each case only KFIN-1 values are required: the first pressure and strain values are taken to be zero, the initial density is GAM, and there are only KFIN-1 intervals for moduli.

NTYPE gives the type of unloading process as described in Section III and shown in Figure 3.

The indicators LPRESS, LDENS, PUNIT, and DUNIT control the interpretation of the pressure-strain values read in next. LPRESS and LDENS are labels with the possible values "D", "M", "P", and "S"; these indicate whether density, modulus, pressure, or strain quantities will be read in. PUNIT and DUNIT are conversion factors to transform pressures and moduli to dyn/cm^2 and densities to g/cm^3 .

The next lines contain the values defining the pressure-strain path during loading and unloading for the porous material. The meaning of these quantities is indicated in Figure 3.

2. Input for the Explosive Model EXPLODE

The model for an explosive can be prescribed in two ways. In the first method the explosive products are a polytropic gas specified by the standard polytropic gas equation. The only required input quantities are Q and Γ . All other properties of the material can be found from Eqs. 42 through 48. The single line read by EXPLODE contains LABEL, Q , and Γ . LABEL is an alphanumeric label of three letters.

The second means of specifying an explosive is to provide a pressure-volume series to define the unloading isentrope for the explosive products. To indicate the use of the pressure-volume series, the LABEL parameter is 'TAB' for a tabular equation of state. In this case Q and Γ have no meaning. Following this line, are a series of lines read by EOSTAB (the tabular equation-of-state subroutine), which is called by EXPLODE.

The first line read by EOSTAB contains IMAX, LABEL, and TYPE. IMAX is the number of pressure-volume (or pressure-density) values in the

series defining the isentrope. LABEL is an alphanumeric label indicating whether densities or volumes are being presented. The default value is 'DENSITY'. If volumes are used, read in 'VOLUME'. TYPE determines whether the interpolation between successive points in the series is linear or logarithmic. Logarithmic interpolation should be used, so TYPE should be set to 'LOG'.

Subsequent lines of the input for EOSTAB contain pairs of density (or volume) and pressure. The units for these are g/cm^3 (or cm^3/g) and dyn/cm^2 because properties for explosives are usually given in these units. The points can be inserted in order of ascending or descending pressure; the subroutine determines which order is being used.

3. Input for the Solid Model, EQST

The solid model EQST requires parameters to define the Hugoniot curve. These quantities are RHO, EQSTC, EQSTD, and EQSTS, which are given in a single line. These correspond to ρ_0 , C, D, and S of Eq. (14).

The EXPLODE model can also be used to represent a nonexplosive gas, such as air. Figure A-2 shows a sample of this treatment. The Q is simply chosen so that the pressure from Eq. (42) is the desired ambient value.

C. Layering

The dimensions and sequencing of the layers and cells are provided next in a series of lines, one line per layer. NLAYRS, the total number of layers including gaps, was read in under the general control information. For each of these layers the user specifies the layer thickness, the material of the layer, and the method for dividing the layer into cells. The layer thickness is provided by the depth of the bottom of the layer in feet. The material number for the layer is M, meaning that it was the M^{th} material described. For a gap, $M = 0$.

The cell sizes are given by a combination of three parameters: MISPL, NSP, and SPL. Uniform cell sizes can be specified in two ways:

- (1) MISPL = 0, SPL = cell size in inches.
- (2) MISPL = 2, NSP = the number of equal size cells.

If nonuniform cell sizes are desired, MISPL can be set to 1 and NSP to the number of cells. Then the cell size for each cell is read into the SP() array.

For accurate calculations the cells should be laid out with care. The following rules are suggested as means for obtaining good results:

- A single cell can be used to represent the presence of a material, but it can provide no information about the wave shape within that material and very little information on the travel time through the material. Five cells in a material can define the presence of the material and its dimensions, but give little information about the wave reverberations within that material. Twenty cells will provide the reverberations also in the material. For judging the appropriate cell sizes and numbers for a problem, it is often necessary to perform a preliminary calculation and examine the resulting wave shapes. For more sharply defined wave fronts, the number of cells should be increased.

- For porous materials, more cells should be provided than would normally be necessary for a solid material.

- Rise times of stress waves are equal to about four traverse times for the cells. Hence, the definition of the stress history can be used as a basis for defining acceptable cell sizes.

- The cell sizes in adjacent layers should be matched so that the propagation times across cells are approximately the same. This propagation time is given by

$$\Delta T = \Delta X / C \quad (49)$$

where ΔX is the cell thickness and C is the nominal sound speed. For a porous material, many possible definitions are available for the sound speed. Here the appropriate value of C is that associated with the main compaction wave. Hence, given the stress-strain relation and an estimate of the peak stress, we can determine the location of the Rayleigh line. From this line, we can find the sound speed C .

$$C^2 = \frac{P/\rho_o}{1 - \rho_o/\rho} \quad (50)$$

In some cases of large compaction it may be necessary to perform a preliminary calculation and revise the cell sizing based on the results of this first calculation.

For nonuniform cells, the sizes should not be allowed to change more than about 5% between cells within a single material.

D. Geostatic Stress Initialization

In the revised ONED the geostatic stress is initialized in detail and the gravity effect is maintained throughout the calculation. The geostatic effect is indicated by providing a nonzero value of GACC. Then READIT computes the overburden for each cell at its center, assuming that X increases in the downward direction. With the overburden stress GEOSTR, READIT computes the strain caused in the cell to reach the overburden stress in the cell. For computing the geostatic stress, READIT calls the various stress-strain subroutines. The strain results in a changed density. The changed density modifies the overburden so the geostatic stress is determined by iteration in each cell. The number of iteration trials is output by the code (see next section). Thus, the initialized material shows a gradually increasing density with depth. The geostatic stress is presumed to affect the pressure only and not the deviatoric stress.

V PRINTED OUTPUT: READIT, HISTRY, SCRIBE, CMPUTE, AND VALMAX

Several types of printed output are provided during and at the conclusion of a calculation. During the reading of the input, the input values are printed by READIT with some additional comments. The material property subroutines read their own input and provide printout. After the input is read, a layout listing is given by READIT. During the calculation, several listings of the layout with current cell variables are made by HISTRY (on a call from CMPUTE) and from CMPUTE. A VALMAX listing of maximum values is made at the end of the calculation. The SCRIBE subroutine is called by ONED to print historical listings of all requested variables.

Samples of the output are presented for simulations of the DISKO-1 event⁸. Three sets of input for the problem are given in Figures A-1 through A-3.

The input of the general control information in Figure A-2 is reflected in the listing of the quantities in Figure 5 from READIT. These parameters are also defined in the Glossary in Appendix B.

Listings of two sets of data for porous materials are shown in Figure 6. The first three lines in each set are written in READIT and reflect the information read by READIT. These lines are common to all types of material models. The next lines are written by SSONED and are specific to porous materials. The first of these contains the control parameters and indicators for SSONED: KFIN, the number of pressure-strain points on the loading and unloading path; NTYPE, the type of unloading requested (see Figure 3); LPRESS and LDENS, labels for the specified values on the paths; and the calibration factors PUNIT and DUNIT. To the right on the line are MAT, THE MATERIAL NUMBER, AND RHOS, the initial density in g/cm^3 . The next two to four lines contain the pressure (or modulus) and strain (or density) values on the loading and unloading path. Finally there are six lines in which the pressures,

ONED: A ONE-DIMENSIONAL WAVE PROPAGATION SOLUTION IN PIECEWISE LINEAR HYSTERETIC MATERIAL

TEST OF SRI ONED TO SIMULATE THE DISKO-1 EVENT, FINAL SET OF DATA
[B6391.ONED]DISKO1E.OND IS DATA, [*.*)RONED.COM, [*.*)MONED.EXE
DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER.

PROBLEM NUMBER 1

COMPUTATION CONTROLS

NOCOMP	=	998	PLOT AT INPUT SCALE FACTORS
LB	=	0	INFINITE BOTTOM BOUNDARY
TEND	=	4.000E-03	END OF PROBLEM TIME (SEC)
NSTOP	=	400	TOTAL NUMBER OF STEPS
GACC	=	9.815E+02	ACC OF GRAVITY FOR GEOSTATIC EFFECT

INPUT CONTROLS

NLAYRS	=	4	NUMBER OF LAYERS
MISP	=	0	SPACING CONST W/IN EACH LAYER
MATTEN	=	0	ALPHA(I) = 1 FOR EVERY NODE
NMTRL	=	4	NUMBER OF MATERIALS
NFORCE	=	0	NUMBER OF PRESSURE-TIME PTS.

OUTPUT CONTROLS

KNTROL1	=	1	STRESS - TIME PLOT
KNTROL2	=	3	PART. VEL. - TIME PLOT
NHIST	=	100	GROUND MOTIONS PRINTED EVERY 'NHIST' TIME INCREMENTS
MVT	=	1	MAX VALUE TABLE PRINTED
NNODE	=	7	NUMBER OF OUTPUT NODES
TNDPLT	=	0.0040	END OF PLOT TIME (SEC)
KVEL	=	1	OUTPUT PARTICLE VELOCITIES IN M /SEC

SCALE FACTORS ARE 1000.0 0.0 60.0 0.0 0.0 0.0 0.0

FIGURE 5 LISTING FROM READIT OF GENERAL CONTROL INFORMATION USED
IN ONED SIMULATION OF DISKO-1 USING AN EXPLICIT TREATMENT
OF THE EXPLOSIVE LAYER

MATERIAL NUMBER 1
 MATERIAL IS SAND DENS= 106.30000 DAMPL,Q= 1.000E-01 4.000E+00 TENSIN= 0.000E+00 NPOR= 1 NPR= 0 NSOL= 0 NVAR= 2
 SHMOD= 1.360E+09 YIELD= 1.000E+04 TANPHI= 2.740E-01
 KFIN= 7 NTYPE= 3 LPRESS= PRES LDENS= DENS PUNIT= 1.00000E+00 DUNIT= 1.00000E+00 MAT= 1 RHOS= 1.70277E+00 FROM -SSONED-
 4.000E+07 7.000E+07 1.700E+08 4.500E+08 1.000E+09 0.000E+00
 1.727E+00 1.746E+00 1.793E+00 1.893E+00 2.065E+00 2.023E+00
 MAT'L = 1 ----- LOADING AND UNLOADING -----
 PT = 1 2 3 4 5 6 7
 STRAIN (%) = 0.0000 1.4114 2.5273 5.2992 11.1720 21.2732 18.8067
 DENS(G/CM3)= 1.702767 1.726800 1.745800 1.793000 1.893000 2.065000 2.023000
 PRESS(PSI) = 0.000E+00 5.801E+02 1.015E+03 2.466E+03 6.527E+03 1.450E+04 0.000E+00
 MODULI (PSI) = 4.110E+04 3.899E+04 5.232E+04 6.915E+04 7.897E+04 5.880E+05 0.000E+00

MATERIAL NUMBER 4
 MATERIAL IS OVERBDN DENS= 95.03500 DAMPL,Q= 1.000E-01 4.000E+00 TENSIN= 0.000E+00 NPOR= 1 NPR= 0 NSOL= 0 NVAR= 2
 SHMOD= 1.360E+09 YIELD= 1.000E+04 TANPHI= 2.740E-01
 KFIN= 10 NTYPE= 1 LPRESS= PRES LDENS= DENS PUNIT= 1.00000E+07 DUNIT= 1.00000E+00 MAT= 2 RHOS= 1.52232E+00 FROM -SSONED-
 8.000E+00 3.900E+01 6.900E+01 1.220E+02 2.300E+02 4.000E+02 1.000E+02 3.000E+01
 0.000E+00
 1.603E+00 1.792E+00 1.904E+00 2.031E+00 2.176E+00 2.343E+00 2.290E+00 2.256E+00
 2.207E+00
 MAT'L = 4 ----- LOADING AND UNLOADING -----
 PT = 1 2 3 4 5 6 7 8 9 10
 STRAIN (%) = 0.0000 5.2999 17.7152 25.0724 33.4150 42.9399 53.9100 50.4285 48.1950 44.9763
 DENS(G/CM3)= 1.522318 1.603000 1.792000 1.904000 2.031000 2.176000 2.343000 2.290000 2.256000 2.207000
 PRESS(PSI) = 0.000E+00 1.160E+03 5.656E+03 1.001E+04 1.769E+04 3.336E+04 5.801E+04 1.450E+04 4.351E+03 0.000E+00
 MODULI (PSI) = 2.189E+04 3.621E+04 5.914E+04 9.214E+04 1.645E+05 2.248E+05 1.250E+06 4.546E+05 1.352E+05 0.000E+00

FIGURE 6 LISTING OF MATERIAL PROPERTY INFORMATION FOR POROUS MATERIALS:
 THREE LINES FROM READIT AND THE REMAINDER FROM SSONED

moduli, strains, and densities are listed in a standard array. This listing is intended to assist users in catching errors in the prescription of the path.

Sample output of the material data for gaseous material is shown in Figure 7. As for porous materials, the first three lines are standard data written by READIT. The next two lines are written by EXPLODE. The first line contains a label "POL" for polytropic and two numerical values: Q and Γ (explosive energy in erg/g and the Grüneisen ratio). The last line, starting with "CONST.VOL...." lists the chemical energy Q , Chapman-Jouguet (C-J) detonation pressure, specific volume at the C-J point, Grüneisen ratio, and the square of the detonation velocity.

The layering information is listed by READIT with the cell layout data, as shown in Figure 8. The input quantities (see Figure 4) are DB, the distance in feet to the bottom of the layer; SPL, the cell size in inches; NSP, the number of cells; and the material number M. The cell listing contains the cell number J, the node position X in cm, the cell size SPL in cm, the density in g/cm^3 , the stress σ_1 (geostatic) in dyn/cm^2 , the cell mass per unit area in g/cm^2 , tensile strength in dyn/cm^2 , material number M, the material name, and the location in the COM array of extra variables. The number of trials used in the iteration to compute the geostatic stress is listed on the right. In this listing, it is apparent that there is a zero-mass cell at the end of each layer. This unused cell permits double nodes to occur at interfaces between materials. The double nodes permit a simple separation process between layers and also allows the user to initialize the layout with a gap between layers.

A second listing of this layout information is provided in the upper part of Figure 9. This listing is like the layout given in the earlier version of ONED. Also in Figure 9 is a short listing of the cells or nodes at which historical information is requested.

During a calculation, two types of snapshot listings are provided to show stresses and other quantities in the cells and nodes at certain times. Both are triggered when NSTEP is a multiple of NHIST. Samples of

MATERIAL NUMBER 2
 MATERIAL IS EXPL1 DENS= 2.74060 DAMPL,Q= 5.000E-02 4.000E+00 TENS= 0.000E+00 NPOR= 0 NPR= 1 NSOL= 0 NVAR= 0
 SHMOD= 0.000E+00 YIELD= 0.000E+00 TANPHI= 0.000E+00
 POL 4.500E+10 2.000E-01
 CONST.VOL.EXPLOSION WITH ENERGY= 4.500E+10 ERG/G, PCJ = 7.902E+08 VCJ = 1.24248E+01 GRUN = 2.00000E-01 SPEED2= 3.960E+10

MATERIAL NUMBER 3
 MATERIAL IS AIR DENS= 0.07295 DAMPL,Q= 5.000E-02 4.000E+00 TENS= 0.000E+00 NPOR= 0 NPR= 1 NSOL= 0 NVAR= 0
 SHMOD= 0.000E+00 YIELD= 0.000E+00 TANPHI= 0.000E+00
 POL 2.139E+09 4.000E-01
 CONST.VOL.EXPLOSION WITH ENERGY= 2.139E+09 ERG/G, PCJ = 2.000E+06 VCJ = 4.99222E+02 GRUN = 4.00000E-01 SPEED2= 4.107E+09

FIGURE 7 LISTING OF MATERIAL PROPERTY INFORMATION FOR EXPLOSIVES AND GASES:
 THREE LINES FROM READIT AND TWO FROM EXPLODE

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LISTING OF THE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-86

J	X	SPL	DENSITY	SIG1	CELMAS	TENS	M	NAME	LVAR	TRIALS
1	0.000E+00	2.540E+00	1.522320	1.898E+03	3.867E+00	0.000E+00	4	OVERBD	1	3
2	2.540E+00	2.540E+00	1.522324	5.693E+03	3.867E+00	0.000E+00	4	OVERBD	3	3
3	5.081E+00	2.540E+00	1.522328	9.489E+03	3.867E+00	0.000E+00	4	OVERBD	5	3
Omitted in the Figure										
31	7.621E+01	2.540E+00	1.522435	1.158E+05	3.867E+00	0.000E+00	4	OVERBD	61	3
32	7.875E+01	2.540E+00	1.522439	1.196E+05	3.867E+00	0.000E+00	4	OVERBD	63	3
33	8.129E+01	2.540E+00	0.000000	0.000E+00	0.000E+00	0.000E+00	0	OVERBD	0	0

LISTING OF THE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-86

J	X	SPL	DENSITY	SIG1	CELMAS	TENS	M	NAME	LVAR	TRIALS
34	8.129E+01	3.810E-01	0.043900	1.215E+05	1.673E-02	0.000E+00	2	EXPL1	0	1
35	8.167E+01	3.810E-01	0.043900	1.215E+05	1.673E-02	0.000E+00	2	EXPL1	0	1
36	8.205E+01	3.810E-01	0.043900	1.215E+05	1.673E-02	0.000E+00	2	EXPL1	0	1
Omitted in the Figure										
43	8.472E+01	3.810E-01	0.043900	1.216E+05	1.673E-02	0.000E+00	2	EXPL1	0	1
44	8.510E+01	3.810E-01	0.000000	0.000E+00	0.000E+00	0.000E+00	0	EXPL1	0	0

LISTING OF THE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-86

J	X	SPL	DENSITY	SIG1	CELMAS	TENS	M	NAME	LVAR	TRIALS
45	8.510E+01	7.620E+00	0.001168	1.216E+05	8.904E-03	0.000E+00	3	AIR	0	1
46	9.272E+01	7.620E+00	0.000000	0.000E+00	0.000E+00	0.000E+00	0	AIR	0	0

LISTING OF THE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-86

J	X	SPL	DENSITY	SIG1	CELMAS	TENS	M	NAME	LVAR	TRIALS
47	9.272E+01	1.016E+00	1.702840	1.225E+05	1.730E+00	0.000E+00	1	SAND	65	3
48	9.374E+01	1.016E+00	1.702841	1.242E+05	1.730E+00	0.000E+00	1	SAND	67	3
49	9.475E+01	1.016E+00	1.702842	1.259E+05	1.730E+00	0.000E+00	1	SAND	69	3
Omitted in the Figure										
296	3.457E+02	1.016E+00	1.703094	5.453E+05	1.730E+00	0.000E+00	1	SAND	563	3
297	3.467E+02	1.016E+00	0.000000	0.000E+00	0.000E+00	0.000E+00	0	SAND	0	0

FIGURE 8 PARTIAL LISTING OF THE LAYER, NODE, AND CELL INFORMATION FROM READIT FOR THE ONED SIMULATION OF DISKO-1 WITH AN EXPLICIT TREATMENT OF THE EXPLOSIVE LAYER

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-87
LAYER PROPERTIES

LAYER NUMBER	NODES	BOTTOM (FT)	GAMMA (PCF)	PCRIT (%)	SPACING (IN)	TENSION (PSI)
1	1 - 33	2.6670	95.04	0.10	1.0001	0.000E+00
2	34 - 44	2.7920	2.74	0.05	0.1500	0.000E+00
3	45 - 46	3.0420	0.07	0.05	3.0000	0.000E+00
4	47 - 298	11.3750	106.30	0.10	0.4000	0.000E+00

DEPTH (ft)	OUTPUT DESIRED AT NODE
3.1420	50
3.5420	62
4.5420	92
5.5420	122
6.5420	152
7.5420	182
8.5420	212

FIGURE 9 LISTING OF THE LAYOUT AND DEPTHS AT WHICH HISTORIES
ARE REQUESTED FOR THE ONED SIMULATION OF DISKO-1
WITH AN EXPLICIT TREATMENT OF THE EXPLOSIVE LAYER

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both these snapshot listings are shown in Figures 10 and 11. The listed quantities in Figure 10 include position X in cm, velocity VEL in cm/s, the mechanical stress RMECH in the direction of propagation in dyn/cm², thermodynamic stress SIG1 in dyn/cm², the artificial viscous stress DSTRS in dyn/cm², the pressure in dyn/cm², the density DENS in g/cm³, the sound speed SSPEED in cm/s, the deviator stress SIGDEV in the direction of propagation in dyn/cm², and the yield strength Y in dyn/cm².

The second snapshot listing, shown in Figure 11, is similar to that obtained in the original version of ONED. This listing provides most of the same quantities, but only for the requested nodes and cells. The applied pressure at the surface, the mechanical stress RMECH, thermodynamic stress SIG1 and the damping stress are in psi. The compressive strain is in percent, acceleration in g's, particle velocity in m/s or in/s (depending on the setting of the indicator KVEL), and the node location in cm or inches (depending on KVEL).

At the end of the calculation, a summary is given of the maximum values that have occurred at all depths. Such a summary is shown in Figure 12. The listed quantities include the depth to the node, the mechanical stress RMECH, the velocity, coordinate position, strain in the cell, and acceleration. The depth, velocity, position, and acceleration pertain to the node J, and the other quantities refer to the following cell. In each case the time at which the maximum quantity occurred is also shown.

The final listing shows a history of some quantities at the requested nodes. A portion of one of these listings from SCRIBE (part of HISTORY) is given in Figure 13. The quantities are RMECH in psi, the acceleration in g's, the velocity in m/s, position in in., strain in percent, impulse in dyn-s/cm², and the stress in the first and second principal directions in MPa. The impulse is obtained as the integral of stress RMECH over time at the cell:

$$I = \int_0^{t_n} R \, dt \quad (51)$$

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 23-JAN-86
 SNAPSHOT AT NSTEP = 400, TIME = 1.03428E-03

J	MAT	L	X CM	VEL CM/SEC	RMECH DYN/CM2	SIG1 DYN/CM2	DSTRS DYN/CM2	PRESS DYN/CM2	DENS G/CM3	SSPEED CM/SEC	SIGDEV DYN/CM2	Y DYN/CM2
1	4	1	8.44917E-07	3.96159E-04	6.043E+02	6.043E+02	5.664E-05	1.298E+00	1.522321	8.552E+04	6.030E+02	1.000E+04
J = 2 to 31 were omitted from the Figure. MAT 4 is the Overburden.												
32	4	63	7.67864E+01	-1.68039E+03	5.639E+07	5.639E+07	0.000E+00	4.755E+07	1.647572	8.220E+04	8.839E+06	1.000E+04
33	0	0	7.91338E+01	-1.48328E+03	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000000	0.000E+00	0.000E+00	0.000E+00
34	2	0	7.91338E+01	-1.48328E+03	6.816E+07	6.816E+07	0.000E+00	6.816E+07	0.010444	8.850E+04	0.000E+00	0.000E+00
J = 35 to 43 were omitted from the Figure. MAT 2 is Primacord Explosive.												
42	2	0	9.12713E+01	2.20655E+03	8.352E+07	8.340E+07	1.207E+05	8.340E+07	0.012357	9.000E+04	0.000E+00	0.000E+00
43	2	0	9.26249E+01	1.74489E+03	8.489E+07	8.474E+07	1.450E+05	8.474E+07	0.012522	9.012E+04	0.000E+00	0.000E+00
44	0	0	9.39606E+01	1.21503E+03	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000000	0.000E+00	0.000E+00	0.000E+00
45	3	0	9.39606E+01	1.21503E+03	8.488E+07	8.488E+07	6.415E+01	8.488E+07	0.029140	6.386E+04	0.000E+00	0.000E+00
46	0	0	9.42661E+01	1.19526E+03	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000000	0.000E+00	0.000E+00	0.000E+00
47	1	65	9.42661E+01	1.19526E+03	8.963E+07	8.941E+07	2.253E+05	7.764E+07	1.802557	1.533E+05	1.177E+07	1.000E+04
J = 48 to 60 were omitted from the Figure. MAT 1 is Flume Sand. J = 62 is at the 6" gage depth.												
61	1	93	1.07948E+02	1.04290E+03	8.239E+07	8.239E+07	0.000E+00	6.032E+07	1.757782	1.552E+05	2.207E+07	1.000E+04
62	1	95	1.08932E+02	1.05206E+03	7.892E+07	7.892E+07	0.000E+00	5.758E+07	1.755109	1.553E+05	2.134E+07	1.000E+04
63	1	97	1.09918E+02	1.12735E+03	8.410E+07	8.403E+07	6.937E+04	6.136E+07	1.754481	1.554E+05	2.267E+07	1.000E+04
64	1	99	1.10904E+02	1.07618E+03	8.910E+07	8.910E+07	0.000E+00	6.502E+07	1.754007	1.554E+05	2.408E+07	1.000E+04
J = 65 to 87 were omitted from the Figure.												
88	1	147	1.34627E+02	1.55218E+03	1.395E+08	1.395E+08	0.000E+00	1.006E+08	1.760313	1.551E+05	3.886E+07	1.000E+04
89	1	149	1.35610E+02	1.60514E+03	1.267E+08	1.259E+08	8.149E+05	9.082E+07	1.755626	1.553E+05	3.510E+07	1.000E+04
90	1	151	1.36595E+02	1.33766E+03	1.115E+08	1.115E+08	0.000E+00	8.105E+07	1.751037	1.555E+05	3.049E+07	1.000E+04
91	1	153	1.37583E+02	1.44746E+03	1.254E+08	1.252E+08	2.069E+05	9.030E+07	1.755382	1.553E+05	3.490E+07	1.000E+04
J = 92 is at the 18" gage depth.												
92	1	155	1.38569E+02	1.34110E+03	9.586E+07	9.476E+07	1.106E+06	6.953E+07	1.745022	1.558E+05	2.523E+07	1.000E+04
93	1	157	1.39560E+02	1.00945E+03	8.500E+07	8.500E+07	0.000E+00	6.320E+07	1.741659	1.559E+05	2.180E+07	1.000E+04
94	1	159	1.40554E+02	1.23000E+03	1.088E+08	1.088E+08	0.000E+00	7.853E+07	1.749813	1.556E+05	3.031E+07	1.000E+04
95	1	161	1.41543E+02	1.24849E+03	9.271E+07	9.162E+07	1.088E+06	6.608E+07	1.743317	1.559E+05	2.554E+07	1.000E+04
96	1	163	1.42535E+02	9.18443E+02	6.433E+07	6.322E+07	1.116E+06	4.559E+07	1.730343	1.565E+05	1.762E+07	1.000E+04
97	1	165	1.43535E+02	5.72848E+02	3.717E+07	3.645E+07	7.259E+05	2.628E+07	1.718559	1.570E+05	1.016E+07	1.000E+04
98	1	167	1.44541E+02	2.91633E+02	1.758E+07	1.731E+07	2.667E+05	1.248E+07	1.710266	1.574E+05	4.830E+06	1.000E+04
99	1	169	1.45553E+02	1.24673E+02	7.249E+06	7.186E+06	6.293E+04	5.178E+06	1.705878	1.576E+05	2.008E+06	1.000E+04
100	1	171	1.46567E+02	4.78922E+01	2.904E+06	2.892E+06	1.132E+04	2.081E+06	1.704017	1.577E+05	8.112E+05	1.000E+04
101	1	173	1.47582E+02	1.86556E+01	1.354E+06	1.352E+06	1.913E+03	9.701E+05	1.703349	1.577E+05	3.819E+05	1.000E+04
102	1	175	1.48598E+02	8.85777E+00	8.363E+05	8.358E+05	5.134E+02	5.978E+05	1.703126	1.577E+05	2.380E+05	1.000E+04
103	1	177	1.49614E+02	4.85159E+00	5.449E+05	5.447E+05	2.562E+02	4.171E+05	1.703017	1.577E+05	1.276E+05	1.000E+04
104	1	179	1.50630E+02	2.32672E+00	3.676E+05	3.675E+05	1.123E+02	3.097E+05	1.702953	1.577E+05	5.782E+04	1.000E+04
105	1	181	1.51646E+02	9.56769E-01	2.781E+05	2.781E+05	4.507E+01	2.558E+05	1.702920	1.577E+05	2.228E+04	1.000E+04
106	1	183	1.52662E+02	3.16232E-01	2.392E+05	2.391E+05	1.698E+01	2.327E+05	1.702906	1.577E+05	6.427E+03	1.000E+04
107	1	185	1.53678E+02	5.38749E-02	2.260E+05	2.260E+05	3.357E+00	2.254E+05	1.702902	1.577E+05	6.372E+02	1.000E+04
108	1	187	1.54694E+02	0.00000E+00	2.261E+05	2.261E+05	0.000E+00	2.261E+05	1.702902	1.577E+05	-2.068E-07	1.000E+04

Note: Remainder of Listing is beyond the wave front.

FIGURE 10 LISTING OF STRESSES, VELOCITIES, AND OTHER PROPERTIES AT CYCLE 400,
 1.03 ms FROM DISKO-1 SIMULATION WITH ONED, USING AN EXPLICIT
 TREATMENT OF THE EXPLOSIVE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-87									
TIME	SURFACE STRESS PSI	NODE NO.	APPLIED STRESS PSI	REACTION STRESS PSI	DAMPING STRESS PSI	STRAIN PERCENT	ACCEL G'S	PARTICLE VELOCITY M/S	POSITION IN
6.474E-04	0.00	50	972.83	971.68	1.15	3.93064	8468.844	11.784	38.080
		62	1908.41	1908.41	0.00	3.77272	-9397.986	14.595	42.703
		92	2.88	2.88	0.00	0.00702	0.000	0.000	54.503
		122	3.62	3.62	0.00	0.00882	0.000	0.000	66.503
		152	4.36	4.36	0.00	0.01061	0.000	0.000	78.502
		182	5.10	5.10	0.00	0.01241	0.000	0.000	90.502
		212	5.84	5.84	0.00	0.01421	0.000	0.000	102.501

FIGURE 11 SNAPSHOT LISTING OF STRESSES, VELOCITIES, AND OTHER PROPERTIES
AT THE REQUESTED NODES AT CYCLE 300 FROM THE DISKO-1 SIMULATION
WITH ONED, USING AN EXPLICIT TREATMENT OF THE EXPLOSIVE LAYOUT

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-87
SUMMARY OF MAXIMUM VALUES

NODE NO.	DEPTH FT.	REACTION		STRESS		VELOCITY		DISPLACEMENT		STRAIN		ACCELERATION	
		VALUE (MPA)	TIME (SEC)	VALUE (MPA)	TIME (SEC)	VALUE (M/S)	TIME (SEC)	VALUE (IN)	TIME (SEC)	VALUE (%)	TIME (SEC)	VALUE (G'S)	TIME (SEC)
Nodes 1 - 16 were omitted from the Figure. Overburden is contained in cells 1 to 32.													
17	1.3335	6.94	1.035E-03	-9.043	1.035E-03	-0.027	1.035E-03	3.216E+00	1.035E-03	-8.169E+03	1.035E-03		
18	1.4168	9.38	1.035E-03	-14.234	1.035E-03	-0.058	1.035E-03	4.445E+00	1.035E-03	-8.371E+03	9.738E-04		
19	1.5002	10.03	1.016E-03	-17.124	1.035E-03	-0.100	1.035E-03	4.793E+00	1.020E-03	-8.584E+03	9.047E-04		
20	1.5835	10.06	9.657E-04	-17.409	9.895E-04	-0.146	1.035E-03	4.808E+00	9.698E-04	-8.839E+03	8.366E-04		
21	1.6669	9.99	9.534E-04	-17.300	9.364E-04	-0.190	1.035E-03	4.772E+00	9.575E-04	-9.173E+03	7.648E-04		
22	1.7502	10.43	9.272E-04	-17.548	9.408E-04	-0.234	1.035E-03	4.982E+00	9.272E-04	-9.641E+03	6.958E-04		
23	1.8336	11.00	8.810E-04	-18.516	9.000E-04	-0.280	1.035E-03	5.254E+00	8.810E-04	-1.035E+04	6.324E-04		
24	1.9169	11.11	8.366E-04	-19.156	8.542E-04	-0.326	1.035E-03	5.305E+00	8.409E-04	-1.123E+04	5.712E-04		
25	2.0002	11.60	7.994E-04	-19.281	8.242E-04	-0.376	1.035E-03	5.512E+00	7.994E-04	-1.201E+04	5.140E-04		
26	2.0836	12.74	6.991E-04	-19.764	5.281E-04	-0.426	1.035E-03	5.977E+00	7.024E-04	1.267E+04	5.618E-04		
27	2.1669	13.73	7.431E-04	-21.639	4.782E-04	-0.480	1.035E-03	6.337E+00	7.431E-04	1.655E+04	5.106E-04		
28	2.2503	13.97	4.154E-04	-22.618	4.327E-04	-0.536	1.035E-03	6.361E+00	6.991E-04	-1.764E+04	6.031E-04		
29	2.3336	14.71	3.733E-04	-23.248	3.931E-04	-0.591	1.035E-03	6.358E+00	6.618E-04	-1.666E+04	2.128E-04		
30	2.4170	16.92	2.982E-04	-24.727	3.352E-04	-0.647	1.035E-03	7.130E+00	2.982E-04	-2.335E+04	1.567E-04		
31	2.5003	18.96	1.299E-04	-26.314	1.545E-04	-0.706	1.035E-03	8.166E+00	2.406E-04	-3.865E+04	2.323E-04		
32	2.5837	26.87	6.911E-05	-35.864	1.006E-04	-0.773	1.035E-03	9.775E+00	8.227E-05	-5.971E+04	5.188E-05		
33	2.6670	0.00	1.035E-03	-54.919	4.587E-05	-0.849	1.035E-03	0.000E+00	1.035E-03	-2.002E+05	1.044E-07		
The preceding cell is unused: 33 represents a node only. The next cells contain explosive.													
34	2.6670	38.18	1.000E-12	-54.919	4.587E-05	-0.849	1.035E-03	-8.542E+01	1.504E-04	-2.002E+05	1.044E-07		
35	2.6795	38.18	2.297E-07	-220.015	1.879E-04	0.694	1.484E-04	-8.264E+01	1.316E-04	-1.701E+06	1.524E-04		
36	2.6920	38.18	1.037E-06	414.680	1.053E-04	1.368	1.445E-04	-7.956E+01	1.133E-04	1.894E+06	3.866E-05		
37	2.7045	38.18	2.711E-06	621.674	9.093E-05	1.871	1.370E-04	-7.563E+01	9.767E-05	2.169E+06	3.610E-05		
38	2.7170	38.18	5.064E-06	826.401	7.994E-05	2.221	1.316E-04	-7.489E+01	1.027E-03	2.494E+06	3.086E-05		
39	2.7295	38.18	2.711E-06	998.172	7.159E-05	2.457	1.265E-04	-7.416E+01	1.009E-03	2.874E+06	2.505E-05		
40	2.7420	38.18	1.349E-06	1123.381	6.647E-05	2.626	1.248E-04	-7.477E+01	8.721E-04	-4.588E+06	6.989E-05		
41	2.7545	38.18	3.801E-07	1249.056	6.375E-05	2.754	1.214E-04	-7.550E+01	8.857E-04	-1.438E+07	6.594E-05		
42	2.7670	74.54	6.401E-05	1311.246	5.511E-05	2.857	1.182E-04	-7.635E+01	2.777E-04	-3.804E+07	6.430E-05		
43	2.7795	172.82	6.353E-05	1323.457	3.610E-05	3.113	1.035E-03	-7.850E+01	2.733E-04	-8.240E+07	6.353E-05		
44	2.7920	0.00	1.035E-03	1396.479	1.338E-05	3.489	1.035E-03	0.000E+00	1.035E-03	-1.480E+08	6.320E-05		
The preceding cell is unused: 44 represents a node only. The next cell contains air.													
45	2.7920	311.25	6.320E-05	1396.479	1.338E-05	3.489	1.035E-03	3.257E+04	6.320E-05	-1.480E+08	6.320E-05		
46	3.0420	0.00	1.035E-03	59.499	6.705E-05	0.609	1.035E-03	0.000E+00	1.035E-03	3.521E+06	6.335E-05		
The preceding cell is unused. The next cells contain soil - the test bed.													
47	3.0420	38.99	7.994E-05	59.499	6.705E-05	0.609	1.035E-03	7.648E+00	8.110E-05	3.521E+06	6.335E-05		
48	3.0753	32.36	9.767E-05	37.337	8.965E-05	0.587	1.035E-03	6.678E+00	1.334E-04	1.372E+05	7.346E-05		
49	3.1087	28.60	1.149E-04	32.407	1.069E-04	0.568	1.035E-03	6.092E+00	1.524E-04	9.061E+04	9.223E-05		
50	3.1420	26.65	1.334E-04	29.874	1.248E-04	0.551	1.035E-03	5.769E+00	1.334E-04	7.567E+04	1.069E-04		
A new title occurs at this line.													
51	3.1753	25.66	1.524E-04	28.598	1.426E-04	0.535	1.035E-03	5.617E+00	1.524E-04	7.278E+04	1.231E-04		
Nodes 52 to 60 are omitted for the Figure. Node 62 is at the 6" gage depth in the Flume sand.													
61	3.5086	20.55	3.435E-04	23.603	3.352E-04	0.396	1.035E-03	4.692E+00	3.435E-04	6.230E+04	3.142E-04		
62	3.5420	19.84	3.619E-04	22.860	3.539E-04	0.383	1.035E-03	4.547E+00	3.619E-04	5.993E+04	3.352E-04		
63	3.5753	19.11	3.818E-04	22.109	3.733E-04	0.371	1.035E-03	4.404E+00	3.818E-04	5.775E+04	3.539E-04		
64	3.6086	18.42	3.988E-04	21.362	3.903E-04	0.359	1.035E-03	4.267E+00	4.016E-04	5.560E+04	3.733E-04		
65	3.6420	17.79	4.211E-04	20.679	4.098E-04	0.348	1.035E-03	4.142E+00	4.211E-04	5.320E+04	3.931E-04		
66	3.6753	17.23	4.386E-04	20.054	4.298E-04	0.337	1.035E-03	4.022E+00	4.417E-04	5.085E+04	4.126E-04		
Nodes 67 to 77 are omitted here.													
78	4.0753	14.46	8.632E-04	16.788	8.542E-04	0.210	1.035E-03	3.477E+00	8.632E-04	3.166E+04	6.583E-04		
79	4.1086	14.40	8.857E-04	16.744	8.765E-04	0.200	1.035E-03	3.464E+00	8.857E-04	3.046E+04	6.790E-04		
Nodes 80 to 90 are omitted here. Node 92 is at the 18" gage depth.													
91	4.5086	12.56	1.035E-03	14.757	1.027E-03	0.064	1.035E-03	3.094E+00	1.035E-03	2.169E+04	1.009E-03		
92	4.5419	11.07	9.856E-04	13.521	1.035E-03	0.052	1.035E-03	2.799E+00	9.895E-04	2.141E+04	1.031E-03		
93	4.5753	11.00	1.009E-03	13.247	1.001E-03	0.042	1.035E-03	2.785E+00	1.009E-03	1.830E+04	9.698E-04		
94	4.6086	10.94	1.031E-03	13.156	1.024E-03	0.033	1.035E-03	2.772E+00	1.031E-03	1.778E+04	9.895E-04		
95	4.6419	9.34	1.035E-03	12.550	1.035E-03	0.022	1.035E-03	2.401E+00	1.035E-03	1.733E+04	1.009E-03		
96	4.6753	6.51	1.035E-03	9.283	1.035E-03	0.013	1.035E-03	1.640E+00	1.035E-03	1.689E+04	1.031E-03		
97	4.7086	3.79	1.035E-03	5.821	1.035E-03	0.007	1.035E-03	9.448E-01	1.035E-03	1.566E+04	1.035E-03		
98	4.7419	1.80	1.035E-03	2.981	1.035E-03	0.003	1.035E-03	4.509E-01	1.035E-03	1.068E+04	1.035E-03		
99	4.7753	0.74	1.035E-03	1.280	1.035E-03	0.001	1.035E-03	1.876E-01	1.035E-03	5.402E+03	1.035E-03		
100	4.8086	0.30	1.035E-03	0.493	1.035E-03	0.000	1.035E-03	7.534E-02	1.035E-03	2.206E+03	1.035E-03		
The remainder of the listing is omitted for the Figure.													

FIGURE 12 PARTIAL LISTING OF THE MAXIMUM VALUE TABLE FOR THE ONED SIMULATION OF DISKO-1 WITH EXPLICIT TREATMENT OF THE EXPLOSIVE

JA-6391-101

DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER. 20-JAN-87

HISTORIES AT NODE NO. 62

NS	TIME SEC	RMECH MPA	ACC G'S	VELOCITY M/SEC	POSITION IN	STRAIN (%)	IMPULSE DYN-S/CM^2	SIGMA-1 MPA	SIGMA-2 MPA
166	2.201E-04	1.633E-02	5.159E+01	4.678E-03	4.250E+01	5.551E-03	3.285E+01	1.633E-02	1.543E-02
167	2.216E-04	1.669E-02	5.844E+01	5.522E-03	4.250E+01	5.628E-03	3.310E+01	1.669E-02	1.558E-02
168	2.231E-04	1.711E-02	6.608E+01	6.484E-03	4.250E+01	5.717E-03	3.335E+01	1.710E-02	1.575E-02
Time steps 169 to 200 are omitted for the Figure.									
201	3.105E-04	2.423E+00	2.636E+04	4.723E+00	4.251E+01	5.914E-01	3.732E+02	2.324E+00	1.352E+00
202	3.142E-04	3.077E+00	3.158E+04	5.892E+00	4.251E+01	7.458E-01	4.667E+02	2.931E+00	1.705E+00
203	3.176E-04	3.807E+00	3.686E+04	7.172E+00	4.251E+01	9.166E-01	5.756E+02	3.602E+00	2.096E+00
204	3.208E-04	4.610E+00	4.219E+04	8.531E+00	4.251E+01	1.103E+00	7.025E+02	4.333E+00	2.521E+00
205	3.238E-04	5.502E+00	4.887E+04	1.001E+01	4.251E+01	1.308E+00	8.502E+02	5.139E+00	2.990E+00
206	3.267E-04	6.466E+00	5.410E+04	1.159E+01	4.251E+01	1.534E+00	1.022E+03	6.003E+00	3.493E+00
207	3.296E-04	7.505E+00	5.781E+04	1.323E+01	4.251E+01	1.783E+00	1.220E+03	6.931E+00	4.033E+00
208	3.324E-04	8.632E+00	5.986E+04	1.491E+01	4.252E+01	2.055E+00	1.447E+03	7.944E+00	4.622E+00
209	3.352E-04	9.825E+00	5.993E+04	1.657E+01	4.252E+01	2.346E+00	1.704E+03	9.029E+00	5.254E+00
210	3.380E-04	1.121E+01	5.799E+04	1.816E+01	4.252E+01	2.652E+00	1.993E+03	1.033E+01	6.009E+00
211	3.408E-04	1.283E+01	5.318E+04	1.961E+01	4.252E+01	2.963E+00	2.318E+03	1.189E+01	6.917E+00
212	3.435E-04	1.436E+01	4.535E+04	2.082E+01	4.252E+01	3.270E+00	2.680E+03	1.342E+01	7.810E+00
213	3.461E-04	1.576E+01	3.644E+04	2.178E+01	4.253E+01	3.562E+00	3.075E+03	1.488E+01	8.660E+00
214	3.487E-04	1.699E+01	2.613E+04	2.246E+01	4.253E+01	3.831E+00	3.501E+03	1.622E+01	9.441E+00
215	3.513E-04	1.803E+01	1.425E+04	2.282E+01	4.253E+01	4.067E+00	3.956E+03	1.741E+01	1.013E+01
216	3.539E-04	1.884E+01	1.429E+03	2.286E+01	4.253E+01	4.265E+00	4.438E+03	1.839E+01	1.070E+01
217	3.566E-04	1.942E+01	-1.156E+04	2.256E+01	4.254E+01	4.415E+00	4.945E+03	1.915E+01	1.114E+01
218	3.592E-04	1.976E+01	-2.392E+04	2.195E+01	4.254E+01	4.511E+00	5.475E+03	1.963E+01	1.142E+01
219	3.619E-04	1.984E+01	-3.479E+04	2.103E+01	4.254E+01	4.547E+00	6.023E+03	1.981E+01	1.153E+01
220	3.647E-04	1.958E+01	-4.334E+04	1.987E+01	4.254E+01	4.519E+00	6.582E+03	1.958E+01	1.139E+01
The new heading was omitted here.									
221	3.675E-04	1.883E+01	-4.824E+04	1.854E+01	4.254E+01	4.425E+00	7.138E+03	1.883E+01	1.096E+01
222	3.704E-04	1.762E+01	-4.789E+04	1.719E+01	4.255E+01	4.274E+00	7.664E+03	1.762E+01	1.025E+01
223	3.733E-04	1.611E+01	-4.202E+04	1.600E+01	4.255E+01	4.086E+00	8.143E+03	1.611E+01	9.372E+00
224	3.761E-04	1.449E+01	-3.168E+04	1.511E+01	4.255E+01	3.885E+00	8.573E+03	1.449E+01	8.433E+00
225	3.789E-04	1.294E+01	-1.863E+04	1.459E+01	4.255E+01	3.692E+00	8.957E+03	1.294E+01	7.532E+00
226	3.818E-04	1.161E+01	-4.871E+03	1.446E+01	4.255E+01	3.526E+00	9.303E+03	1.161E+01	6.755E+00
227	3.846E-04	1.061E+01	7.725E+03	1.467E+01	4.255E+01	3.402E+00	9.618E+03	1.061E+01	6.177E+00
228	3.874E-04	1.005E+01	1.755E+04	1.516E+01	4.256E+01	3.332E+00	9.913E+03	1.005E+01	5.847E+00
229	3.903E-04	9.950E+00	2.342E+04	1.581E+01	4.256E+01	3.320E+00	1.020E+04	9.950E+00	5.790E+00
230	3.931E-04	1.030E+01	2.465E+04	1.651E+01	4.256E+01	3.362E+00	1.049E+04	1.027E+01	5.997E+00

FIGURE 13 PARTIAL LISTING OF THE HISTORIES FOR NODE 62 (6-in.-GAGE DEPTH) FROM ONED SIMULATION OF DISKO-1 WITH AN EXPLICIT TREATMENT OF THE EXPLOSIVE LAYER

where t_n is the current time. The quantities SIGMA-1 AND SIGMA-2 are the stresses in the direction of propagation and orthogonal to that direction, respectively.

In cases where the wave propagation is caused by a pressure history instead of a detonation, the history is read and printed by READIT during initialization. A sample of the pressure history listing (THIST, PHIST) is shown in Figure 14. This pressure history was obtained from a GUINSY calculation⁹ of the DISKO-1 stress gage records. The input for this case is given in Figure A-3.

Some of the data printed in the preceding figures are also available for plotting. The SCRIBE listings of this histories have been stored on file 7 and can be plotted. Also the maximum value table in Figure 12 is available on file 7, following the SCRIBE data.

DISKO 1 FOR WES, ONED SIMULATION WITH PRESSURES FROM GUINSY20-JAN-86
FORCE HISTORY ON FIRST NODE

TIME	PRESS	TIME	PRESS	TIME	PRESS	TIME	PRESS	TIME	PRESS
0.000E+00	0.000E+00	2.000E-06	2.326E+02	4.000E-06	4.651E+02	6.000E-06	6.977E+02	1.000E-05	8.335E+02
1.400E-05	9.678E+02	1.700E-05	1.100E+03	2.100E-05	1.229E+03	2.500E-05	1.354E+03	2.900E-05	1.477E+03
3.300E-05	1.595E+03	3.700E-05	1.711E+03	4.000E-05	1.825E+03	4.400E-05	1.938E+03	4.800E-05	2.051E+03
5.200E-05	2.164E+03	5.600E-05	2.278E+03	5.900E-05	2.393E+03	6.300E-05	2.514E+03	6.700E-05	2.636E+03
7.100E-05	2.762E+03	7.500E-05	2.894E+03	7.900E-05	3.028E+03	8.200E-05	3.167E+03	9.400E-05	2.914E+03
1.040E-04	2.697E+03	1.130E-04	2.512E+03	1.230E-04	2.354E+03	1.330E-04	2.217E+03	1.420E-04	2.097E+03
1.520E-04	1.991E+03	1.640E-04	1.896E+03	1.740E-04	1.811E+03	1.840E-04	1.733E+03	1.930E-04	1.660E+03
2.030E-04	1.592E+03	2.130E-04	1.530E+03	2.220E-04	1.471E+03	2.340E-04	1.418E+03	2.440E-04	1.370E+03
2.540E-04	1.329E+03	2.630E-04	1.295E+03	2.730E-04	1.271E+03	2.830E-04	1.257E+03	2.900E-04	1.297E+03
2.970E-04	1.355E+03	3.020E-04	1.428E+03	3.090E-04	1.510E+03	3.170E-04	1.600E+03	3.240E-04	1.691E+03
3.290E-04	1.782E+03	3.360E-04	1.867E+03	3.430E-04	1.945E+03	3.510E-04	2.015E+03	3.550E-04	2.075E+03
3.630E-04	2.126E+03	3.700E-04	2.169E+03	3.750E-04	2.211E+03	3.820E-04	2.255E+03	4.030E-04	2.130E+03
4.250E-04	2.003E+03	4.470E-04	1.877E+03	4.680E-04	1.756E+03	4.890E-04	1.641E+03	5.110E-04	1.533E+03
5.320E-04	1.434E+03	5.540E-04	1.343E+03	5.750E-04	1.260E+03	5.970E-04	1.185E+03	6.180E-04	1.117E+03
6.390E-04	1.054E+03	6.610E-04	9.964E+02	6.830E-04	9.424E+02	7.040E-04	8.910E+02	7.250E-04	8.420E+02
7.460E-04	7.950E+02	7.680E-04	7.499E+02	7.900E-04	7.078E+02	8.110E-04	6.696E+02		

DEPTH(ft)	OUTPUT DESIRED AT NODE
0.1000	4
0.2500	8
0.5000	16
1.5000	46
1.6700	51
1.8300	55
2.5000	76
3.5000	106
4.5000	136
5.5000	166

FIGURE 14 LISTING FROM READIT OF THE FORCE HISTORY AND OF THE LOCATIONS WHERE PRINTED AND PLOTTED OUTPUT IS DESIRED: ONED SIMULATION OF DISKO-1 USING THE PRESSURE HISTORY FROM GUINSY

REFERENCES

1. Radhakrishnan, N., and Rohani, B. November 1971. "A One-Dimensional Plane Wave Propagation Code for Layered Nonlinear Hysteretic Media," Technical Report S-71-12, US Army Engineer Waterways Experiment Station, Vicksburg, MS.
2. Von Neumann, J., and Richtmyer, R. D. 1950. "A Method for the Numerical Calculation of Hydrodynamic Shocks," J. Appl. Phys., Vol 21, pg 232.
3. Richtmyer, R. D., and Morton, K. W. 1967. Difference Methods for Initial-Value Problems, 2d ed., Interscience Publishers, New York, NY.
4. Herrmann, W., Holzhauser, P., and Thompson, R. J. February 1967. "WONDY, a Computer Program for Calculating Problems of Motion in One Dimension," SC-RR-66-601, Sandia Corporation, Albuquerque, NM.
5. Seaman, L., Barbee, T. W. Jr., and Curran, D. R. December 1971. "Dynamic Fracture Criteria of Homogeneous Materials," Report No. AFWL-TR-71-156, Air Force Weapons Laboratory, Kirtland Air Force Base, NM.
6. Hill, R. 1950. The Mathematical Theory of Plasticity. Clarendon Press, Oxford.
7. Terzaghi, K. 1943. Theoretical Soil Mechanics. John Wiley and Sons, Inc., New York, NY.
8. Seaman, L. July 1987. "Analysis of Dynamic In Situ Backfill Property Tests: Report 4, Computation of K_0 Stress-Strain and Stress-Path Relations for Dry Flume Sand in the DISKO-1 Experiment," Technical Report SL-87-11, US Army Engineer Waterways Experiment Station, Vicksburg, MS.
9. Seaman, L. July 1987. "Analysis of Dynamic In Situ Backfill Property Tests: Report 2, An Improved Lagrangian Analysis for Stress and Particle Velocity Gage Arrays," Technical Report SL-87-11, US Army Engineer Waterways Experiment Station, Vicksburg, MS.

Appendix A

SAMPLE INPUT FOR ONED

This appendix provides sample input files and supplements the input description in Section IV. The input files show sample problems in which either a surface pressure history or an explosive detonation drives the waves. Material properties characteristic of solids, porous materials, explosives, and air are provided.

Figures A-1 through A-3 show three treatments of the DISKO-1 event⁸. In Figure A-1, the loading is applied by an average pressure history from the blast pressure gages, and in Figure A-2 by detonation of the explosive layer. The explosive is treated as a polytropic gas. Figure A-3 contains another input for DISKO-1 in which the loading is a pressure history derived from a Lagrangian analysis performed using the GUINSY code⁹. In Figure A-4 is an input file for a synthetic problem containing several layers of porous and solid materials, an explosive, and a gap. The gap is simply an empty layer; it closes during the calculation. The isentrope (pressure-volume curve) for the explosive is provided by a table of volume and pressure values.

TEST OF SRI ONED TO SIMULATE THE DISKO-1 EVENT, FINAL SET OF DATA
 [B6391.ONED]DISKO1BL.OND IS DATA, [*.]*RONED.COM, [*.]*MONED.EXE
 1 DISKO 1 FOR WES, ONED SIMULATION WITH PRESSURES FROM BLAST GAGES.
 998, 0, 0.004, 400, 1. NOCOMP, BOTTOM BOUNDARY INDICATOR, TEND, NSTOP, GACC
 1, 0, 0, 0, 1, 139, NLAYRS, MISP, NSP, MATTEN, NMIRLS, NFORCE
 10, 1, 3, 0, 0, 0, 0, 0, 100, 1, 1, 1, 0.004 NNODE, KNTROL (7), NHIST, MVT, NSV, KVEL, TND
 0.1, 0.25, 0.5, 1.5, 1.67, 1.83, 2.5, 3.5, 4.5, 5.5 DEPTHS FOR HIST. (FT)
 1000., 0.0, 60., 0., 0., 0., 0. SK (7)
 'SAND', 106.3, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENSIN, POR, PR, SOL, VAR FOR SAND
 1.36E+9, 1.0E+4, 0.274 SHMOD, YIELD, TANPHI
 7, 3, 'P', 'D', 1., 1.0 SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
 4.E7, 7.E7, 1.7E8, 4.5E8, 1.E9, 0., FLUME SAND, P, DYN/CM² (K=2.92E9)
 1.7268, 1.7458, 1.793, 1.893, 2.065, 2.023 DENSITIES, G/CM³
 8.333 0.4, 0, 1 DB (DIST TO BOT, FT), SPL (CELL SIZE, IN), NSP (# CELL), MAT #
 0.000E+00 0.000E+00
 2.000E-06 1.896E+03
 4.000E-06 2.136E+03
 6.000E-06 2.336E+03
 8.000E-06 2.515E+03
 1.000E-05 2.664E+03
 1.200E-05 2.794E+03
 1.400E-05 2.894E+03
 1.600E-05 2.984E+03
 1.800E-05 3.053E+03
 2.000E-05 3.293E+03
 2.200E-05 3.503E+03
 2.400E-05 3.692E+03
 2.600E-05 3.832E+03
 2.800E-05 3.922E+03
 3.000E-05 3.961E+03
 3.200E-05 3.951E+03
 3.400E-05 3.892E+03
 3.600E-05 3.822E+03
 3.800E-05 3.743E+03
 4.000E-05 3.662E+03
 4.200E-05 3.572E+03
 4.400E-05 3.482E+03
 4.600E-05 3.393E+03
 4.800E-05 3.313E+03
 5.000E-05 3.234E+03
 5.200E-05 3.163E+03
 5.400E-05 3.093E+03
 5.600E-05 3.034E+03
 5.800E-05 2.973E+03
 6.000E-05 2.914E+03
 6.200E-05 2.863E+03
 6.400E-05 2.824E+03
 Lines were omitted for the Figure.
 1.550E-03 6.775E+02
 1.590E-03 6.865E+02
 1.660E-03 6.945E+02
 1.720E-03 7.374E+02
 1.780E-03 7.235E+02
 1.000E-02 7.235E+02

FIGURE A-1 INPUT FILE FOR ONED SIMULATION OF DISKO-1 USING THE PRESSURE HISTORY
 FROM THE BLAST PRESSURE GAGES

JA-6391-104

TEST OF SRI ONED TO SIMULATE THE DISKO-1 EVENT, FINAL SET OF DATA
 [B6391.ONED]DISKO1E.OND IS DATA, [*.]*RONED.COM, [*.]*MONED.EXE

1 DISKO 1 FOR WES, ONED SIMULATION OF THE EXPLOSIVE LAYER.

998, 0, 0.004, 400, 1. NOCOMP, BOTTOM BOUNDARY INDICATOR, TEND, NSTOP, GACC
 4, 0, 0, 0, 4, 0, NLAYRS, MISP, NSP, MATTEN, NMIRLS, NFORCE
 7, 1, 3, 0, 0, 0, 0, 100, 1, 1, 1, 0.004 NNODE, KNTROL (7), NHIST, MVT, NSV, KVEL, TND
 3.142, 3.542, 4.542, 5.542, 6.542, 7.542, 8.542 DEPTHS FOR HIST. (FT)
 1000., 0.0, 60., 0., 0., 0., 0. SK (7)
 'SAND', 106.3, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR SAND
 1.36E+9, 1.0E+4, 0.274 SHMOD, YIELD, TANPHI
 7, 3, 'P', 'D', 1., 1.0 SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
 4.E7, 7.E7, 1.7E8, 4.5E8, 1.E9, 0., FLUME SAND, P, DYN/CM² (K=2.92E9)
 1.7268, 1.7458, 1.793, 1.893, 2.065, 2.023 DENSITIES, G/CM³
 'EXPL1', 2.7406, 0.05, 4., 0.0, 0, 1, 0, 0 GAM, DAMP, TENS, POR, PR, SOL, VAR (EXPL)
 0., 0., 0. SHMOD, YIELD, TANPHI
 'POLY', 4.5E10, 0.2 Q AND GAMMA
 'AIR', .072946, 0.05, 4., 0.0, 0, 1, 0, 0 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR AIR
 0., 0., 0. SHMOD, YIELD, TANPHI
 'POLY', 2.139E9, 0.4 Q AND GAMMA
 'OVERBDN', 95.035, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, NVAR
 1.36E+9, 1.0E+4, 0.274 SHMOD, YIELD, TANPHI
 10, 1, 'P', 'D', 1.E7, 1.0 SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
 8., 39., 69., 122., 230., 400., 100., 30., 0., SAND OVERBDN MODEL, PRESS., MPA
 1.603, 1.792, 1.904, 2.031, 2.176, 2.343, 2.290, 2.256, 2.207 DENSITIES
 2.667 1.0 0, 4 32 INCHES OF SAND OVERBURDEN
 2.792 0.15 0, 2 1.5 INCHES OF PRIMACORD-AIR MIXTURE
 3.042 3.0 0, 3 3.0 INCHES OF AIR (REPRESENTING A MIXTURE OF FOAM AND AIR)
 11.375 0.4 0, 1 100 INCHES OF FLUME SAND

FIGURE A-2 INPUT FILE FOR THE ONED SIMULATION OF DISKO-1 USING THE EXPLICIT TREATMENT OF THE EXPLOSIVE LAYER

JA-6391-105

TEST OF SRI ONED TO SIMULATE THE DISKO-1 EVENT, FINAL SET OF DATA
 [B6391.ONED]DISKO1P.OND IS DATA, [*.*)RONED.COM, [*.*)MONED.EXE
 1 DISKO 1 FOR WES, ONED SIMULATION WITH PRESSURES FROM GUINSY CALC. 7/10/85
 998, 0, 0.004, 400, 1. NOCOMP, BOTTOM BOUNDARY INDICATOR, TEND, NSTOP, GACC
 1, 0, 0, 0, 1, 79, NLAYRS, MISP, NSP, MATTEN, NMTRLS, NFORCE
 10, 1, 3, 0, 0, 0, 0, 100, 1, 1, 1, 0.004 NNODE, KNTROL (7), NHIST, MVT, NSV, KVEL, TND
 0.1, 0.25, 0.5, 1.5, 1.67, 1.83, 2.5, 3.5, 4.5, 5.5 DEPTHS FOR HIST. (FT)
 1000., 0.0, 60., 0., 0., 0., 0. SK (7)
 'SAND', 106.3, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENSIN, POR, PR, SOL, VAR FOR SAND
 1.36E+9, 1.0E+4, 0.274 SHMOD, YIELD, TANPHI
 7, 3, 'P', 'D', 1., 1.0 SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
 4.E7, 7.E7, 1.7E8, 4.5E8, 1.E9, 0., FLUME SAND, P, DYN/CM² (K=2.92E9)
 1.7268, 1.7458, 1.793, 1.893, 2.065, 2.023 DENSITIES, G/CM³
 8.333 0.4, 0, 1 DB (DIST TO BOT, FT), SPL (CELL SIZE, IN), NSP (# CELL), MAT #
 0.000E+00 0.000E+00
 2.000E-06 2.326E+02
 4.000E-06 4.651E+02
 6.000E-06 6.977E+02
 1.000E-05 8.335E+02
 1.400E-05 9.678E+02
 1.700E-05 1.100E+03
 2.100E-05 1.229E+03
 2.500E-05 1.354E+03
 2.900E-05 1.477E+03
 3.300E-05 1.595E+03
 3.700E-05 1.711E+03
 4.000E-05 1.825E+03
 4.400E-05 1.938E+03
 4.800E-05 2.051E+03
 5.200E-05 2.164E+03
 5.600E-05 2.278E+03
 5.900E-05 2.393E+03
 6.300E-05 2.514E+03
 6.700E-05 2.636E+03
 7.100E-05 2.762E+03
 7.500E-05 2.894E+03
 7.900E-05 3.028E+03
 8.200E-05 3.167E+03
 9.400E-05 2.914E+03
 1.040E-04 2.697E+03
 1.130E-04 2.512E+03
 1.230E-04 2.354E+03
 1.330E-04 2.217E+03
 Lines were omitted here for the figure
 7.900E-04 7.078E+02
 8.110E-04 6.696E+02
 1.000E-02 6.696E+02

FIGURE A-3 INPUT FILE FOR THE ONED SIMULATION OF DISKO-1 WITH THE PRESSURE HISTORY
 CONSTRUCTED BY GUINSY FROM THE STRESS GAGE RECORDS

JA-6391-106

TEST OF SRI ONED TO SIMULATE A FICTITIOUS SITUATION TO DEMONSTRATE
THE TREATMENT OF SOLIDS, POROUS MATLS, TABULAR ISENTROPE, AND A GAP.

```

1 ONED SIMULATION OF SIX LAYER PROBLEM, INCLUDING A GAP
998, 0, 0.004, 400, 1. NOCOMP, BOTTOM BOUNDARY INDICATOR, TEND, NSTOP, GACC
6, 0, 0, 0, 5, 0, NLAYRS, MISP, NSP, MATTEN, NMIRLS, NFORCE
7, 1, 3, 0, 0, 0, 0, 100, 1, 1, 1, 0.004 NNODE, KNTROL (7), NHIST, MVT, NSV, KVEL, TND
3.142, 3.542, 4.542, 5.542, 6.542, 7.542, 8.542 DEPTHS FOR HIST. (FT)
1000., 0.0, 60., 0., 0., 0., 0. SK (7)

'SAND', 106.3, 0.1, 4., 0.0, 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR FOR SAND
1.36E+9, 1.0E+4, 0.274 SHMOD, YIELD, TANPHI
7, 3, 'P', 'D', 1., 1.0 SSONED: KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
4.E7, 7.E7, 1.7E8, 4.5E8, 1.E9, 0., FLUME SAND, P, DYN/CM^2 (K=2.92E9)
1.7268, 1.7458, 1.793, 1.893, 2.065, 2.023 DENSITIES, G/CM^3

'STEEL', 489.8, 0.1, 4., 0.0, 0, 0, 1, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR -----
1.344E+10, 7.034E+11, 0. SHMOD (DYN/CM2), YIELD (DYN/CM2), TANPHI
2.203E7, 7.477E7, 8.E8 EQSTC, EQSTD, EQSTS FOR STEEL

'EXPL1', 2.7406, 0.05, 4., 0.0, 0, 1, 0, 0 GAM, DAMP, TENS, POR, PR, SOL, VAR (EXPL)
0., 0., 0. SHMOD, YIELD, TANPHI
'POLY', 4.5E10, 0.2 Q AND GAMMA

'ANFO', 48.8, 0.05, 4., 0.0, 0, 1, 0, 0 GAM, DAMP, TENS, POR, PR, SOL, VAR (EXPL)
0., 0., 0. SHMOD, YIELD, TANPHI
'TAB', 4.5E10, 0.2 Q AND GAMMA -NOT USED-
10, 'VOLUME', 'LOG' IMAX, LABEL, TYPE FOR TABULAR EQ. OF STATE
9.142E-01, 6.500E+10, 1.5322E00, 1.871E+10, 2.568E+00, 5.932E+09
4.304E+00, 2.060E+09, 7.214E+00, 7.790E+08, 1.2091E01, 3.174E+08
2.0265E01, 1.372E+08, 3.3966E01, 6.200E+07, 5.6928E01, 3.100E+07
1.000E+02, 1.530E+07

'PLYWUD', 42., 0.05, 4., 0., 1, 0, 0, 2 GAM, DAMP, TENS, POR, PR, SOL, VAR
0., 0., 0. SHMOD, YIELD, TANPHI
8, 1, 'MODU', 'STRA', 6.914627E4, 1. KFIN, NTYPE, LPRESS, LDENS, PUNIT, DUNIT
65000., 16000., 8000., 18000., 21500., 30000., 2.E5, 2.E5, PLYWOOD, MODULI
1.0, 2.5, 7., 9.5, 12.75, 100., 0.0, STRAINS

2.667 1.0 0, 2 32 INCHES OF STEEL (OVERBURDEN)
2.792 0.15 0, 3 1.5 INCHES OF EXPL1 (PRIMACORD-AIR MIXTURE)
3.042 3.0 0, 4 3.0 INCHES OF ANFO
3.1045 0.75 0, 5 3/4 INCH OF PLYWOOD
3.1253 0.25 0, 0 1/4 INCH GAP
11.4587 0.4 0, 1 100 INCHES OF FLUME SAND

```

Notes: The blank lines between materials were introduced for readability.
They must not appear in the input file.

FIGURE A-4 INPUT FILE FOR A SYNTHETIC PROBLEM WITH A SOLID, A GAP, A POROUS MATERIAL,
AND AN EXPLOSIVE REQUIRING A TABULAR ISENTROPE

JA-6391-107

Appendix B

GLOSSARY OF TERMS USED IN THE PROGRAM

ACC(J)	Acceleration of the J th node, cm/s ² .
ALPHA(J)	Ratio of area at top to area at node J (attenuation factor).
BETA	Array containing problem title, input.
CELMAS(J)	Mass of material in cell J between nodes J and J+1, g/cm ² .
COM(L)	Array containing extra variables for each cell. Used with the NVAR(M) and LVAR(J) arrays.
DAMPL(M)	Coefficient for the linear artificial viscosity term, input, dimensionless.
DAMPQ(M)	Coefficient for the quadratic artificial viscosity term, initialized at 4.0, dimensionless.
DENS(J)	Density of the J th cell, g/cm ³ .
DENSCAL	Conversion factor for changing densities from pcf to g/cm ³ : 0.0160185 g/cm ³ /(pcf).
DEP()	Depth in feet at which historical information is requested, input.
DISTCAL	Conversion factor for changing distances from inches to cm: 2.54 cm/in.
DT	Time step for each computational cycle, computed in the program, sec.
DUNIT	Unit conversion factor to transform the input values of density to g/cm ³ . (Used for SSONED input)
EL(i,M)	An array of moduli for loading and unloading in a porous material treated by SSONED. EL(i,M) is used in the strain interval from SL(i,M) to SL(i+1,M).
ELMOD(M)	Longitudinal elastic modulus for each material, computed in the program, dyn/cm ² .
EQSTC(M)	Bulk modulus used with the solid model, EQST. Input in psi.

EQSTD(M) Second coefficient in the series expansion for bulk modulus used in EQST. Input in psi.

EQSTG(M) Gruneisen ratio = $\Gamma = \gamma - 1$ where γ is the polytropic exponent for an explosive, input to EXPLODE, dimensionless.

EQSTS(M) Third coefficient in the series expansion for bulk modulus used in EQST. Input in psi.

FL(i,M) An array of pressures for the loading and unloading curves used in SSONED. FL(i,M) is the pressure at the strain SL(i,M) in dyn/cm².

G Acceleration of gravity, 981.456 cm/s².

GACC Indicator for initializing the geostatic stress option and using a gravity effect throughout the calculation, input, dimensionless at input. When GACC is read in with a nonzero value, it is reset to $G = 981.456 \text{ cm/s}^2$.

GAM(M) Initial density for each material, input, lb/ft³

KNT Index used with the KNTROL() array to indicate plots desired. If KNT = 0, no plotting occurs.

KNTROL() Indicator array for the type of plots required:

- = 0 for no plots. KNT is set to zero.
- = 1 for stress - time plot.
- = 2 for acceleration - time plot.
- = 3 for particle velocity - time plot.
- = 4 for node location - time plot.
- = 5 for strain - time plot.
- = 6 for impulse - time plot.
- = 7 for average strain - time plot.

KVEL Input indicator to control units for printing particle velocity: 0 for in./s, > 0 for m/s.

LB Indicator for the boundary condition at the bottom of the last layer:

- = 0 means free boundary
- = 1 means rigid boundary

LDENS Label with the same meaning as LPRESS.

LPRESS Label for type of quantities to be used for pressure-strain relation for SSONED. Four labels are provided: "D" for density, "S" for strain, "M" for modulus, and "P" for pressure. (Only the first character is interpreted.)

LVAR(J) A locator array for the Jth cell providing the starting position for extra variables in the COM array. For L = LVAR(J), the first variable is at COM(L).

MASSCAL Conversion factor to change masses from pounds to grams: 2.204624E-3 lb/g.

MAT(J) Material number of the Jth cell, dimensionless.

MATTEN Indicator for the use of the attenuation factors, ALPHA(J), input. 0 for no attenuation; > 0 for use of the attenuation factors.

MISP Indicator used to control cell spacing (input):
 = 0 uniform spacing given by SPL.
 = 1 cell spacing is read in for each layer.
 > 1 and NSP > 0, uniform cell sizes and NSP is the number of cells.

MVT Parameter controlling printing of the maximum value table, input. Zero value omits printing.

N Total number of nodes in the problem.

NAME(M) Four-letter title for each material, input.

NFORCE Number of pressure-time points in the loading function, input.

NHIST Frequency for snapshot listings of current status of the requested cells and nodes, input.

NLAYRS Number of layers used in the input. A gap counts as a layer.

NMTRLS Number of materials to be read in for the calculation, input.

NNODE Number of nodes or cells at which output is required, input.

NOCOMP Parameter controlling plot scales:
 < 998 means compute and plot at computed scales.
 = 998 means compute and plot at input scales.
 > 998 means plot at input scales, no computation.

NOUT() Array containing the numbers of cells or nodes at which historical data are requested.

NPOR(M) Indicator for a porous material, input; nonzero means SSONED is to be called.

NPR(M)	Indicator for a pressure model for a material, here used for an explosive, input: nonzero means EXPLODE is to be called.
NSOL(M)	Indicator for a solid model, input: nonzero means EQST is to be called.
NSP	Number of cells in each layer (input). Used either with MISPL = 1 for nonuniform layout or MISPL = 2 for a uniform layout.
NSTEP	Current count of the computational cycle for the problem.
NSTOP	Number of computational cycles at which problem will be stopped, input, dimensionless.
NSV	Parameter controlling saving of a plot tape for future plotting, input.
NTYPE(M)	Indicator for type of unloading process used in SSONED. Input to SSONED. See Section III.C.
NVAR(M)	Number of extra variables in the COM array to be provided for each cell of the material.
NVELCAL	Label used with velocity output to indicate either in./s or m/s.
PHIST()	Array containing the pressures for the applied pressure on the top boundary, input in psi.
PRESCAL	Conversion factor for changing pressure or stress in psi to dyn/cm ² : 6.914627E4 dyn/cm ² /psi.
PUNIT	Unit conversion factor to transform from the input values of modulus and pressure to dyn/cm ² (used for SSONED input).
PRESS(J)	Pressure in the J th cell, dyn/cm ² .
QEXPL(M)	Chemical energy of an explosive, input in erg/g.
RHO(M)	Initial density of the material, g/cm ³ .
RMECH(J)	Mechanical stress in the J th cell, dyn/cm ² .
SAVDMX()	A two-dimensional array for the maximum mechanical stress, acceleration, particle velocity, position, and strain that occur at the J th cell and node during the computation.
SHMOD(M)	Shear modulus for each material, input, dyn/cm ² .

SIGI(J)	Thermodynamic stress in the direction of propagation, dyn/cm ² .
SL(i,M)	Strain array describing loading and unloading curves for a porous material treated by SSONED. These curves are subdivided into a number of strain intervals: the SL(i,M) bound these intervals.
SSPEED(J)	Sound speed in the J th cell, cm/s.
TANPHI(M)	Tangent of the coefficient of friction for a yield strength model with Coulomb friction, input, dimensionless.
TEND	Stop time for the calculation, input, s.
TENSIL(J)	Tensile strength of the J th cell, dyn/cm ² .
TENSN(M)	Tensile strength of the material, input, psi. A zero value is interpreted as infinite strength.
THIST()	Array containing the times of the pressure history applied to the top boundary, input in s.
TIME	Current problem time during a calculation, s.
TMAX()	A two-dimensional array containing the times at which the maxima SAVDMX occur.
TNDPLT	Duration of plotting time, input, s.
VEL(J)	Particle velocity of the J th node, cm/s.
VELCAL	Conversion for changing particle velocity in in./s to cm/s.
X(J)	Position of the J th node, cm.
Y(J)	Yield strength of the J th cell, dyn/cm ² .
YIELD(M)	Yield strength of the material, input in dyn/cm ² .

Appendix C

INSERTION PROCEDURE

As new material models are generated, they can be added to ONED for performing wave propagation calculations. This appendix describes the procedure for inserting material model subroutines.

A wave propagation code normally has four main categories of operations: reading the input data, initializing a finite difference grid, performing calculations for each time increment at each grid point, and printing the computed information. A material model subroutine may be involved in all or some of these operations. Call statements must be provided in ONED at appropriate locations to accomplish these tasks. Also, the new subroutine should be provided with separate sections for each operation and an indicator to show which operation to perform. For example, in SSONED the formal parameter NCALL indicates the operation required, as follows:

NCALL = 0	Initialize the routine and read data for one material
1	Calculate pressure

The CALL for NCALL = 0 is in READIT. For NCALL = 1, the CALL statement is in CMPUTE.

At the point of insertion of the CALL statement, four elements are provided:

- (1) The appropriate branching statements are needed to switch to the new model when it is required. SSONED was treated as a porous model and designated by NPOR(M) = 1. Then branching statements in READIT and CMPUTE were written to route the computation to SSONED.
- (2) Variables must be initialized, calibrated, or given sign changes just preceding the CALL statement.
- (3) The CALL statement is provided.
- (4) Some variables may need to be reset following the calculations in the routine. Then a jump is provided to the appropriate section of READIT or CMPUTE to continue the calculation.

Items (2), (3), and (4) are discussed further below following introduction of a CALL statement.

A sample of the calling procedure for SHEAR2 is listed here because it encompasses most of the problems that might be encountered in an insertion. SHEAR2 is a stress-strain relation describing elastic-plastic behavior with shear banding. It was written for two-dimensional calculations and hence employs most of the stress and strain tensor components. The pressure is positive in compression in SHEAR2, but the stress is positive in tension. Twenty-five extra variables (from the COM array) are required by SHEAR2. In addition, most of the basic material properties are transmitted to SHEAR2 in an array called ESC. Thus, it is necessary to dimension ESC and to fill it with properties before the CALL. Thus, the first step is the dimension statement:

```
DIMENSION ESC(20,20)
```

The first subscript indicates the material parameter, and the second subscript is the material number. The array could be filled with a set of statements as follows:

```
IF (ESC(1,M) .GT. 0.) GO TO 25
ESC(2,M) = RHO(M)
ESC(2,M) = EQSTC(M)
ESC(3,M) = EQSTD(M)
ESC(4,M) = EQSTS(M)
ESC(5,M) = SHMOD(M)
ESC(9,M) = Grüneisen ratio
```

Note that the Hugoniot coefficients (used in EQST) must be provided to CMPUTE before this equivalencing can work. In the example that follows, the stress, strain, and other variables must be prepared for the CALL to SHEAR2. The first of the preparatory statements defines NCALL, which is 2 for the usual stress calculation, but 3 for a stress calculation plus a request that SHEAR2 print its shear banding quantities.

```
NCALL = 2
```

```
If (NHIST .GT. 0 .AND. MOD(N,NHIST) .EQ. 0) NCALL = 3
```

The input file is set to 5, although it is not used here.

```
INPUT = 5
```

An integer indicator is used with SHEAR2. It may be reset within the subroutine.

```
IH3 = COM(L)
```

The deviator stresses are initialized with the sign convention that stress is positive in tension.

```
SX = - SIGDEV
```

```
SY = - 0.5*SIGDEV
```

```
SZ = SY
```

```
TXY = 0.
```

The strain increments are initialized to be positive in tension, also.

```
EX = -2.*(DENS(J)-DOLD)/(DENS(J)+DOLD)
```

```
EY = 0.
```

```
EZ = 0.
```

```
EXY = 0.
```

The rotation quantities are set at zero because there is no rotation in ONED.

```
ROT = 0.
```

```
DROT = 0.
```

The melt energy should be initialized in erg/g. F is the thermal strength reduction factor.

```
EMELT = 1.E10
```

```
F = 1.
```

```
CALL SHEAR2 (NCALL, INPUT, MATL, J, J, IH3, SX, SY, SZ,  
1 TXY, PRESS(J), COM(L+1), DENS(J), DOLD, DT, EH, EOLD,  
2 COM(L+2), EMELT, COM(L+3), EX, EY, EZ, EXY, F, Y(J),  
3 COM(L+4), ROT, DROT, ESC, COM(L+5) )
```

In the CALL statement, J appears twice because the statement is for a two-dimensional code. DENS(J) and DOLD are the new density and the density at the previous cycle, DT is the time step, and EH and EOLD are the new internal energy and the internal energy at the previous cycle. The use of the COM array is illustrated in the CALL. The first member of the array for the Jth cell, COM(L), was used earlier for IH3. Now the next four members of the array are used to represent individual variables. The last of the formal parameters, COM(L+5), is just the first of a set of 20 variables used in one group by SHEAR2. After the CALL statement, the stress and IH3 are returned to their standard locations:

SIGDEV = -SX

COM(L) = IH3

Before insertion into ONED, any new subroutine should be thoroughly tested by itself. For example, SSONED was tested with a one-page program called TSS, which was written for these tests. TSS first calls SSONED (in the same way that READIT would) to read and initialize variables. Then TSS calls SSONED in a 50-step loop (like CMPUTE would) to compute pressure. In this loop density changes were provided that caused several cycles of loading, unloading, and reloading. The resulting pressure-density path was plotted to verify that the correct path was followed.

Following insertion of a new material model into ONED, it is a good plan to run a simple problem with NHIST set to 5 or 10 to determine whether the routine is performing satisfactorily.

Appendix D

LISTING OF ONED AND ITS SUBROUTINES

The following listing contains all the subroutines used with ONED, listed in alphabetical order after the main program. Included are ONED, the COMMON block used with all the routines, CMPUTE, EOSTAB, EQST, EXPLODE, GRAFONED, HISTRY, PLOTIT, READIT, and SSONED. HISTRY contains a number of ENTRYs, which make them also appear as subroutine CALLs in the other routines. These ENTRYs are FORCI, PLOTSC, RECORD, SCRIBE, UPDATE, VALMAX, and ZERO.

GRAFONED calls an SRI graphing routine called GRAPH4. A user will need to replace these CALLs in GRAFONED by CALLs appropriate to the graphics software at his computer.

PROGRAM ONED	ONED	1
C ONE-DIMENSIONAL WAVE PROPAGATION PROGRAM FOR PLANAR FLOW.	ONED	2
C BASED ON ONED154 OF WES. PROGRAMMERS ROHANI AND DAVIS. REVISED BY	ONED	3
C RADHAKRISHNAN, SEPT.69.MODIFIED BY SEAMAN IN SEPTEMBER 1984 TO INCLUDE	ONED	4
C AN EXPLICIT LEAP-FROG SOLUTION SCHEME WITH ARTIFICIAL VISCOSITY, LARGE	ONED	5
C DEFORMATIONS, AND SEPARATE SUBROUTINES WITH STRESS-STRAIN RELATIONS	ONED	6
C FOR SOILS, SOLIDS, AND EXPLOSIVES.	ONED	7
C	ONED	8
C MAIN PROGRAM CALLS ROUTINES READIT, CMPUTE AND PLOTIT.	ONED	9
C	ONED	10
C TIME SEQUENCING	ONED	11
C STRESS, ACCEL, POSITION, STRAIN ARE AT TIME AND TIME+DT	ONED	12
C VELOCITY IS DEFINED AT TIME+DT/2	ONED	13
C ORDER OF CALCULATIONS IS:	ONED	14
C ACCEL, VELOCITY, POSITION, STRESS	ONED	15
C	ONED	16
C INCLUDE '\$DISK3:[SEAMAN.ONED]ONEDCOM.FOR'	ONED	17
C	ONED	18
C ITAPE = 0	ONED	19
C 10 REWIND 7	ONED	20
C ***** READIT READS, DIGESTS, AND OUTPUTS THE INPUT DATA *****	ONED	21
C CALL READIT	ONED	22
C IF (NOCOMP .EQ. 999) GO TO 30	ONED	23
C ***** CMPUTE COMPUTES AND PRINTS THE GROUND MOTIONS AND STRESSES AND	ONED	24
C RECORDS THE INFORMATION ON TAPE FOR PLOTIT *****	ONED	25
C CALL CMPUTE	ONED	26
C READ (5,950,END=22) ID	ONED	27
C GO TO 24	ONED	28
C 22 ID = 0	ONED	29
C IF ID = 0, END OF RUN; OTHERWISE, ANOTHER SET OF DATA FOLLOWS.	ONED	30
C 24 IF (KNT .LE. 0) GO TO 55	ONED	31
C 30 REWIND 7	ONED	32
C ***** PLOTIT USES THE DATA FROM THE TAPE GENERATED IN CMPUTE TO	ONED	33
C GENERATE A PLOT TAPE OF THE GROUND MOTIONS AND STRESSES *****	ONED	34
C CALL PLOTIT	ONED	35
C ITAPE = 1	ONED	36
C IF (NSV .EQ. 0) GO TO 55	ONED	37
C SAVE TAPE BY ADDING SUITABLE CODING AT THIS POINT	ONED	38
C 55 IF (KNT .EQ. 0) WRITE (6,934)	ONED	39
C CALL SCRIBE	ONED	40
C IF (ID .EQ. 0) GO TO 70	ONED	41
C WRITE (6,965)	ONED	42
C GO TO 10	ONED	43
C 70 WRITE (6,970)	ONED	44
C STOP ' ONED, NORMAL END'	ONED	45
C FORMAT STATEMENTS	ONED	46
C 934 FORMAT (// 9X,20H NO PLOTS WERE MADE)	ONED	47
C 950 FORMAT (78X,I2)	ONED	48
C 965 FORMAT (// 9X,16H END OF PROBLEM)	ONED	49
C 970 FORMAT (// 9X,12H END OF RUN)	ONED	50
C END	ONED	51

C	[SEAMAN.ONED]ONEDCOM.FOR	COM	1
	IMPLICIT REAL*8 (A-H,O-Z)	COM	2
	PARAMETER (NON = 300)	COM	3
	CHARACTER*8 NAME	COM	4
	COMMON G,GACC,NOCOMP,FORCE,NMTRLS,NSTEP,NSTOP	COM	5
	COMMON /ID/ BETA(12),IDENT	COM	6
	COMMON /PLOT/ DEP(20),KNTROL(7),KNT,NOUT(20),NSV,SK(7),TNDPLT	COM	7
	COMMON /TIME/ DT,TIME,TEND	COM	8
	COMMON /PROP/ DAMPL(6),DAMPQ(6),ELMOD(6),GAM(6),NAME(6),NPR(6),	COM	9
@	NPOR(6),NSOL(6),NVAR(6),RHO(6),SHMOD(6),SQRTNP(6),TANPHI(6),	COM	10
@	YIELD(6)	COM	11
	COMMON /LAYER/ LB,N,NSP	COM	12
	COMMON /EDIT/ KVEL,LCOUNT,LPRNT,MVT,NHIST,NNODE	COM	13
	COMMON /TENS/ TENS(6),NTENS(6)	COM	14
	COMMON /ARRAY/ ACC(NON),ALPHA(NON),CELMAS(NON),DENS(NON),	COM	15
@	PRESS(NON),RMECH(NON),SIG1(NON),SSPEED(NON),TENSIL(NON),	COM	16
@	VEL(NON),X(NON),Y(NON),MAT(NON)	COM	17
	COMMON /MAX/ SAVDMX(5,NON),TMAX(5,NON)	COM	18
	COMMON /COM/ LVAR(NON),COM(5000)	COM	19
	COMMON /CALIB/ DENS(6),DISTCAL,PRESCAL,VELCAL,AVELCAL,	COM	20
@	NVELCAL	COM	21
	COMMON /FORC/ NFORCE,PHIST(900),THIST(900)	COM	22

	SUBROUTINE CMPUTE	CMPT	1
C	"CMPUTE" CONTAINS THE MAIN TIME-STEPPING LOOP OVER ALL CELLS,	CMPT	2
C	COMPUTES ACCEL., VELOCITY, DISPL., AND CALLS THE	CMPT	3
C	STRESS-STRAIN SUBROUTINES TO COMPUTE STRESS.	CMPT	4
C	DETERMINES THE TIME STEP SIZE,	CMPT	5
C	CALLS "FORCI" TO COMPUTE DRIVING STRESS ON FIRST NODE,	CMPT	6
C	CALLS "HISTRY" FOR LISTING OF VALUES AT CERTAIN TIMES,	CMPT	7
C	CALLS "UPDATE" TO STORE MAXIMUM VALUES OF VARIABLES,	CMPT	8
C	CALLS "VALMAX" TO PRINT THE MAXIMUM VALUES, AND	CMPT	9
C	CALLS "RECORD" TO STORE HISTORICAL DATA FOR PRINTING.	CMPT	10
C		CMPT	11
C	- - - DEFINITIONS - - -	CMPT	12
C	ACC(I) = ACCELERATION AT NODE I, (IN/SEC^2)	CMPT	13
C	ALPHA(I)= ATTENUATION FACTOR TO APPROXIMATE MULTI-DIM FLOW	CMPT	14
C	CELMAS(I)= MASS OF CELL I BETWEEN NODES I AND I+1, (G/CM^2)	CMPT	15
C	DENS(I) = DENSITY OF CELL I BETWEEN NODES I AND I+1, (G/CM^3)	CMPT	16
C	RMECH(I)= MECHANICAL STRESS IN CELL I, (DYN/CM^2)	CMPT	17
C	SIG1(I) = THERMODYNAMIC STRESS IN CELL I, (DYN/CM^2)	CMPT	18
C	SSPEED(I) = SOUND SPEED IN CELL I, (CM/SEC)	CMPT	19
C	VEL(I) = VELOCITY AT NODE I, (CM/SEC)	CMPT	20
C	X(I) = POSITION OF NODE I, (CM)	CMPT	21
C	Y(I) = CURRENT YIELD STRENGTH OF NODE I, (DYN/CM^2)	CMPT	22
C		CMPT	23
C	NPRNT AND NHIST CONTROL WHEN GROUND MOTIONS ARE (ARE NOT) PRINTED	CMPT	24
C		CMPT	25
C	INCLUDE ' [SEAMAN.ONED]ONEDCOM.FOR'	CMPT	26
C		CMPT	27
C	*****	*****CMPT	28
C	TIME = 0.	CMPT	29
C	FORCE = 0.	CMPT	30
C	SAFETY = 0.8	CMPT	31
C	NPRNT = NHIST - 1	CMPT	32
C	BEGIN LOOP OVER TIME STEPS	CMPT	33
C	DT = 1.E-12	CMPT	34
C	DTOLD = DT	CMPT	35
C	NSTEP = 0	CMPT	36
C	100 NSTEP = NSTEP+1	CMPT	37
C	TIME = TIME+DT	CMPT	38
C	DTHALF = 0.5*(DT+DTOLD)	CMPT	39
C	DTMIN = 1000.	CMPT	40
C	CALL "FORCI" TO OBTAIN THE DRIVING STRESS ON NODE 1.	CMPT	41
C	IF (NFORCE .GT. 0) CALL FORCI	CMPT	42
C	*****	*****CMPT	43
C		CMPT	44
C	COMPUTE ACCELERATION, VELOCITY AND POSITION	CMPT	45
C		CMPT	46
C	*****	*****CMPT	47
C	ACC(1) = (FORCE-RMECH(1))/(0.5*CELMAS(1)) + GACC	CMPT	48
C	VOLD = VEL(1)	CMPT	49
C	XOLD = X(1)	CMPT	50
C	VEL(1) = VEL(1)+DTHALF*ACC(1)	CMPT	51
C	X(1) = X(1)+VEL(1)*DT	CMPT	52
C	*****	*****CMPT	53
C	LOOP OVER ALL NODES FOR ACC, VEL, POSITION	CMPT	54
C	*****	*****CMPT	55
C	ISPALL = 0	CMPT	56
C	DO 200 J=2,N	CMPT	57
C	REAR BOUNDARY CONDITIONS, LB=1 MEANS RIGID, LB=0 MEANS FREE	CMPT	58
C	IF (J .GE. N .AND. LB .EQ. 1) GO TO 200	CMPT	59
C	IF (J .GE. N .AND. LB .EQ. 0) GO TO 160	CMPT	60
C	TEST FOR INTERFACE CONDITIONS	CMPT	61
C	120 IF (MAT(J-1) .NE. 0 .AND. MAT(J) .NE. 0) GO TO 160	CMPT	62

C	TEST FOR RIGHT SIDE OF INTERFACE, MAT(J-1) = 0	CMPT	63
	IF (MAT(J-1) .EQ. 0) GO TO 155	CMPT	64
C	TEST CONDITIONS ON LEFT SIDE OF INTERFACE, MAT(J) = 0	CMPT	65
	IF (X(J+1)-X(J) .GT. 1.E-8) GO TO 160	CMPT	66
C	COMBINED INTERFACE, COMPUTED FOR J = NODE ON LEFT	CMPT	67
	ACC(J) = (RMECH(J-1)-RMECH(J+1)) / (0.5*(CELMAS(J-1)+CELMAS(J+1)))	CMPT	68
	@ + GACC	CMPT	69
	VEL(J) = (CELMAS(J-1)*VEL(J)+CELMAS(J+1)*VEL(J+1)) /	CMPT	70
	@ (CELMAS(J-1)+CELMAS(J+1)) + DTHALF*ACC(J)	CMPT	71
	IF (ABS(VEL(J)) .LE. 1.) VEL(J) = 0.	CMPT	72
	X(J) = X(J)+VEL(J)*DT	CMPT	73
	ACC(J+1) = ACC(J)	CMPT	74
	VEL(J+1) = VEL(J)	CMPT	75
	X(J+1) = X(J)	CMPT	76
	ISPALL = 1	CMPT	77
	GO TO 200	CMPT	78
C	RIGHT SIDE OF INTERFACE	CMPT	79
	155 IF (ISPALL .EQ. 1) GO TO 200	CMPT	80
C	*****	CMPT	81
C	STANDARD ROUTE FOR INTERIOR NODE OR SPALLED NODE	CMPT	82
	160 ACC(J) = (RMECH(J-1)-RMECH(J)) / (0.5*(CELMAS(J-1)+CELMAS(J))) +GACC	CMPT	83
	VEL(J) = VEL(J)+ACC(J)*DTHALF	CMPT	84
	IF (ABS(VEL(J)) .LE. 0.01) VEL(J) = 0.	CMPT	85
	X(J) = X(J)+VEL(J)*DT	CMPT	86
	ISPALL = 0	CMPT	87
	200 CONTINUE	CMPT	88
C		CMPT	89
C	*****	CMPT	90
C	LOOP OVER EACH CELL TO COMPUTE STRAIN AND STRESS	CMPT	91
C	*****	CMPT	92
C		CMPT	93
	IF (NHIST .GT. 0 .AND. MOD(NSTEP,NHIST) .EQ. 0)	CMPT	94
	@ WRITE (6,1644) BETA,NSTEP,TIME	CMPT	95
	1644 FORMAT (/10X,12A6/' SNAPSHOT AT NSTEP =',I4,', TIME =',1PE12.5/	CMPT	96
	@ 8H J MAT,5X,1HL,11X,1HX,9X,3HVEL,5X,5HRMECH,6X,4HSIG1,	CMPT	97
	@ 5X,5HDSTRS,5X,5HPRESS,6X,4HDENS,4X,6HSSPEED,4X,6HSIGDEV,9X,1HY/	CMPT	98
	@ 24X,2HCM,6X,6HCM/SEC,3X,7HDYN/CM2,3X,7HDYN/CM2,3X,7HDYN/CM2,	CMPT	99
	@ 3X,7HDYN/CM2,5X,5HG/CM3,4X,6HCM/SEC,3X,7HDYN/CM2,3X,7HDYN/CM2)	CMPT	100
	DO 650 J = 1,N	CMPT	101
	RMOLD = RMECH(J)	CMPT	102
	DOLD = DENS(J)	CMPT	103
	SOLD = SIG1(J)	CMPT	104
	POLD = PRESS(J)	CMPT	105
	DSTRS = 0.	CMPT	106
	SPEED = SSPEED(J)	CMPT	107
	L = LVAR(J)	CMPT	108
	SIGDEV = SIG1(J)-PRESS(J)	CMPT	109
	IF (MAT(J) .LE. 0 .OR. J .EQ. N) GO TO 648	CMPT	110
	MATL = MAT(J)	CMPT	111
	DENS(J) = CELMAS(J) / (X(J+1)-X(J))	CMPT	112
C	*****	CMPT	113
C	CONSTITUTIVE RELATIONS	CMPT	114
C	*****	CMPT	115
C	MODELS FOR PRESSURE	CMPT	116
C	*****	CMPT	117
C	POROUS MATERIAL: THE -SSONED- MODEL	CMPT	118
	IF (NPOR(MATL) .EQ. 0) GO TO 260	CMPT	119
	CALL SSONED(2,J,NSTEP,MATL,DENS(J),DOLD,PRESS(J),COM(L),	CMPT	120
	@ TENSIL(J),COM(L+1),SPEED2)	CMPT	121
	GO TO 450	CMPT	122
C	*****	CMPT	123
C	EXPLOSIVE	CMPT	124

260 IF (NPR(MATL) .EQ. 0) GO TO 280	CMPT 125
CALL EXPLODE(2,J,NSTEP,MATL,DENS(J),PRESS(J),SPEED2)	CMPT 126
GO TO 600	CMPT 127
C *****	CMPT 128
C MIE-GRUENEISEN MODEL FOR A SOLID	CMPT 129
280 IF (NSOL(MATL) .EQ. 0) GO TO 450	CMPT 130
CALL EQST(2,J,NSTEP,MATL,PRESS(J),DENS(J),SPEED2)	CMPT 131
C *****	CMPT 132
C DEVIATOR STRESS MODELS	CMPT 133
C *****	CMPT 134
C ELASTIC, PLASTIC, WITH COULOMB-TRESCA YIELD STRENGTH	CMPT 135
450 IF (Y(J) .EQ. 0. .AND. TANPHI(MATL) .EQ. 0.) GO TO 590	CMPT 136
IF (TANPHI(MATL) .EQ. 0.) GO TO 500	CMPT 137
PLIMIT = -Y(J)*SQRTNP(MATL)/(SQRTNP(MATL)**2-1.)	CMPT 138
IF (PRESS(J) .LE. PLIMIT) GO TO 590	CMPT 139
DEPS = 2.*(DENS(J)-DOLD)/(DENS(J)+DOLD)	CMPT 140
SIGDEV = SIGDEV+1.333*SHMOD(MATL)*DEPS	CMPT 141
IF (SIGDEV .GT. 0.) SIGDEV = MIN(SIGDEV,(Y(J)*SQRTNP(MATL) +	CMPT 142
@ PRESS(J)*(SQRTNP(MATL)**2-1.))/(1.+0.5*SQRTNP(MATL)**2))	CMPT 143
IF (SIGDEV .LT. 0.) SIGDEV = MAX(SIGDEV,-(Y(J)*SQRTNP(MATL) +	CMPT 144
@ PRESS(J)*(SQRTNP(MATL)**2-1.))/(0.5+SQRTNP(MATL)**2))	CMPT 145
GO TO 600	CMPT 146
C DEVIATOR MODEL: ELASTIC, PLASTIC WITH CONSTANT YIELD STRENGTH	CMPT 147
500 DEPS = 2.*(DENS(J)-DOLD)/(DENS(J)+DOLD)	CMPT 148
SIGDEV = SIGDEV+1.333*SHMOD(MATL)*DEPS	CMPT 149
SIGDEV = SIGN(MIN(ABS(SIGDEV),0.666667*Y(J)),SIGDEV)	CMPT 150
GO TO 600	CMPT 151
590 SIGDEV = 0.	CMPT 152
600 SIG1(J) = PRESS(J)+SIGDEV	CMPT 153
SSPEED(J) = SQRT(SPEED2+1.333*SHMOD(MATL)/DENS(J))	CMPT 154
C *****	CMPT 155
C ARTIFICIAL VISCOUS STRESS	CMPT 156
DELTAV = VEL(J+1)-VEL(J)	CMPT 157
IF (DELTAV .GT. 0.) GO TO 635	CMPT 158
DSTRS = RHO(MATL)*DELTAV*(-DAMPL(MATL)*SQRT(ABS(PRESS(J)/DENS(J))	CMPT 159
@ + DAMPQ(MATL)*DELTAV)	CMPT 160
C *****	CMPT 161
C SOUND SPEED	CMPT 162
635 SPEED = SSPEED(J)	CMPT 163
IF (ABS(DENS(J)-DOLD) .LT. 1.E-4) GO TO 640	CMPT 164
EFFMOD = (SIG1(J)-SOLD+2.*DSTRS)/(DENS(J)-DOLD)	CMPT 165
IF (EFFMOD .GT. SPEED**2) SPEED = SQRT(EFFMOD)	CMPT 166
C *****	CMPT 167
C COMPUTE NATURAL TIME STEP FOR EACH CELL	CMPT 168
640 DTNAT = (X(J+1)-X(J)+DT*(VEL(J+1)-VEL(J)))/(SPEED+MAX(VEL(J),0.D0)	CMPT 169
@ - MIN(VEL(J+1),0.D0))	CMPT 170
IF (DTNAT .GT. DTMIN) GO TO 645	CMPT 171
DTMIN = DTNAT	CMPT 172
JTS = J	CMPT 173
DXJTS = X(J+1)-X(J)	CMPT 174
SPJTS = SPEED	CMPT 175
645 RMECH(J) = SIG1(J)+DSTRS	CMPT 176
648 IF (NHIST .GT. 0 .AND. MOD(NSTEP,NHIST) .EQ. 0)	CMPT 177
@ WRITE(6,1645) J, MAT(J), L, X(J), VEL(J), RMECH(J), SIG1(J),	CMPT 178
@ DSTRS, PRESS(J), DENS(J), SSPEED(J), SIGDEV,Y(J)	CMPT 179
1645 FORMAT(I4,I4,I6,1P2E12.5,4E10.3,0PF10.6,1P3E10.3)	CMPT 180
650 CONTINUE	CMPT 181
C *****	CMPT 182
C SET TIME STEP FOR NEXT CYCLE	CMPT 183
DTOLD = DT	CMPT 184
DT = MIN(SAFETY*DTMIN,TEND-TIME,MAX(1.2*DT,0.035*SAFETY*DTMIN))	CMPT 185
IF (NHIST .GT. 0 .AND. MOD(NSTEP,NHIST) .EQ. 0)	CMPT 186

@	WRITE (6,1655) NSTEP, DT, DTMIN, TIME, JTS, DXJTS, SPJTS	CMPT 187
1655	FORMAT (' CMP TIME, NSTEP=',I4,' DT,DTMIN,TIME=',1P3E12.5,' JTS=',	CMPT 188
@	I4,' DX,SPEED AT JTS=',2E10.3)	CMPT 189
C	*****	CMPT 190
C	PRINT SNAPSHOT EDIT OF VALUES AT CURRENT TIME	CMPT 191
	IF (NHIST .LE. 0) GO TO 701	CMPT 192
	NPRNT = NPRNT + 1	CMPT 193
	IF (NPRNT .LT. NHIST) GO TO 701	CMPT 194
	NPRNT = 0	CMPT 195
	CALL HISTRY	CMPT 196
C	WRITE HISTORICAL DATA TO TAPE 7 FOR PLOTTING	CMPT 197
701	IF (KNT .GE. 1 .OR. NSTEP .EQ. 1) CALL RECORD	CMPT 198
C	STORE MAXIMUM VALUES FOR CURRENT CYCLE	CMPT 199
	CALL UPDATE	CMPT 200
	IF (TIME .LT. TEND .AND. NSTEP .LT. NSTOP) GO TO 100	CMPT 201
	END FILE 7	CMPT 202
C	*****	CMPT 203
C	PRINT MAX VALUES OF STRESS, ACC, VEL, POSITION, STRAIN	CMPT 204
	CALL VALMAX	CMPT 205
	END FILE 7	CMPT 206
	REWIND 7	CMPT 207
	IF (KNT .LE. 0) RETURN	CMPT 208
	IF (NOCOMP .EQ. 998) RETURN	CMPT 209
C	WRITE PLOTTING SCALES	CMPT 210
C	*****	CMPT 211
	CALL PLOTSC	CMPT 212
	RETURN	CMPT 213
	END	CMPT 214

	SUBROUTINE EOSTAB (NCALL, IN, M, DN, ENGN, PRESN)	ETAB	1
C	TABULAR ISENTROPE TO BE USED FOR EXPLOSIVE GASES	ETAB	2
	IMPLICIT REAL*8 (A-H, O-Z)	ETAB	3
C		ETAB	4
C	CURRENT REVISION - AUGUST, 1984	ETAB	5
C		ETAB	6
	DIMENSION D(30,6), ENG(30,6), PRES(30,6), EX(30,6), MAXN(6),	ETAB	7
@	PRINV(30), TYPE(6), VORD(30)	ETAB	8
	CHARACTER*8 LABEL, TYPE	ETAB	9
C		ETAB	10
	IF (NCALL .GT. 0) GO TO 200	ETAB	11
C		ETAB	12
C	INITIALIZE AND READ DATA ***** NCALL = 0 *****	ETAB	13
C		ETAB	14
	READ (IN,*) IMAX, LABEL, TYPE(M)	ETAB	15
	MAXN(M) = IMAX	ETAB	16
	IF (LABEL .EQ. 'VOLUME') GO TO 10	ETAB	17
	LABEL = 'DENSITY'	ETAB	18
10	IF (TYPE(M) .EQ. 'LOG') GO TO 20	ETAB	19
	TYPE(M) = 'LINEAR'	ETAB	20
20	WRITE (6,9010) IMAX, LABEL, TYPE(M), IN	ETAB	21
	READ (IN,*) (VORD(I), PRES(I,M), I=1, IMAX)	ETAB	22
	WRITE (6,9030) (VORD(I), PRES(I,M), I=1, IMAX)	ETAB	23
	IF (LABEL .NE. 'VOLUME') GO TO 31	ETAB	24
C	TRANSFORMATION OF VOLUME TO DENSITY	ETAB	25
	DO 30 I=1, IMAX	ETAB	26
30	VORD(I)=1./VORD(I)	ETAB	27
31	IF (VORD(1) .GT. VORD(2)) GO TO 38	ETAB	28
C	INVERT ORDER OF DENSITY VALUES TO GIVE LARGEST VALUES FIRST	ETAB	29
	DO 32 I = 1, IMAX	ETAB	30
32	PRINV(I) = PRES(I,M)	ETAB	31
	DO 35 I = 1, IMAX	ETAB	32
	IS = IMAX+1-I	ETAB	33
	PRES(I,M) = PRINV(IS)	ETAB	34
35	D(I,M) = VORD(IS)	ETAB	35
	GO TO 45	ETAB	36
C	PLACE DENSITY AND PRESSURE IN MAIN ARRAYS	ETAB	37
38	DO 40 I = 1, IMAX	ETAB	38
40	D(I,M) = VORD(I)	ETAB	39
45	IM1=IMAX-1	ETAB	40
	WRITE (6,1045) (PRES(I,M), D(I,M), I=1, IMAX)	ETAB	41
1045	FORMAT (' 45 EOSTAB P,D=', 1P6E10.3/(10X, 6E10.3))	ETAB	42
C	COMPUTE DP/DRHO OR POLYTROPIC EXPONENT	ETAB	43
	DO 50 I=1, IM1	ETAB	44
50	EX(I,M)=(PRES(I+1,M)-PRES(I,M))/(D(I+1,M)-D(I,M))	ETAB	45
	IF (TYPE(M) .NE. 'LOG') GO TO 80	ETAB	46
	DO 65 I=1, IM1	ETAB	47
	IF (PRES(I,M) .LE. 0. .OR. PRES(I+1,M) .LE. 0.) GO TO 65	ETAB	48
	EX(I,M)=LOG(PRES(I+1,M)/PRES(I,M))/LOG(D(I+1,M)/D(I,M))	ETAB	49
65	CONTINUE	ETAB	50
80	CONTINUE	ETAB	51
	WRITE (6,1080) (EX(I,M), I=1, IMAX)	ETAB	52
1080	FORMAT (' 80 EOSTAB EX=', 1P6E10.3/(10X, 6E10.3))	ETAB	53
	RETURN	ETAB	54
C	*****	ETAB	55
C	-----CALCULATE PRESSURE *****	ETAB	56
C	*****	ETAB	57
C	IT - ARRAY SUBSCRIPT OF LEFT LIMIT OF INTERVAL BEING CHECKED	ETAB	58
200	IT = MAXN(M) - 1	ETAB	59
C	CHECK IF INPUT DENSITY IS LESS THAN ANY VALUE IN INTERVAL	ETAB	60
C	CONTAINING LOWEST DENSITY	ETAB	61
	IF (DN .LT. D(IT,M)) GO TO 275	ETAB	62

IT=1		
C CHECK IF INPUT DENSITY IS GREATER THAN ANY VALUE IN INTERVAL	ETAB	63
C CONTAINING HIGHEST DENSITY	ETAB	64
IF (DN .GT. D(IT,M)) GO TO 275	ETAB	65
IM2 = MAXN(M) - 1	ETAB	66
DO 240 I=1,IM2	ETAB	67
IT=MAXN(M)-I+1	ETAB	68
IF (DN .LT. D(IT,M)) GO TO 275	ETAB	69
240 CONTINUE	ETAB	70
275 IF (TYPE(M) .EQ. 'LOG') GO TO 290	ETAB	71
280 PRESN = PRES(IT,M) + (DN-D(IT,M)) * EX(IT,M)	ETAB	72
GO TO 300	ETAB	73
290 IF (PRES(IT,M) .LE. 0. .OR. PRES(IT+1,M) .LE. 0.) GO TO 280	ETAB	74
PRESN = PRES(IT,M) * (DN/D(IT,M)) ** EX(IT,M)	ETAB	75
300 CONTINUE	ETAB	76
RETURN	ETAB	77
C	ETAB	78
9010 FORMAT (' IMAX=',I3,' LABEL=',A8,' TYPE=',A8,2X,4H,IN=,I1,	ETAB	79
@ 7H -EOST-)	ETAB	80
9030 FORMAT (' V OR D, P=',1P6E10.3/(11X,6E10.3))	ETAB	81
END	ETAB	82
	ETAB	83

	SUBROUTINE EQST (LS, J, N, M, P, D, SPEED2)	EQST	1
C	PRESSURE FROM $P = C \cdot \mu + D \cdot \mu^2 + S \cdot \mu^3$	EQST	2
	IMPLICIT REAL*8 (A-H, O-Z)	EQST	3
C		EQST	4
	COMMON /CALIB/ DENSCAL, DISTCAL, PRESCAL, VELCAL	EQST	5
C		EQST	6
	DIMENSION EQSTC(6), EQSTD(6), EQSTS(6), RHO(6)	EQST	7
	IF (LS .GT. 0) GO TO 100	EQST	8
C	INITIALIZATION PORTION	EQST	9
	READ (5, *) EQSTC(M), EQSTD(M), EQSTS(M)	EQST	10
	WRITE (6, 1002) M, EQSTC(M), EQSTD(M), EQSTS(M)	EQST	11
1002	FORMAT (' EOS DATA FOR M=', I2, ': C, D, S=', 1P8E10.3)	EQST	12
	RHO(M) = D	EQST	13
	EQSTC(M) = EQSTC(M) * PRESCAL	EQST	14
	EQSTD(M) = EQSTD(M) * PRESCAL	EQST	15
	EQSTS(M) = EQSTS(M) * PRESCAL	EQST	16
	SPEED2 = EQSTC(M) / RHO(M)	EQST	17
	WRITE (6, 1093) RHO(M), EQSTC(M), EQSTD(M), EQSTS(M), SPEED2	EQST	18
1093	FORMAT (' EOS CALIB. DATA: RHO, C, D, S=', 1P4E10.3, ' SP2=', E10.3)	EQST	19
	RETURN	EQST	20
C	COMPUTE PRESSURE	EQST	21
100	EMU = D / RHO(M) - 1.	EQST	22
	P = EMU * (EQSTC(M) + EMU * (EQSTD(M) + EMU * EQSTS(M)))	EQST	23
	SPEED2 = (EQSTC(M) + 2. * EMU * EQSTD(M) + 3. * EMU ** 2 * EQSTS(M)) / RHO(M)	EQST	24
	RETURN	EQST	25
	END	EQST	26

SUBROUTINE EXPLODE (NCALL, J, N, M, D, P, SPEED2)	EXPL	1
IMPLICIT REAL*8 (A-H,O-Z)	EXPL	2
C	EXPL	3
C THIS SUBROUTINE FOR EXPLOSIONS HAS TWO FUNCTIONS AND	EXPL	4
C IS DIVIDED INTO TWO PARTS:	EXPL	5
C 1. INITIALIZE THE MATERIAL VARIABLES AT THE TIME OF READING	EXPL	6
C MATERIAL PROPERTIES.	EXPL	7
C 2. COMPUTE PRESSURE IN THE EXPLOSIVE PRODUCTS DURING THE	EXPL	8
C CALCULATION.	EXPL	9
C SPECIAL FEATURES:	EXPL	10
C * TABULAR EQUATION OF STATE BY LABELLING -QEXPL- WITH 'TAB'	EXPL	11
C IN THE TITLE OF THE INPUT LINE	EXPL	12
C	EXPL	13
C COMMON /CALIB/ DENSICAL, DISTCAL, PRESCAL, VELCAL	EXPL	14
C	EXPL	15
C DIMENSION RHO(3),MTRANS(6),PCJ(3),QEXPL(3),TAB(3),VCJ(3)	EXPL	16
C DIMENSION EQSTG(3)	EXPL	17
C CHARACTER*3 LABEL	EXPL	18
C	EXPL	19
C DATA MAT/0/	EXPL	20
C IN=5	EXPL	21
C IF (NCALL .GT. 1) GO TO 300	EXPL	22
C NCALL = 1 TO INIT MAT PROP, 2 TO CALC PRESS.	EXPL	23
C *****	EXPL	24
C INITIALIZE MATERIAL VARIABLES	EXPL	25
C *****	EXPL	26
C MAT = MAT+1	EXPL	27
C MTRANS(M) = MAT	EXPL	28
C READ (IN,*) LABEL,QEXPL(MAT),EQSTG(MAT)	EXPL	29
C WRITE (6,9010) LABEL,QEXPL(MAT),EQSTG(MAT)	EXPL	30
C RHO(MAT) = D	EXPL	31
C TAB(MAT) = 0.	EXPL	32
C SPEED2 = 2.*QEXPL(MAT)*EQSTG(MAT)*(EQSTG(MAT)+2.)	EXPL	33
C VCJ(MAT)=(EQSTG(MAT)+1.)/((EQSTG(MAT)+2.)*RHO(MAT))	EXPL	34
C PCJ(MAT)=2.*RHO(MAT)*QEXPL(MAT)*EQSTG(MAT)	EXPL	35
C WRITE (6,9020) QEXPL(MAT),PCJ(MAT),VCJ(MAT),EQSTG(MAT),SPEED2	EXPL	36
C IF (LABEL .NE. 'TAB') RETURN	EXPL	37
C	EXPL	38
C*****PREPARE FOR TABULAR EQUATION OF STATE FOR PRODUCTS	EXPL	39
C CALL EOSTAB (0,IN,MAT,RHO(MAT),E)	EXPL	40
C TAB(MAT) = 1.	EXPL	41
C RETURN	EXPL	42
C	EXPL	43
C *****	EXPL	44
C COMPUTE DETONATION PROCESS	EXPL	45
C *****	EXPL	46
C	EXPL	47
C 300 MAT = MTRANS(M)	EXPL	48
C IF (TAB(MAT) .EQ. 1.) GO TO 350	EXPL	49
C P = PCJ(MAT)*(VCJ(MAT)*D)**(EQSTG(MAT)+1.)	EXPL	50
C SPEED2 = (EQSTG(MAT)+1.)*P/D	EXPL	51
C RETURN	EXPL	52
C	EXPL	53
C*****TABULAR EQUATION OF STATE	EXPL	54
C 350 CALL EOSTAB(1,5,MAT,D,E,P)	EXPL	55
C SPEED2 = 3.*P/D	EXPL	56
C RETURN	EXPL	57
C	EXPL	58
C 9010 FORMAT(1X,A3,2X,1P2E10.3)	EXPL	59
C 9020 FORMAT(3X,'CONST.VOL.EXPLOSION WITH ENERGY='1PE10.3,	EXPL	60
C @ ' ERG/G, PCJ =' E10.3,' VCJ ='E12.5,' GRUN =' ,E12.5,' SPEED2=' ,	EXPL	61
C @ 1PE10.3)	EXPL	62

END

EXPL 63

	SUBROUTINE GRAFONED (TIME, SS, ST1, ST2, NST, NNODE, IDAT, BETA)	GRAF	1
	CHARACTER*10 DISCPT(10), IDAT	GRAF	2
	CHARACTER*80 LABEL(3)	GRAF	3
	DIMENSION TIME(1), SS(1), ST1(1), ST2(1), IAR(7)	GRAF	4
	REAL*8 KK, BETA(12)	GRAF	5
C		GRAF	6
C	GRAPHS DATA FROM ONED	GRAF	7
C	STRESS 1 VS TIME	GRAF	8
C	STRESS 2 VS TIME	GRAF	9
C	STRESS 1 VS STRAIN	GRAF	10
C		GRAF	11
C	INITIALIZE	GRAF	12
C		GRAF	13
	IAR(1)=-1	GRAF	14
	IAR(2)=1	GRAF	15
	IAR(3)=0	GRAF	16
	IAR(4)=0	GRAF	17
	IAR(5)=0	GRAF	18
	IAR(6)=0	GRAF	19
	IAR(7)=5	GRAF	20
C		GRAF	21
	XMIN=0.	GRAF	22
	XMAX=0.	GRAF	23
	YMIN=0.	GRAF	24
	YMAX=0.	GRAF	25
C		GRAF	26
	LABEL(1) = IDAT	GRAF	27
	ENCODE (42,1000,LABEL(1)(11:52)) BETA(1),BETA(2),BETA(3),	GRAF	28
@	BETA(4),BETA(5),BETA(6),BETA(7)	GRAF	29
1000	FORMAT (7A6)	GRAF	30
C		GRAF	31
	LABEL(2) = 'TIME -- *SEC\$'	GRAF	32
	LABEL(3) = 'PRINCIPAL STRESS 1 (MPa), NODE'	GRAF	33
	ENCODE (3,1010,LABEL(3)(32:34)) NNODE	GRAF	34
1010	FORMAT (I3)	GRAF	35
	CALL GRAPH4 (TIME,ST1,NST,1,XMAX,XMIN,YMAX,YMIN,LABEL,IAR)	GRAF	36
C		GRAF	37
	LABEL(3) = 'PRINCIPAL STRESS 2 (MPa), NODE'	GRAF	38
	ENCODE (3,1010,LABEL(3)(32:34)) NNODE	GRAF	39
	CALL GRAPH4 (TIME,ST2,NST,1,XMAX,XMIN,YMAX,YMIN,LABEL,IAR)	GRAF	40
C		GRAF	41
	LABEL(2) = 'STRAIN (*PERCENT\$)'	GRAF	42
	LABEL(3) = 'PRINCIPAL STRESS 1 (MPa), NODE'	GRAF	43
	ENCODE (3,1010,LABEL(3)(32:34)) NNODE	GRAF	44
	CALL GRAPH4 (SS,ST1,NST,1,XMAX,XMIN,YMAX,YMIN,LABEL,IAR)	GRAF	45
	RETURN	GRAF	46
	END	GRAF	47

	SUBROUTINE HISTRY	HIST	1
C	THE HISTRY SUBROUTINE CONTAINS A SERIES OF SEPARATE SUBROUTINES	HIST	2
C	CALLED FOR PRINTING AND STORAGE OPERATIONS.	HIST	3
C	- - ACTIONS OF THE SUBROUTINES - -	HIST	4
C	"ZERO" INITIALIZES VALUES OF THE ARRAYS	HIST	5
C	"HISTRY" PRINTS THE GROUND MOTIONS AT SELECTED TIMES	HIST	6
C	"RECORD" STORES ON TAPE 7 THE VARIABLES AT SELECTED NODES	HIST	7
C	"SCRIBE" PRINTS THE HISTORIES OF THE VARIABLES AT SELECTED NODES	HIST	8
C	"UPDATE" UPDATES THE MAX VALUE TABLE	HIST	9
C	"VALMAX" PRINTS THE MAXIMUM VALUE TABLE	HIST	10
C	"FORCI" INTERPOLATES THE PRESSURE-TIME INPUT VALUES TO	HIST	11
C	DETERMINE THE SURFACE STRESS AT EACH TIME INCREMENT	HIST	12
C	"PLOTSC" COMPUTES THE SCALE FACTORS FOR THE PLOTS	HIST	13
C		HIST	14
	INCLUDE '[SEAMAN.ONED]ONEDCOM.FOR'	HIST	15
C		HIST	16
	DIMENSION AA(20), RIMP(20), RR(20), RROLD(20), SS(20),	HIST	17
@	ST1(20), ST2(20), VV(20), XX(20)	HIST	18
	DIMENSION XSAVE(20), DSAVE(20), XZERO(900)	HIST	19
	CHARACTER*10 IDAT	HIST	20
	DATA ITIME/0/	HIST	21
C	*****	HIST	22
	CALL DATE(IDAT)	HIST	23
C	ENTRY HISTRY	HIST	24
C	- HISTRY - PRINTS A SNAPSHOT OF THE STATUS OF STRESS AND OTHER VALUES	HIST	25
C	AT THE REQUESTED NODES AT PERIODIC CALLS FROM -CMPUTE-.	HIST	26
C	THE CALL TO HISTRY IS AT CYCLES THAT ARE A MULTIPLE OF	HIST	27
C	NHIST.	HIST	28
	LCOUNT = LCOUNT+1	HIST	29
C	IF (LCOUNT-LPRNT) 210,205,205	HIST	30
205	WRITE (6,1018) BETA	HIST	31
	WRITE (6,1019) NVELCAL	HIST	32
	LCOUNT = 0	HIST	33
	RPSI = FORCE/PRESCAL	HIST	34
210	WRITE (6,1021) TIME,RPSI	HIST	35
	DO 215 J=1,NNODE	HIST	36
	KJ = NOUT(J)	HIST	37
	QA = RMECH(KJ)*ALPHA(KJ)/PRESCAL	HIST	38
	RA = SIG1(KJ)*ALPHA(KJ)/PRESCAL	HIST	39
	DAMPF = QA-RA	HIST	40
	MATL = MAT(KJ)	HIST	41
	ST = (DENS(KJ)/RHO(MATL)-1.)*100.	HIST	42
	VELOC = VEL(KJ)*AVELCAL	HIST	43
	AG = ACC(KJ)/G	HIST	44
	XIN = X(KJ)/2.54	HIST	45
215	WRITE (6,1023) KJ,QA,RA,DAMPF,ST,AG,VELOC,XIN	HIST	46
	RETURN	HIST	47
C	*****	HIST	48
	ENTRY RECORD	HIST	49
C	- RECORD - SAVES CURRENT VALUES AT THE REQUESTED NODES FOR LATER	HIST	50
C	PRINTING BY -SCRIBE-. RECORD IS CALLED AT EACH CYCLE	HIST	51
C	BY -CMPUTE-.	HIST	52
C	GLOSSARY OF VALUES STORED ON TAPE 7 FOR HISTORIES	HIST	53
C		HIST	54
C	RR = RMECH(J) MECHANICAL STRESS	HIST	55
C	AA = ACC(J)/G ACCELERATION IN G'S	HIST	56
C	VV = VEL(J) PARTICLE VELOCITY	HIST	57
C	XX = POSITION	HIST	58
C	RIMP = IMPULSE	HIST	59
C	SS = REACTION STRAIN	HIST	60
C	ST1 = THERMODYNAMIC STRESS IN FIRST (AXIAL) DIRECTION.	HIST	61
C	ST2 = THERMODYNAMIC STRESS IN SECOND (LATERAL) DIRECTION.	HIST	62

C	SAVDMX(1,I)...(5,I) ARE RESPECTIVELY THE MAXIMUM VALUES OF	HIST	63
C	STRESS, ACCELERATION, PARTICLE VELOCITY, DISPLACEMENT AND	HIST	64
C	STRAIN AT I.	HIST	65
C	TMAX(1,I)...TMAX(5,I) ARE THE TIMES CORRESPONDING TO THE MAX'S.	HIST	66
C	SK(1)...SK(7) ARE THE SCALE FACTORS FOR PLOTTING THE ABOVE VS	HIST	67
C	TIME.	HIST	68
	TOPSTRS = FORCE*1.E-7	HIST	69
	IF (NSTEP .GT. 1) GO TO 220	HIST	70
C	-DO 218-	HIST	71
	DO 218 J = 1,NNODE	HIST	72
	KJ = NOUT(J)	HIST	73
	DSAVE(J) = DENS(KJ)	HIST	74
	XSAVE(J) = X(KJ)	HIST	75
218	CONTINUE	HIST	76
	DO 219 J = 1,N	HIST	77
219	XZERO(J) = X(J)	HIST	78
C	PREPARE FIRST RECORD ON TAPE 7	HIST	79
C	'NNODE' IS THE NUMBER OF OUTPUT NODES, 'NOUT' GIVES THE CELL NO.S OF	HIST	80
C	THE OUTPUT NODES, 'BETA' IS THE PROBLEM DESCRIPTION, 'NSTEP' IS THE	HIST	81
C	NUMBER OF CYCLES, AND 'N' IS THE NUMBER OF CELLS.	HIST	82
	NEND = 10000	HIST	83
	WRITE (7) NNODE, NOUT, BETA, NEND, N	HIST	84
220	CONTINUE	HIST	85
C	-DO 280-	HIST	86
	DO 280 J=1,NNODE	HIST	87
	KJ = NOUT(J)	HIST	88
C	CONVERT STRESS FROM DYN/CM^2 TO MPA	HIST	89
	RR(J) = RMECH(KJ)*ALPHA(KJ)*1.E-7	HIST	90
	ST1(J) = SIG1(KJ)*1.E-7	HIST	91
	ST2(J) = (1.5*PRESS(KJ)-0.5*SIG1(KJ))*1.E-7	HIST	92
	RIMP(J) = RIMP(J) + (RR(J) + RROLD(J))/2.*DT*1.E7	HIST	93
	IF (J .EQ. 1) WRITE (6,1224) J,KJ,RIMP(J),RR(J),RROLD(J),DT	HIST	94
1224	FORMAT (' 224 HIST J,KJ=',2I4,' RIMP,RR=',1P2E10.3,' RROLD,DT=',	HIST	95
1	2E10.3)	HIST	96
	RROLD(J) = RR(J)	HIST	97
	MATL = MAT(KJ)	HIST	98
	SS(J) = (DENS(KJ)/RHO(MATL)-1.)*100.	HIST	99
	AA(J) = ACC(KJ)/G	HIST	100
	VV(J) = VEL(KJ)*AVELCAL	HIST	101
280	XX(J) = X(KJ)/DISTCAL	HIST	102
	WRITE (7) TIME, TOPSTRS, (RR(J), AA(J), VV(J), XX(J), SS(J), RIMP(J),	HIST	103
@	ST1(J), ST2(J), J=1, NNODE)	HIST	104
290	RETURN	HIST	105
C	*****	HIST	106
	ENTRY SCRIBE	HIST	107
C	- SCRIBE - IS CALLED AT THE END OF A CALCULATION TO PRINT THE	HIST	108
C	HISTORICAL DATA STORED ON TAPE 7 BY -RECORD-. THE	HIST	109
C	CALL IS FROM -ONED-.	HIST	110
	DO 305 I = 1,NNODE	HIST	111
	REWIND 7	HIST	112
	READ (7) NSPACE	HIST	113
	IF (I .EQ. 1) WRITE (6,1299) BETA, NOUT(I), NVELCAL	HIST	114
	IF (I .GT. 1) WRITE (6,1300) BETA, NOUT(I), NVELCAL	HIST	115
	DO 300 NS = 1, NSTEP	HIST	116
	READ (7) TIME, TOPSTRS, (RR(J), AA(J), VV(J), XX(J), SS(J), RIMP(J),	HIST	117
@	ST1(J), ST2(J), J=1, NNODE)	HIST	118
	IF (I .EQ. 1) WRITE (6,1301) NS, TIME, TOPSTRS, RR(I), AA(I),	HIST	119
@	VV(I), XX(I), SS(I), RIMP(I), ST1(I), ST2(I)	HIST	120
	IF (I .GT. 1) WRITE (6,1301) NS, TIME, RR(I), AA(I), VV(I), XX(I),	HIST	121
@	SS(I), RIMP(I), ST1(I), ST2(I)	HIST	122
	IF (MOD(NS,55) .EQ. 0 .AND. I .EQ. 1) WRITE (6,1299) BETA,	HIST	123
@	NOUT(I), NVELCAL	HIST	124

IF (MOD(NS,55) .EQ. 0 .AND. I .GT. 1) WRITE (6,1300) BETA,	HIST 125
@ NOUT(I), NVELCAL	HIST 126
300 CONTINUE	HIST 127
305 CONTINUE	HIST 128
1299 FORMAT (1H1,10X,12A6/	HIST 129
@ 24H HISTORIES AT NODE NO. ,I3/4X,2HNS,8X,4HTIME,	HIST 130
@ 2X,10HTOP STRESS,7X,5HRMECH,9X,3HACC,4X,8HVELOCITY,4X,	HIST 131
@ 8HPOSITION,6X,6HSTRAIN,5X,7HIMPULSE,5X,7HSIGMA-1,5X,7HSIGMA-2/	HIST 132
@ 15X,3HSEC,9X,3HMPA,9X,3HMPA,9X,3HG'S,6X,A2,4H/SEC,10X,2HIN,	HIST 133
@ 9X,3H(%),2X,10HDYN-S/CM^2,9X,3HMPA,9X,3HMPA)	HIST 134
1300 FORMAT (1H1,10X,12A6/	HIST 135
@ 24H HISTORIES AT NODE NO. ,I3/4X,2HNS,8X,4HTIME,	HIST 136
@ 7X,5HRMECH,9X,3HACC,4X,8HVELOCITY,4X,8HPOSITION,6X,6HSTRAIN,	HIST 137
@ 5X,7HIMPULSE,5X,7HSIGMA-1,5X,7HSIGMA-2/	HIST 138
@ 15X,3HSEC,9X,3HMPA,9X,3HG'S,6X,A2,4H/SEC,10X,2HIN,	HIST 139
@ 9X,3H(%),2X,10HDYN-S/CM^2,9X,3HMPA,9X,3HMPA)	HIST 140
1301 FORMAT (I6,1P10E12.3)	HIST 141
RETURN	HIST 142
C *****	HIST 143
ENTRY UPDATE	HIST 144
C - UPDATE - IS CALLED TO UPDATE THE MAXIMUM VALUE TABLE AT EACH CYCLE.	HIST 145
C THESE VALUES ARE STORED AT THE REQUESTED NODES. UPDATE	HIST 146
C IS CALLED BY -CMPUTE-.	HIST 147
IF (MVT) 319,317,319	HIST 148
317 IF (KNT) 500,500,318	HIST 149
318 NSTRT = NOUT(1)	HIST 150
NFINSH = NOUT(1)+1	HIST 151
GO TO 320	HIST 152
319 NSTRT = 1	HIST 153
NFINSH = N	HIST 154
320 DO 365 I = NSTRT,NFINSH	HIST 155
IF (ABS(RMECH(I)) .LT. ABS(SAVDMX(1,I))) GO TO 325	HIST 156
SAVDMX(1,I) = RMECH(I)	HIST 157
TMAX(1,I) = TIME	HIST 158
325 IF (ABS(ACC(I)) .LT. ABS(SAVDMX(2,I))) GO TO 330	HIST 159
SAVDMX(2,I) = ACC(I)	HIST 160
TMAX(2,I) = TIME	HIST 161
330 IF (ABS(VEL(I)) .LT. ABS(SAVDMX(3,I))) GO TO 335	HIST 162
SAVDMX(3,I) = VEL(I)	HIST 163
TMAX(3,I) = TIME	HIST 164
335 IF (ABS(X(I)-XZERO(I)) .LT. ABS(SAVDMX(4,I))) GO TO 340	HIST 165
SAVDMX(4,I) = X(I)-XZERO(I)	HIST 166
TMAX(4,I) = TIME	HIST 167
340 MATL = MAT(I)	HIST 168
IF (ABS(DENS(I)-RHO(MATL)) .LT. ABS(SAVDMX(5,I))) GO TO 365	HIST 169
SAVDMX(5,I) = DENS(I)-RHO(MATL)	HIST 170
TMAX(5,I) = TIME	HIST 171
365 CONTINUE	HIST 172
RETURN	HIST 173
C *****	HIST 174
ENTRY VALMAX	HIST 175
C - VALMAX - PRINTS THE TABLE OF MAXIMUM VALUES OF STRESS,	HIST 176
C ACCELERATION, VELOCITY, DISPLACEMENT, AND STRAIN,	HIST 177
C AND FOR ALL NODES AND CELLS.	HIST 178
C THE CALL IS FROM -CMPUTE-.	HIST 179
IF (MVT .EQ. 0) GO TO 420	HIST 180
LCOUNT = 50	HIST 181
DO 410 I=1,N	HIST 182
C MODIFY THE MAXIMUM VALUES WITH CALIBRATION FACTORS	HIST 183
SAVDMX(1,I) = SAVDMX(1,I)*ALPHA(I)*1.E-7	HIST 184
SAVDMX(2,I) = SAVDMX(2,I)/G	HIST 185
SAVDMX(3,I) = SAVDMX(3,I)*AVELCAL	HIST 186

SAVDMX(4,I) = SAVDMX(4,I)/DISTCAL	HIST 187
MATL = MAT(I)	HIST 188
IF (MATL .EQ. 0 .OR. RHO(MATL) .EQ. 0.) MATL=MAT(I-1)	HIST 189
SAVDMX(5,I) = SAVDMX(5,I)/RHO(MATL)*100.	HIST 190
LCOUNT = LCOUNT+1	HIST 191
IF (LCOUNT-50) 408,406,406	HIST 192
406 LCOUNT = 0	HIST 193
WRITE (6,1014) BETA	HIST 194
WRITE (6,1015) NVELCAL	HIST 195
408 CONTINUE	HIST 196
XFT = XZERO(I)/30.48	HIST 197
WRITE (6,1017),I,XFT, SAVDMX(1,I),TMAX(1,I), SAVDMX(3,I),	HIST 198
@ TMAX(3,I), SAVDMX(4,I), TMAX(4,I), SAVDMX(5,I), TMAX(5,I),	HIST 199
@ SAVDMX(2,I),TMAX(2,I)	HIST 200
WRITE (7) I,CELMAS(I),(SAVDMX(J,I),TMAX(J,I),J=1,5)	HIST 201
410 CONTINUE	HIST 202
WRITE (7) NNODE, NOUT, BETA, NSTEP, N	HIST 203
420 RETURN	HIST 204
C *****	HIST 205
C ENTRY PLOTSC	HIST 206
C - PLOTSC - DETERMINE THE SCALE FACTORS FOR PLOTTING	HIST 207
C THIS ROUTINE HAS NOT BEEN UPDATED OR TESTED.	HIST 208
C	HIST 209
430 K1 = NOUT(1)	HIST 210
DO 440 J=1,6	HIST 211
TEMX = ABS(SAVDMX(J,K1))	HIST 212
IF (J .EQ. 6) TEMX = RRM	HIST 213
IF (TEMX-16.) 431,435,435	HIST 214
431 IF (TEMX-1.6) 432,433,433	HIST 215
432 SKF = 0.1	HIST 216
GO TO 438	HIST 217
433 SKF = 1.0	HIST 218
GO TO 438	HIST 219
435 IF (TEMX-160.) 436,437,437	HIST 220
436 SKF = 10.	HIST 221
GO TO 438	HIST 222
437 IF (TEMX-1600.) 4371,4372,4372	HIST 223
4371 SKF = 100.	HIST 224
GO TO 438	HIST 225
4372 SKF = 1000.	HIST 226
438 ISK = TEMX/4./SKF+1.	HIST 227
SKI = ISK	HIST 228
SK(J) = SKI*SKF	HIST 229
440 WRITE (6,1440),J,SK(J)	HIST 230
SK(7) = SK(5)	HIST 231
WRITE (6,1440),7,SK(7)	HIST 232
RETURN	HIST 233
C *****	HIST 234
C ENTRY ZERO	HIST 235
C - ZERO - SETS THE MAIN ARRAYS TO ZERO DURING INITIALIZATION. -ZERO-	HIST 236
C IS CALLED BY -READIT-.	HIST 237
LPRNT = 55/(NNODE+1)	HIST 238
LCOUNT = LPRNT	HIST 239
DO 455 I=1,N	HIST 240
RMECH(I) = 0.	HIST 241
CELMAS(I) = 0.	HIST 242
ACC(I) = 0.	HIST 243
VEL(I) = 0.	HIST 244
DO 455 J=1,5	HIST 245
SAVDMX(J,I) = 0.	HIST 246
TMAX(J,I) = 0.	HIST 247
	HIST 248

455	CONTINUE	HIST 249
	DO 460 J=1,NNODE	HIST 250
	RROLD(J) = 0.	HIST 251
460	RIMP(J) = 0.	HIST 252
500	RETURN	HIST 253
C	*****	HIST 254
	ENTRY FORCI	HIST 255
C	- FORCI - IS CALLED BY -CMPUTE- TO PROVIDE THE PRESSURE AT THE	HIST 256
C	TOP BOUNDARY AT EACH TIME STEP. -FORCI- INTERPOLATES	HIST 257
C	BETWEEN VALUES OF THE PRESSURE HISTORY.	HIST 258
C		HIST 259
	IF (ITIME .GT. 0) GO TO 45	HIST 260
C	ACQUIRE FIRST TWO POINTS	HIST 261
10	TLAST = THIST(1)	HIST 262
	PLAST = PHIST(1)*PRESCAL	HIST 263
	ITIME = 2	HIST 264
	GO TO 35	HIST 265
C	ACQUIRE THE NEXT POINT	HIST 266
30	IF (ITIME .GE. NFORCE) GO TO 50	HIST 267
	TLAST = TNOW	HIST 268
	PLAST = PNOW	HIST 269
	ITIME = ITIME+1	HIST 270
35	TNOW = THIST(ITIME)	HIST 271
	PNOW = PHIST(ITIME)*PRESCAL	HIST 272
C	TEST FOR VERTICAL SLOPE	HIST 273
	DELTAT = TNOW-TLAST	HIST 274
	IF (DELTAT .LT. 1.E-12) GO TO 30	HIST 275
C	CALCULATE SLOPE	HIST 276
	SLOPE = (PNOW-PLAST)/DELTAT	HIST 277
45	IF (TNOW .LT. TIME .AND. ITIME .LT. NFORCE) GO TO 30	HIST 278
C	COMPUTE FORCE ON TOP SURFACE	HIST 279
50	FORCE = PLAST+SLOPE*(TIME-TLAST)	HIST 280
	RETURN	HIST 281
C		HIST 282
C	FORMATS	HIST 283
1014	FORMAT (1H1,10X,12A6/23X,	HIST 284
	@ 25HSUMMARY OF MAXIMUM VALUES // 2X,4HNODE, 3X,	HIST 285
	@ 5HDEPTH,4X,16HREACTION STRESS,10X,8HVELOCITY,11X,	HIST 286
	@ 12HDISPLACEMENT,11X,6HSTRAIN,12X,12HACCELERATION,10X /	HIST 287
	@ 2X,3HNO.,5X,3HFT.,5(6X,5HVALUE,6X,4HTIME))	HIST 288
1015	FORMAT (19X,5H(MPA),6X,5H(SEC),4X,1H(,A2,3H/S),6X,5H(SEC),6X,	HIST 289
	@ 4H(IN),6X,5H(SEC),6X,5H(%),6X,5H(SEC),6X,5H(G'S),	HIST 290
	@ 6X,5H(SEC))	HIST 291
1017	FORMAT (2X,I3,F10.4,F10.2,2(1PE11.3,0PF10.3),1P5E11.3)	HIST 292
1018	FORMAT (//10X,12A6/	HIST 293
	@ 10X,50HTIME SURFACE NODE APPLIED REACTION DAMPING	HIST 294
	@ 42H STRAIN ACCEL PARTICLE POSITION / 18X,	HIST 295
	@ 41HSTRESS NO. STRESS STRESS STRESS,24X,8HVELOCITY)	HIST 296
1019	FORMAT (20X,3HPSI,13X,3HPSI,6X,3HPSI,7X,3HPSI,4X,7HPERCENT,7X,	HIST 297
	@ 3HG'S,7X,A2,2H/S,6X,2HIN)	HIST 298
1021	FORMAT (2X,1PE12.3,0PF10.2)	HIST 299
1023	FORMAT (27X,I5,3F9.2,F10.5,F12.3,F10.3,F10.3)	HIST 300
1440	FORMAT (//9X,14H SCALE FACTOR,I1,8H EQUALS ,F6.1)	HIST 301
	END	HIST 302

	PROGRAM PLOTIT	PLOT	1
C	PROGRAM TO PLOT HISTORIES FROM -ONED-	PLOT	2
	IMPLICIT REAL*8 (A-H,O-Z)	PLOT	3
	DIMENSION BETA(12), NOUT(20)	PLOT	4
	DIMENSION AA(20), RIMP(20), RR(20), RROLD(20), SS(20),	PLOT	5
@	ST1(20), ST2(20), VV(20), XX(20)	PLOT	6
	CHARACTER*10 IDAT	PLOT	7
	REAL*4 PTIME(800), PSS(800), PST1(800), PST2(800)	PLOT	8
C		PLOT	9
C	INITIALIZE PLOTTING PARAMETERS	PLOT	10
	CALL GINITL(0)	PLOT	11
	WRITE(95,7216)	PLOT	12
7216	FORMAT ('SET VIEWPORT 1 (-.95 .95 -.95 .95)')	PLOT	13
C	CALL DATE FOR THE CALCULATION	PLOT	14
	CALL DATE(IDAT)	PLOT	15
C	READ THE NUMBER OF NODES (NNODE) AND NUMBER OF CYCLES (NSTEP)	PLOT	16
	READ (7) NNODE, NOUT, BETA, NSTEP, N	PLOT	17
	WRITE (6,1001) NNODE, NOUT, BETA, NSTEP, N	PLOT	18
1001	FORMAT (' NNODE=',I3/' NOUT=',20I3/' BETA=',12A6/' NSTEP=',I5/	PLOT	19
@	' NUMBER OF CELLS, N=',I4)	PLOT	20
C	-----	PLOT	21
C	FIND NUMBER OF COMPUTATIONAL STEPS: NSTEP	PLOT	22
	DO 400 I = 1,NSTEP	PLOT	23
	IEND = I-1	PLOT	24
	READ (7,END=405,ERR=405) N1	PLOT	25
400	CONTINUE	PLOT	26
405	NSTEP = IEND-N	PLOT	27
C	-----	PLOT	28
C	BEGIN MAJOR LOOP OVER EACH OUTPUT NODE	PLOT	29
	DO 610 J = 1,NNODE	PLOT	30
	REWIND 7	PLOT	31
	READ (7) NSPACE	PLOT	32
	NST = 0	PLOT	33
C	BEGIN LOOP OVER EACH TIME STEP FOR EACH NODE	PLOT	34
	DO 600 I = 1,NSTEP	PLOT	35
	NST = NST+1	PLOT	36
	READ (7,ERR=605,END=605) TIME, TOPSTRS, (RR(JJ), AA(JJ), VV(JJ),	PLOT	37
@	XX(JJ), SS(JJ), RROLD(JJ), ST1(JJ), ST2(JJ), JJ=1, NNODE)	PLOT	38
	PTIME(NST) = TIME	PLOT	39
	PSS(NST) = SS(J)	PLOT	40
	PST1(NST) = ST1(J)	PLOT	41
	PST2(NST) = ST2(J)	PLOT	42
600	CONTINUE	PLOT	43
	GO TO 608	PLOT	44
605	NSTEP = NST-1	PLOT	45
	WRITE (6,1002) NSTEP	PLOT	46
1002	FORMAT (' NSTEP =',I4)	PLOT	47
608	CONTINUE	PLOT	48
	CALL GRAFONED(PTIME, PSS, PST1, PST2, NST, NOUT(J), IDAT, BETA)	PLOT	49
610	CONTINUE	PLOT	50
	CALL GPLOT(.0, .0, 999)	PLOT	51
	END	PLOT	52

	SUBROUTINE READIT	READ	1
C	"READIT" READS THE INPUT DATA, EXCEPT FOR FORCE HISTORY AND	READ	2
C	SOME MATERIALS DATA, LAYS OUT THE CELLS, AND FILLS THE ARRAYS.	READ	3
C		READ	4
	INCLUDE '[SEAMAN.ONED]ONEDCOM.FOR'	READ	5
C		READ	6
	CHARACTER*10 IDAT	READ	7
	DIMENSION NFIRST(30),NLAST(30),SP(300),SPMAT(6)	READ	8
	DATA G, DENSAL, PRESCAL, VELCAL, DISTCAL/	READ	9
	@ 981.456, 0.0160185, 6.8948E4, 2.54, 2.54/	READ	10
C		READ	11
C	*** CALL -ZERO- TO INITIALIZE THE ARRAYS ****	READ	12
	CALL ZERO	READ	13
	CIN3 = 3456.	READ	14
	CALL DATE(IDAT)	READ	15
	WRITE (6,1000)	READ	16
C		READ	17
C	*** READ THE IDENT NUMBER AND TITLE FOR THE CALCULATION ***	READ	18
	DO 100 K=1,2	READ	19
	READ (5,1008) BETA	READ	20
100	WRITE (6,1008) BETA	READ	21
	READ (5,1001) IDENT, BETA	READ	22
	WRITE (6,1002) BETA	READ	23
	WRITE (6,1003) IDENT	READ	24
	DECODE (10,1008,IDAT) BETA(11),BETA(12)	READ	25
C	***	READ	26
	READ (5,*) NOCOMP, LB, TEND, NSTOP, GACC	READ	27
	WRITE (6,1005)	READ	28
	IF (NOCOMP .LT. 998) WRITE (6,1006) NOCOMP	READ	29
	IF (NOCOMP .EQ. 998) WRITE (6,1060) NOCOMP	READ	30
	IF (NOCOMP .GT. 998) WRITE (6,1007) NOCOMP	READ	31
	IF (LB .NE. 0) WRITE (6,1011) LB	READ	32
	IF (LB .EQ. 0) WRITE (6,1012) LB	READ	33
	WRITE (6,1014) TEND, NSTOP	READ	34
	IF (GACC .NE. 0.) GACC = G	READ	35
	WRITE (6,1015) GACC	READ	36
C	***	READ	37
	READ (5,*) NLAYRS, MISP, NSP, MATTEN, NMTRLS, NFORCE	READ	38
	WRITE (6,1020)	READ	39
	WRITE (6,1021) NLAYRS	READ	40
	IF (MISP .LE. 0) WRITE (6,1022) MISP	READ	41
	IF (MISP .EQ. 1) WRITE (6,1023) MISP	READ	42
	IF (MISP .GE. 2) WRITE (6,1024) MISP, NSP	READ	43
	IF (MATTEN .LE. 0) WRITE (6,1025) MATTEN	READ	44
	IF (MATTEN .GE. 1) WRITE (6,1028) MATTEN	READ	45
	WRITE (6,1029) NMTRLS,NFORCE	READ	46
C	***	READ	47
	READ (5,*) NNODE, KNTROL, NHIST, MVT, NSV, KVEL, TNDPLT	READ	48
	WRITE (6,1031)	READ	49
	IF (KNTROL(1) .GT. 0) GO TO 131	READ	50
	WRITE (6,1032)	READ	51
	GO TO 139	READ	52
131	DO 138 I = 1,7	READ	53
	IF (KNTROL(I) .EQ. 0) GO TO 139	READ	54
	KNT = I	READ	55
	IF (KNTROL(I) .EQ. 1) WRITE (6,1033) KNT,KNTROL(I)	READ	56
	IF (KNTROL(I) .EQ. 2) WRITE (6,1035) KNT,KNTROL(I)	READ	57
	IF (KNTROL(I) .EQ. 3) WRITE (6,1036) KNT,KNTROL(I)	READ	58
	IF (KNTROL(I) .EQ. 4) WRITE (6,1037) KNT,KNTROL(I)	READ	59
	IF (KNTROL(I) .EQ. 5) WRITE (6,1038) KNT,KNTROL(I)	READ	60
	IF (KNTROL(I) .EQ. 6) WRITE (6,1039) KNT,KNTROL(I)	READ	61
	IF (KNTROL(I) .EQ. 7) WRITE (6,1040) KNT,KNTROL(I)	READ	62

138	CONTINUE	READ	63
139	IF (NHIST .LE. 0) WRITE (6,1042) NHIST	READ	64
	IF (NHIST .GE. 1) WRITE (6,1043) NHIST	READ	65
	IF (MVT .EQ. 0) WRITE (6,1045) MVT	READ	66
	IF (MVT .NE. 0) WRITE (6,1044) MVT	READ	67
	WRITE (6,1046) NNODE	READ	68
	IF (TNDPLT .LT. TEND) TNDPLT = TEND	READ	69
	WRITE (6,1153) TNDPLT	READ	70
	NVELCAL = ' M'	READ	71
	AVELCAL = 0.01	READ	72
	IF (KVEL .NE. 0) GO TO 147	READ	73
	NVELCAL = ' IN'	READ	74
	AVELCAL = 1./VELCAL	READ	75
147	WRITE (6,1154) KVEL,NVELCAL	READ	76
C ***		***	READ 77
	READ (5,*) (DEP(L),L=1,NNODE)		READ 78
	IF (NOCOMP .LT. 998) GO TO 150		READ 79
C ***		***	READ 80
148	READ (5,*) SK		READ 81
	WRITE (6,1048) SK		READ 82
	IF (NOCOMP .GE. 999) RETURN		READ 83
C			READ 84
C	INPUT MATERIAL PROPERTIES FOR EACH MATERIAL		READ 85
C			READ 86
150	NSTEP = 0		READ 87
	J = 0		READ 88
	DO 300 M = 1, NMTRLS		READ 89
	DAMPQ(M) = 4.		READ 90
	LS = 0		READ 91
C ***		***	READ 92
	READ (5,*) NAME(M), GAM(M), DAMPL(M), DAMPQ(M), TENS(N), NPOR(M),		READ 93
@	NPR(M), NSOL(M), NVAR(M)		READ 94
	WRITE (6,1158) M, NAME(M), GAM(M), DAMPL(M), DAMPQ(M), TENS(N),		READ 95
@	NPOR(M), NPR(M), NSOL(M), NVAR(M)		READ 96
	TENS(N) = -ABS(TENS(N))		READ 97
	IF (TENS(N) .NE. 0.) NTENS(N) = 1		READ 98
	READ (5,*) SHMOD(M), YIELD(M), TANPHI(M)		READ 99
	WRITE (6,1160) SHMOD(M), YIELD(M), TANPHI(M)		READ 100
	SQRTNP(M) = SQRT(1.+TANPHI(M)**2)+TANPHI(M)		READ 101
	RHO(M) = GAM(M)*DENS(CAL)		READ 102
	TENS(N) = TENS(N)*PRES(CAL)		READ 103
	IF (GAM(M) .EQ. 0.) GO TO 300		READ 104
	IF (NPOR(M) .EQ. 0) GO TO 190		READ 105
	CALL SSONED(LS,J,NSTEP,M,RHO(M),RHO(M),P,EMAX,TENS(N),P(MAX),		READ 106
@	SPMAT(M)		READ 107
	GO TO 290		READ 108
190	IF (NPR(M) .EQ. 0) GO TO 210		READ 109
	CALL EXPLODE(LS,J,NSTEP,M,RHO(M),P,SPMAT(M))		READ 110
	GO TO 290		READ 111
210	IF (NSOL(M) .EQ. 0) GO TO 300		READ 112
	CALL EQST(LS,J,NSTEP,M,P,RHO(M),SPMAT(M))		READ 113
C	COMPUTE SOUND SPEED FROM SOUND-SPEED-SQUARED FROM PRESSURE		READ 114
C	RELATIONS AND SHEAR MODULUS		READ 115
	290 SPMAT(M) = SQRT(SPMAT(M) + 1.333*SHMOD(M)/RHO(M))		READ 116
	300 CONTINUE		READ 117
C	END OF LOOP TO READ IN MATERIAL PROPERTIES		READ 118
	NCEL = 1		READ 119
	TOP = 0.		READ 120
	LAST = 0		READ 121
	LVMAX = 1		READ 122
	X(1) = 0.		READ 123
	GEOSTR = 0.		READ 124

DO 600 NL = 1,NLAYRS	READ 125
C ***	READ 126
READ (5,*) DB, SPL, NSP, M	READ 127
THICK = DB-TOP	READ 128
IF (THICK .GT. 0.) GO TO 310	READ 129
WRITE (6,1312) DB, TOP, THICK	READ 130
1312 FORMAT (' STOP IN READIT, NEGATIVE LAYER THICKNESS. BOTTOM OF '	READ 131
@ '/' CURRENT LAYER, DB =',F12.6,' BOTTOM OF PREVIOUS LAYER =',	READ 132
@ F12.6,/' THICKNESS =',F12.6)	READ 133
STOP 'READIT, NEGATIVE THICKNESS'	READ 134
310 IF (M .GT. 0) GO TO 315	READ 135
WRITE (6,1315) THICK	READ 136
1315 FORMAT (/' GAP IN THE LAYERS = ',F12.6,' FEET',/)	READ 137
X(LAST+1) = DB*12.*DISTCAL	READ 138
GO TO 600	READ 139
315 CONTINUE	READ 140
TOP = DB	READ 141
IF (GAM(M) .EQ. 0.) NSP = 1	READ 142
IF (MISP .EQ. 1) GO TO 330	READ 143
C UNIFORM SIZES FOR CELLS	READ 144
IF (MISP .GT. 0 .AND. NSP .GT. 0) SPL = THICK*12./NSP	READ 145
NSP = THICK*12./SPL+0.5	READ 146
SPL = THICK*12./NSP*DISTCAL	READ 147
SP(1) = SPL	READ 148
GO TO 340	READ 149
C *** FOR MISP = 1: INPUT VARIABLE NODE SPACING	READ 150
330 READ (5,*) (SP(I),I=1,NSP)	READ 151
DO 335 I = 1,NSP	READ 152
335 SP(I) = SP(I)*DISTCAL	READ 153
340 NFIRST(NL) = LAST+1	READ 154
NLAST(NL) = LAST+NSP	READ 155
LAST = NLAST(NL)+1	READ 156
NFST = NFIRST(NL)	READ 157
NLST = NLAST(NL)	READ 158
I = 0	READ 159
LS = 2	READ 160
WRITE (6,1343) BETA	READ 161
1343 FORMAT (/' LISTING OF THE LAYOUT' /10X,12A6)	READ 162
WRITE (6,1344)	READ 163
1344 FORMAT (4X,1HJ,10X,1HX,8X,3HSPL,4X,7HDENSITY,7X,4HSIG1,5X,	READ 164
@ 6HCELMAS,7X,4HTENS,3H M,3X,4HNAME,6H LVAR,3X,6HTRIALS)	READ 165
NLST1 = NLST+1	READ 166
DENSLST = RHO(M)	READ 167
DO 560 J = NFST,NLST1	READ 168
ITRY = 0	READ 169
IF (J .EQ. NLST1) GO TO 550	READ 170
I = I+1	READ 171
IF (MISP .EQ. 1) SPL = SP(I)	READ 172
X(J+1) = X(J)+SPL	READ 173
DENS(J) = DENSLST	READ 174
SSPEED(J) = SPMAT(M)	READ 175
SPEED2 = SPMAT(M)**2	READ 176
Y(J) = YIELD(M)	READ 177
MAT(J) = 0	READ 178
IF (GAM(M) .EQ. 0.) GO TO 540	READ 179
C MODIFY INITIAL DENSITY FOR GEOSTATIC STRESS	READ 180
IF (GACC .EQ. 0.) GO TO 425	READ 181
IF (NPR(M) .NE. 1) DENS(J) = DENS(J) +	READ 182
@ DENS(J)*SPL*GACC/SPEED2	READ 183
400 ITRY = ITRY+1	READ 184
GEOEST = GEOSTR + 0.5*DENS(J)*SPL*GACC	READ 185
C GET STRESS FROM THE EQUATIONS OF STATE FOR EACH MATERIAL	READ 186

IF (NPOR(M) .EQ. 0) GO TO 410	READ 187
PRESS(J) = 0.	READ 188
EMAX = 0.	READ 189
PMAX = 0.	READ 190
CALL SSONED(LS,J,NSTEP,M,DENS(J),DENS(J),PRESS(J),EMAX,TENSN(M),	READ 191
@ PMAX,SPEED2)	READ 192
GO TO 420	READ 193
410 IF (NPR(M) .EQ. 0) GO TO 415	READ 194
PRESS(J) = GEOEST	READ 195
SIG1(J) = PRESS(J)	READ 196
GO TO 425	READ 197
415 IF (NSOL(M) .EQ. 0) GO TO 420	READ 198
CALL EQST(LS,J,NSTEP,M,PRESS(J),DENS(J),SPEED2)	READ 199
420 SIG1(J) = PRESS(J)	READ 200
C WRITE (6,1422) NL,J,ITRY,GEOEST,SIG1(J),DENS(J),SPEED2,DDENS	READ 201
C 1422 FORMAT (' GEOEST READIT, NL,J,ITRY=',3I3,' GEOEST,SIG1=',1P2E17.10	READ 202
C @ ' D=',E15.8,' SP2=',E10.3,' DDENS=',E10.3)	READ 203
IF (ABS(SIG1(J) - GEOEST) .LT. 1.E-6*GEOEST .AND. ITRY .GT. 1)	READ 204
@ GO TO 425	READ 205
TRY = 0.5*ITRY	READ 206
DENSOLD = DENS(J)	READ 207
DENS(J) = DENS(J)+MIN(1.D0,TRY)*(GEOEST-SIG1(J))*	READ 208
@ (DENS(J)-RHO(M))/SIG1(J)	READ 209
DDENS = DENS(J) - DENSOLD	READ 210
IF (ITRY .LT. 20) GO TO 400	READ 211
WRITE (6,1349) NL,J,GEOEST,SIG1(J),RHO(M),DENS(J)	READ 212
STOP 'READIT, 424'	READ 213
425 CELMAS(J) = SPL*DENS(J)	READ 214
RMECH(J) = SIG1(J)	READ 215
ALPHA(J) = 1.	READ 216
MAT(J) = M	READ 217
TENSIL(J) = TENSN(M)	READ 218
DENSLST = DENS(J)	READ 219
GEOSTR = GEOSTR+CELMAS(J)*GACC	READ 220
540 IF (NVAR(M) .EQ. 0) GO TO 550	READ 221
LVAR(J) = LVMAX	READ 222
LVMAX = LVMAX+NVAR(M)	READ 223
550 WRITE (6,1542) J,X(J),SPL,DENS(J),SIG1(J),CELMAS(J),TENSIL(J),	READ 224
@ MAT(J),NAME(M),LVAR(J),ITRY	READ 225
1542 FORMAT (I5,1P2E11.3,0PF11.6,1P3E11.3,I3,2X,A6,I5,I4)	READ 226
C END LOOP OVER EACH J WITHIN A LAYER	READ 227
560 CONTINUE	READ 228
SPL = SP(1)	READ 229
X(NLST+2) = X(NLST+1)	READ 230
MAT(NLST+1) = 0	READ 231
C END LOOP OVER EACH LAYER	READ 232
600 CONTINUE	READ 233
NLAST(NLAYRS) = NLAST(NLAYRS)+1	READ 234
N = LAST	READ 235
CELMAS(N) = CELMAS(N-1)	READ 236
RMECH(N) = RMECH(N-1)+CELMAS(N)*GACC	READ 237
WRITE (6,1603) N,RMECH(N)	READ 238
1603 FORMAT (' LAST NODE N,RMECH(N)=' ,I5,1PE12.5)	READ 239
C	READ 240
WRITE (6,4101) BETA	READ 241
DO 610 NL=1,NLAYRS	READ 242
J = NFIRST(NL)	READ 243
M = MAT(J)	READ 244
JEND = NLAST(NL)+1	READ 245
DB = X(JEND)/12./DISTCAL	READ 246
SPL = (X(J+1)-X(J))/DISTCAL	READ 247
TENS = TENSN(M)/PRESCAL	READ 248

WRITE (6,4102) NL,NFIRST(NL),JEND,DB,GAM(M),DAMPL(M),	READ 249
@ SPL,TENS	READ 250
610 CONTINUE	READ 251
C	READ 252
C SPATIAL ATTENUATION FACTORS	READ 253
C	READ 254
IF (MATTEN .LE. 1) GO TO 720	READ 255
C ***	READ 256
READ (5,*) (ALPHA(I),I=1,N)	READ 257
DO 700 I = 1,N	READ 258
700 CELMAS(I) = CELMAS(I)/ALPHA(I)	READ 259
WRITE (6,4202) (ALPHA(I),I=1,N)	READ 260
C *****	READ 261
C READ IN THE FORCE HISTORY	READ 262
720 IF (NFORCE .LE. 0) GO TO 750	READ 263
WRITE (6,1720) BETA	READ 264
READ (5,*) (THIST(I),PHIST(I),I=1,NFORCE)	READ 265
WRITE (6,1004) (THIST(I),PHIST(I),I=1,NFORCE)	READ 266
C	READ 267
C COMPUTE PROPERTIES AT EACH NODE	READ 268
C	READ 269
750 CONTINUE	READ 270
C	READ 271
C DETERMINE FOR WHICH CELLS OUTPUT IS REQUIRED	READ 272
C	READ 273
WRITE (6,4600)	READ 274
DO 860 L = 1,NNODE	READ 275
DEPTH = DEP(L)*12.*DISTCAL	READ 276
DO 840 J = 2,N	READ 277
IF (DEPTH .GT. X(J)) GO TO 840	READ 278
NOUT(L) = J-1	READ 279
IF (MAT(J-1) .EQ. 0) NOUT(L) = J-2	READ 280
GO TO 850	READ 281
840 CONTINUE	READ 282
850 WRITE (6,4601) DEP(L),NOUT(L)	READ 283
860 CONTINUE	READ 284
C	READ 285
C OUTPUT NODAL PROPERTIES	READ 286
C	READ 287
RETURN	READ 288
C	READ 289
C	READ 290
C	READ 291
1000 FORMAT (1H1 // 9X,41H ONED: A ONE-DIMENSIONAL WAVE PROPAGATION,	READ 292
@ 49H SOLUTION IN PIECEWISE LINEAR HYSTERETIC MATERIAL //)	READ 293
1001 FORMAT (I5,1X,12A6)	READ 294
1002 FORMAT (10X,12A6)	READ 295
1003 FORMAT (/9X, 15H PROBLEM NUMBER, I5 //)	READ 296
1004 FORMAT (5(7X,4HTIME,6X,5HPRESS)/(1P10E11.3))	READ 297
1005 FORMAT (// 9X, 22H COMPUTATION CONTROLS /)	READ 298
1006 FORMAT (9X,10H NOCOMP =,I5,7X,31H PLOT AT COMPUTED SCALE FACTORS)	READ 299
1007 FORMAT (9X,10H NOCOMP =,I5,7X,30H PLOT AT INPUT SCALE FACTORS /	READ 300
@ 29X,46H USE SCRATCH TAPE (CALCULATIONS NOT REQUIRED))	READ 301
1008 FORMAT (12A6)	READ 302
1011 FORMAT (9X,10H LB =,I5,7X,24H RIGID BOTTOM BOUNDARY)	READ 303
1012 FORMAT (9X,10H LB =,I5,7X,26H INFINITE BOTTOM BOUNDARY)	READ 304
1014 FORMAT (9X,10H TEND =,1PE12.3,27H END OF PROBLEM TIME (SEC) ,	READ 305
@ /9X,10H NSTOP =,I5,7X,24H TOTAL NUMBER OF STEPS)	READ 306
1015 FORMAT (9X,10H GACC =,1PE12.3,20H ACC OF GRAVITY FOR ,	READ 307
@ 16HGEOSTATIC EFFECT)	READ 308
1020 FORMAT (///// 9X,16H INPUT CONTROLS /)	READ 309
1021 FORMAT (9X,10H NLAYRS =,I5,7X,18H NUMBER OF LAYERS)	READ 310

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1022 FORMAT (9X,10H MISP      =,I5,7X,30H SPACING CONST W/IN EACH LAYER) READ 311
1023 FORMAT (9X,10H MISP      =,I5,7X,30H SPACING VARIABLE - MUST INPUT) READ 312
1024 FORMAT (9X,10H MISP      =,I5,7X,26H SPACING COMPUTED FOR EACH,      READ 313
@ 8H LAYER. ,I2,18H SPACES PER LAYER ) READ 314
1025 FORMAT (9X,10H MATTEN    =,I5,7X,28H ALPHA(I) = 1 FOR EVERY NODE ) READ 315
1028 FORMAT (9X,10H MATTEN    =,I5,7X,30H ALPHA(I) IS INPUT FOR EA NODE) READ 316
1029 FORMAT (9X,10H NMTRLS    =,I5,7X,20H NUMBER OF MATERIALS/      READ 317
@ 9X,10H NFORCE    =,I5,7X,29H NUMBER OF PRESSURE-TIME PTS.) READ 318
1031 FORMAT (//// 9X,17H OUTPUT CONTROLS /) READ 319
1032 FORMAT (9X,15H KNTROL1 = 0,7X,10H NO PLOTS ) READ 320
1033 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,20HSTRESS - TIME PLOT ) READ 321
1035 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HACCELERATION - TIME PLOT) READ 322
1036 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HPART. VEL. - TIME PLOT ) READ 323
1037 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HDISPLACEMENT - TIME PLOT) READ 324
1038 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HSTRAIN - TIME PLOT ) READ 325
1039 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HIMPULSE - TIME ) READ 326
1040 FORMAT (9X,7H KNTROL,I1,2H =,I5,8X,24HAVG STRAIN - TIME PLOT ) READ 327
1042 FORMAT (9X,10H NHIST     =,I5,7X,27H GROUND MOTIONS NOT PRINTED) READ 328
1043 FORMAT (9X,10H NHIST     =,I5,7X,29H GROUND MOTIONS PRINTED EVERY, READ 329
@ 25H 'NHIST' TIME INCREMENTS ) READ 330
1044 FORMAT (9X,10H MVT       =,I5,7X,24H MAX VALUE TABLE PRINTED ) READ 331
1045 FORMAT (9X,10H MVT       =,I5,7X,28H MAX VALUE TABLE NOT PRINTED) READ 332
1046 FORMAT (9X,10H NNODE     =,I5,7X,24H NUMBER OF OUTPUT NODES ) READ 333
1048 FORMAT (/// 9X,19H SCALE FACTORS ARE ,7F10.1 ) READ 334
1060 FORMAT (9X,10H NOCOMP    =,I5,7X,28H PLOT AT INPUT SCALE FACTORS) READ 335
1153 FORMAT (9X,10H TNDPLT    =,F9.4,3X,24H END OF PLOT TIME (SEC) ) READ 336
1154 FORMAT (9X,10H KVEL      =,I5,7X,27H OUTPUT PARTICLE VELOCITIES, READ 337
@ 3H IN,A4,4H/SEC) READ 338
1158 FORMAT (//10X,' MATERIAL NUMBER ',I3/ READ 339
@ ' MATERIAL IS ',A8,' DENS=',F10.5,' DAMPL,Q=',1P2E10.3, READ 340
@ ' TENSIN=',E10.3,' NPOR=',I2,' NPR=',I2,' NSOL=',I2,' NVAR='I2) READ 341
1160 FORMAT (' SHMOD=',1PE10.3,' YIELD=',E10.3,' TANPHI=',E10.3) READ 342
1349 FORMAT (' STOP IN READIT, UNABLE TO INITIALIZE AT GEOSTATIC ', READ 343
@ 'STRESS',/' NL,J=',2I4,' GEOEST,SIG1=',1P2E10.3,' RHO,DENS=', READ 344
@ 2E12.5) READ 345
C READ 346
1720 FORMAT (//10X,12A6/' FORCE HISTORY ON FIRST NODE') READ 347
4101 FORMAT (1H1,12A6, READ 348
@ /30X,18H LAYER PROPERTIES ///4X,5HLAYER,24X,6HBOTTOM, READ 349
@ 5X,5HGAMMA,5X,15HPCRT SPACING,3X,7HTENSION/9H NUMBER,9X, READ 350
@ 5HNODES,12X,4H(FT),6X,5H(PCF),5X,5H( % ),5X,4H(IN),5X,5H(PSI) //) READ 351
4102 FORMAT (5X,I2,9X,I3,3H - ,I3,4X,F10.4,2F10.2,F10.4,E11.3) READ 352
4103 FORMAT (//4X,5HLAYER,18X,26H LOADING MODULI ( PSI ) ,28X, READ 353
@ 28H UNLOADING MODULI ( PSI ) /3X,6HNUMBER,6X,3H1ST,7X,3H2ND, READ 354
@ 7X,3H3RD,7X,3H4TH,7X,3H5TH,7X,3H6TH,12X,3H1ST,7X,3H2ND,7X,3HERD, READ 355
@ 7X,3H4TH/) READ 356
4104 FORMAT (5X,I2,3X,6F10.1,5X,4F10.1) READ 357
4105 FORMAT (//4X,5HLAYER,5X,41HSTRAINS AT KNEES OF LOADING CURVES ( % READ 358
@ ,15X,43H% OF MAX STRESS AT KNEES OF UNLOADING CURVE /3X, READ 359
@ 6HNUMBER,6X,3H1ST,7X,3H2ND,7X,3H3RD,7X,3H4TH,7X,3H5TH,22X,3H2ND, READ 360
@ 7X,3H3RD,7X,3H4TH /) READ 361
4106 FORMAT (5X,I2,3X,5F10.3,15X,3F10.3) READ 362
4202 FORMAT (/// 40X,30H SPATIAL ATTENUATION FACTORS//(10X,10F10.4)) READ 363
4600 FORMAT (///10X,18H OUTPUT DESIRED AT /5X,9HDEPTH(ft),13X,4HNODE /) READ 364
4601 FORMAT (2X,F10.4,13X,I5) READ 365
END

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SUBROUTINE SSONED (LS, J, N, M, D, DOLD, P, EMAX, TENSIL, PMAX,	SS1D	1
@ SPEED2)	SS1D	2
IMPLICIT REAL*8 (A-H,O-Z)	SS1D	3
C ADAPTATION OF STANDARD VARIABLE MODULUS MODEL USED AT WES	SS1D	4
C EXPANDED FOR MULTI-DIMENSIONAL PROBLEMS	SS1D	5
C DEFINITION OF NTYPE FOR TREATMENT OF UNLOADING PROCESSES	SS1D	6
C NTYPE = 1, UNLOADING CURVE TRANSLATES, MODULI AND STRESS ARE	SS1D	7
C UNCHANGED.	SS1D	8
C NTYPE = 2, UNLOADING CURVE IS MODIFIED IN PROPORTION TO THE	SS1D	9
C PEAK STRESS REACHED, MODULI ARE UNCHANGED.	SS1D	10
C NTYPE = 3, UNLOADING CURVE IS MODIFIED IN PROPORTION TO THE	SS1D	11
C PEAK STRESS REACHED, STRAIN INCREMENTS ARE UNCHANGED.	SS1D	12
C EL(I,M) PERTAINS TO THE STRAIN INTERVAL FROM I TO I+1.	SS1D	13
DIMENSION EL(10,6),FL(10,6),SL(10,6),RHOS(6),KPEAK(6),KFIN(6)	SS1D	14
DIMENSION MTRANS(6),NTYPE(6),DORSTR(10),PORMOD(10)	SS1D	15
DATA MAT/0/	SS1D	16
DATA PRESCAL/6.8948E4/	SS1D	17
CHARACTER*4 LPRESS,LDENS	SS1D	18
C *****	SS1D	19
C ***** INITIALIZATION *****	SS1D	20
NPRNT = 1	SS1D	21
IF (LS .GT. 0) GO TO 100	SS1D	22
MAT = MAT+1	SS1D	23
MTRANS(M) = MAT	SS1D	24
RHOS(MAT) = D	SS1D	25
READ (5,*) KFIN(MAT),NTYPE(MAT),LPRESS,LDENS,PUNIT,DUNIT	SS1D	26
KFINM = KFIN(MAT)	SS1D	27
IP = 0	SS1D	28
ID = 0	SS1D	29
IF (LPRESS.EQ.'PRES' .OR. LPRESS.EQ.'PRE' .OR. LPRESS.EQ.'PR'	SS1D	30
@ .OR. LPRESS .EQ. 'P') IP = 1	SS1D	31
IF (LDENS.EQ.'PRES' .OR. LDENS.EQ.'PRE' .OR. LDENS.EQ.'PR'	SS1D	32
@ .OR. LDENS .EQ. 'P') IP = 1	SS1D	33
IF (LPRESS.EQ.'DENS' .OR. LPRESS.EQ.'DEN' .OR. LPRESS.EQ.'DE'	SS1D	34
@ .OR. LPRESS .EQ. 'D') ID = 1	SS1D	35
IF (LDENS.EQ.'DENS' .OR. LDENS.EQ.'DEN' .OR. LDENS.EQ.'DE'	SS1D	36
@ .OR. LDENS .EQ. 'D') ID = 1	SS1D	37
IF (IP .EQ. 1) LPRESS = 'PRES'	SS1D	38
IF (ID .EQ. 1) LDENS = 'DENS'	SS1D	39
IF (PUNIT .EQ. 0.) PUNIT = PRESCAL	SS1D	40
IF (DUNIT .EQ. 0.) DUNIT = 1.	SS1D	41
WRITE (6,1001) KFINM,NTYPE(MAT),LPRESS,LDENS,PUNIT,DUNIT,MAT,	SS1D	42
@ RHOS(MAT)	SS1D	43
KMAX = KFINM	SS1D	44
IF (LPRESS .EQ. 'PRES') KMAX = KFINM-1	SS1D	45
READ (5,*) (PORMOD(I),I=1,KMAX)	SS1D	46
WRITE (6,1005) (PORMOD(I),I=1,KMAX)	SS1D	47
READ (5,*) (DORSTR(I),I=2,KFINM)	SS1D	48
WRITE (6,1005) (DORSTR(I),I=2,KFINM)	SS1D	49
FL(1,MAT) = 0.	SS1D	50
DORSTR(1) = RHOS(MAT)	SS1D	51
DO 30 I = 1,KFINM	SS1D	52
EL(I,MAT) = PORMOD(I)	SS1D	53
IF (I .EQ. 1) GO TO 30	SS1D	54
FL(I,MAT) = PORMOD(I-1)	SS1D	55
SL(I,MAT) = DORSTR(I)*DUNIT	SS1D	56
IF (LDENS .EQ. 'DENS') SL(I,MAT) = (DORSTR(I)*DUNIT/RHOS(MAT)-1.) *SS1D	SS1D	57
@ 100.	SS1D	58
DORSTR(I) = RHOS(MAT)*(0.01*SL(I,MAT)+1.)	SS1D	59
IF (LPRESS .NE. 'PRES')	SS1D	60
@ FL(I,MAT) = FL(I-1,MAT)+EL(I-1,MAT)*(SL(I,MAT)-SL(I-1,MAT))/100.	SS1D	61
IF (SL(I,MAT) .LT. SL(I-1,MAT) .AND. KPEAK(MAT) .EQ. 0)	SS1D	62

@ KPEAK(MAT) = I-1	SS1D 63
IF (LPRESS .EQ. 'PRES')	SS1D 64
@ EL(I-1,MAT) = (FL(I,MAT)-FL(I-1,MAT))/(SL(I,MAT)-SL(I-1,MAT))	SS1D 65
@ *100.	SS1D 66
30 CONTINUE	SS1D 67
C CALIBRATE MODULI AND PRESSURES ON THE HYDROSTAT (TO DYN/CM2)	SS1D 68
DO 55 I = 1,KFINM	SS1D 69
EL(I,MAT) = EL(I,MAT)*PUNIT	SS1D 70
FL(I,MAT) = FL(I,MAT)*PUNIT	SS1D 71
55 CONTINUE	SS1D 72
C PUT PRESSURES AND MODULI IN PSI UNITS	SS1D 73
DO 60 I = 1,KFINM	SS1D 74
EL(I,MAT) = EL(I,MAT)/PRESCAL	SS1D 75
60 FL(I,MAT) = FL(I,MAT)/PRESCAL	SS1D 76
C PRINT THE SET OF STRAINS, DENSITIES, PRESSURES, AND MODULI	SS1D 77
WRITE (6,1040) M, (I,I=1,KFINM)	SS1D 78
WRITE (6,1045) (SL(I,MAT),I=1,KFINM)	SS1D 79
WRITE (6,1060) (DORSTR(I),I=1,KFINM)	SS1D 80
WRITE (6,1050) (FL(I,MAT),I=1,KFINM)	SS1D 81
WRITE (6,1055) (EL(I,MAT),I=1,KFINM)	SS1D 82
C PUT PRESSURES AND MODULI BACK INTO DYN/CM2	SS1D 83
DO 65 I = 1,KFINM	SS1D 84
EL(I,MAT) = EL(I,MAT)*PRESCAL	SS1D 85
FL(I,MAT) = FL(I,MAT)*PRESCAL	SS1D 86
65 SL(I,MAT) = SL(I,MAT)*0.01	SS1D 87
SPEED2 = EL(1,MAT)/RHOS(MAT)	SS1D 88
1001 FORMAT (' KFIN=',I3,' NTYPE=',I3,' LPRESS= ',A4,	SS1D 89
@ ' LDENS= ',A4,' PUNIT=',1PE12.5,' DUNIT=',E12.5,	SS1D 90
@ ' MAT=',I3,' RHOS=',1PE12.5,' FROM -SSONED-')	SS1D 91
1005 FORMAT (3X,1P8E10.3)	SS1D 92
1040 FORMAT (6X,8H MAT'L =,I2,9X,26(1H-),23H LOADING AND UNLOADING ,	SS1D 93
@ 26(1H-)/6X,9H PT = ,6X,10I10)	SS1D 94
1045 FORMAT (6X,13H STRAIN (%) =,10F10.4)	SS1D 95
1050 FORMAT (6X,13H PRESS(PSI) =,1P10E10.3)	SS1D 96
1055 FORMAT (6X,15H MODULI (PSI) =,3X,1P10E10.3)	SS1D 97
1060 FORMAT (6X,13H DENS(G/CM3)=,10F10.6)	SS1D 98
RETURN	SS1D 99
C *****	SS1D 100
C COMPUTE PRESSURE	SS1D 101
C *****	SS1D 102
100 MAT = MTRANS(M)	SS1D 103
NPRNT = 1	SS1D 104
KLOAD = 0	SS1D 105
KFINM = KFIN(MAT)	SS1D 106
KPEKM = KPEAK(MAT)	SS1D 107
KPEKP1 = KPEAK(MAT)+1	SS1D 108
KPEKM2 = KPEAK(MAT)-2	SS1D 109
KFINM1 = KFINM-1	SS1D 110
POLD = P	SS1D 111
EPS = D/RHOS(MAT)-1.	SS1D 112
IF (EPS .LT. EMAX) GO TO 190	SS1D 113
C *****	SS1D 114
C VIRGIN LOADING CURVE	SS1D 115
DO 120 IL = 1,KPEKM2	SS1D 116
IF (EPS .LT. SL(IL+1,MAT)) GO TO 150	SS1D 117
120 CONTINUE	SS1D 118
IL = KPEKM-1	SS1D 119
EMAX = EPS	SS1D 120
150 P = FL(IL,MAT)+EL(IL,MAT)*(EPS-SL(IL,MAT))	SS1D 121
PMAX = MAX(P,PMAX)	SS1D 122
EMAX = MAX(EMAX,EPS)	SS1D 123
STFACT = 1.	SS1D 124

KLOAD = 1	SS1D 125
C *****	SS1D 126
C UNLOADING AND RELOADING ROUTE (AND SOUND SPEED^2 FOR ALL CASES)	SS1D 127
190 GO TO (200,250,300) NTYPE(MAT)	SS1D 128
C PSEUDO STRAIN FOR TYPE 1 UNLOADING - TYPE 1 -	SS1D 129
200 DO 210 IP = KPEKP1,KFINM	SS1D 130
IF (PMAX .GE. FL(IP,MAT)) GO TO 220	SS1D 131
210 CONTINUE	SS1D 132
IP = KFINM+1	SS1D 133
220 CONTINUE	SS1D 134
DEPSH = -(FL(IP-1,MAT)-PMAX)/EL(IP-1,MAT) -EMAX +SL(IP-1,MAT)	SS1D 135
EPSUDO = EPS+DEPSH	SS1D 136
GO TO 350	SS1D 137
C PSEUDO STRAIN FOR TYPE 2 UNLOADING - TYPE 2 -	SS1D 138
250 EPSUDO = EPS	SS1D 139
IF (PMAX .GT. 0.)	SS1D 140
@ EPSUDO = SL(KPEKM,MAT)+FL(KPEKM,MAT)/PMAX*(EPS-EMAX)	SS1D 141
GO TO 350	SS1D 142
C PSEUDO STRAIN FOR TYPE 3 UNLOADING - TYPE 3 -	SS1D 143
300 EPSUDO = SL(KPEKM,MAT) + EPS-EMAX	SS1D 144
STFACT = PMAX/FL(KPEKM,MAT)	SS1D 145
C COMMON ROUTE FOR PRESSURE AND SOUND SPEED SQUARED	SS1D 146
350 DO 360 I = KPEKM,KFINM1	SS1D 147
IF (EPSUDO .GT. SL(I+1,MAT)) GO TO 380	SS1D 148
360 CONTINUE	SS1D 149
I = KFINM	SS1D 150
380 IF (KLOAD .EQ. 0) P = (FL(I,MAT)+EL(I,MAT)*(EPSUDO-SL(I,MAT)))*	SS1D 151
@ STFACT	SS1D 152
SPEED2 = EL(I,MAT)/D	SS1D 153
IF (ABS(D-DOLD) .GT. 1.E-6) SPEED2 = (P-POLD)/(D-DOLD)	SS1D 154
IF (SPEED2 .GT. 0.) GO TO 385	SS1D 155
C WRITE (6,1390) J,N,KLOAD,I,SPEED2,P,POLD,D,DOLD	SS1D 156
C 1390 FORMAT (' J,N=',2I4,' K,I=',2I3,' SP=',1PE10.3,' P,PO=',2E10.3,	SS1D 157
C @ ' D,DO=',2E12.5)	SS1D 158
SPEED2 = EL(I,MAT)/D	SS1D 159
GO TO 387	SS1D 160
385 IF (NPRNT .EQ. 0) GO TO 390	SS1D 161
387 CONTINUE	SS1D 162
C IF (KLOAD .GT. 0) WRITE (6,1153) J,N,I,IL,MAT,D,EPS,EMAX,P,PMAX,	SS1D 163
C @ SPEED2	SS1D 164
C IF (KLOAD .EQ. 0) WRITE (6,1382) J,N,I,MAT,D,EPS,EMAX,EPSUDO,P,	SS1D 165
C @ PMAX,SPEED2	SS1D 166
390 IF (P .GT. TENSIL) RETURN	SS1D 167
400 P = TENSIL	SS1D 168
SPEED2 = EL(KFINM,MAT)/D	SS1D 169
RETURN	SS1D 170
C 1153 FORMAT (' SS J,N,I,IL,MAT=',5I4,' D=',1PE12.5,' E,EMAX=',1P2E10.3	SS1D 171
C @ ' P,PMAX=',2E10.3,' SP=',E10.3)	SS1D 172
C 1382 FORMAT (' SS-UN J,N,I,MAT=',4I4,' D=',1PE12.5,' E,EMAX,ESUD=',	SS1D 173
C @ 3E10.3,' P,PMAX=',2E10.3,' SP=',E10.3)	SS1D 174
END	SS1D 175

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